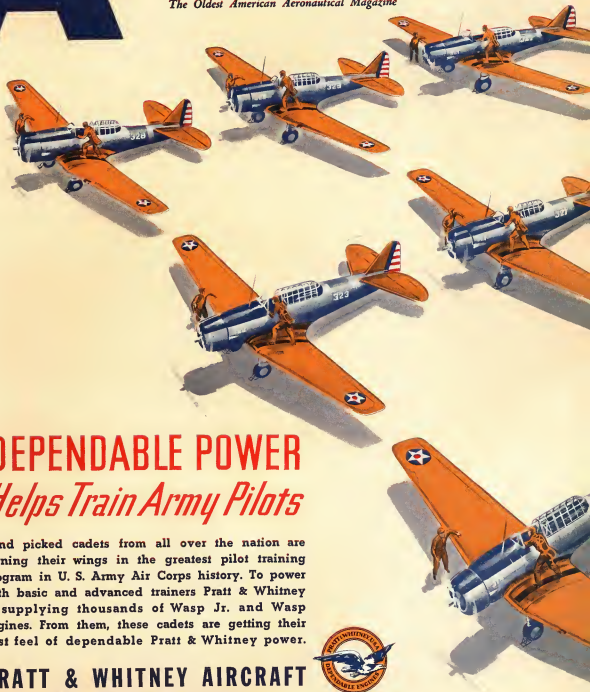


AVIATION

The Oldest American Aeronautical Magazine



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Hand picked cadets from all over the nation are earning their wings in the greatest pilot training program in U. S. Army Air Corps history. To power both basic and advanced trainers Pratt & Whitney is supplying thousands of Wasp Jr. and Wasp engines. From them, these cadets are getting their first feel of dependable Pratt & Whitney power.

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with Mechanical Engine Starter

Introduced only last year, the three-place Piper Cruiser has quickly become the favorite of fliers during low fuel cost, economical operation and greater load capacity. It fulfills your every flight requirement. With dual controls in place the Cruiser is an ideal 2-5 trainer. An engine starter, 65-horsepower engine, compressed compass and dual hydraulic brakes provide a safe, easy-

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FREE FLYING COURSE in his own plane is offered the purchaser of a new Piper. It consists of eight hours of dual flight instruction: 1) takeoffs, landings, taxiing and fundamental air maneuvers. At the completion of the first course the average pilot is ready to solo.

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| ★ ENGINE STARTER
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For Fleetness in Flight

In the Army's fastest fighter planes—the Bell Mustang, the Lockheed P-38 Interceptor and the Curtiss P-40—there's an Allison liquid-cooled engine whose fine design helps make them the fast, fit fighters they are.

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your favorite broadcast program—or even as marine direction finder. Plus interphone communication.

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operates from AC, DC, or self-contained dry-cell supply. Use it in your ship with external antenna and headphones or use it on the ground with as much as 1000 ft. high fidelity loud speaker. Power designed for maximum use of operator—bright flow control. Soundproofing range like variable at radio unit.

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Major Al Williams, also "Harvard Wing-Ten," Maj. Carl Ericson Products, Carl Building, 2000-10th, Pa.

I think that I shall never see,
A guy as good as G.A.G.
A fact that guys is very true:
To be these 'guys' you have
to...
A guy that I hope is many ways.
Used in every 'case' these days.
Planes are flown by guys like me.
But only G.A.G. makes G.A.G.

While it comes to working duties, there seems to be a number of different schools of thought. For instance, there is the backing, or lockdown, as is the job.



Seed in your stream; boys and dams
try this in your own place!

I evaluated my production, and, people were that he was, he offered us to be the first some Gulf American Churches, which he knew



I allowed him to put two gallons in the tank, and then pointing the little engine back out, I turned back for the cake oil.

The damage was assessed at \$10,000-200,000 which I honestly believe is conserved even, and they're telling me, so I'm saving you
Hawaii

NORMAN F. DALLMEYER
University of Illinois at Chicago

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Starch helps the body's energy building is usually dependent on the last meal eaten. TO MAKE SURE YOU GET THE MOST FROM YOUR STARCH, you can be intelligent and self-protecting for life. For more on benefits of starch, visit www.starch.com or contact your local starch distributor for more info and more facts. © 1999 Starch

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*is Equipped
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PRECISION BEARINGS

This high-speed all metal plane—the product of Brewster Aeronautical Corporation, Long Island City, N. Y., and Newark, N. J.—is one of the latest and fastest engined fighters developed in the United States. It has been supplied, and is being produced, in large numbers for the U. S. Navy, as well as for foreign governments.

NORMA-HOFFMANN PRECISION BEARINGS are used in the control system in rudder pedals, ball centers, links, and hinges for rudder, elevators, ailerons, flaps, and tabs—the points where friction-free operation, combined with rigidity and dependability, are vital.

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NORMA-HOFFMANN BEARINGS CORP., STAMFORD, CONN., U.S.A.

AVIATION March 1941

**BENDIX REVOLUTIONIZES
THE HYDRAULIC
FOUR-WAY VALVE!**



**ANOTHER DEVELOPMENT BY
BENDIX
OF BUREAU**

Advanced features of design and operation have been incorporated by Bendix into a Four-Way Valve weighing just 10 pounds and which requires only 75 inches of clearance.



Extensive comparison with light weight and reliability are combined in the new Hydraulic Four-Way Valve developed by Bendix of Bureau.

The unit incorporates poppet valve elements which reduce from a single control cam shaft. This construction provides the least cubic volume and weight per square inch of port area and results in the shortest possible fluid paths. Backlash or side action are avoided and the valve requires a minimum of space for panel mounting.

Especially designed valve elements ensure smooth, quiet operation, even in excess of control strokes, and the absolute reliability and freedom from appreciable wear of this exclusive construction has been demonstrated in life tests of over 100,000 cycles.

The new Bendix Four-Way Valves may be equipped with integral solenoid and thermal relief valves. They are available in three sizes, covering all tube requirements from 1/2 inch to 1 1/2 inch. Write or wire for the Bendix Hydraulic Catalog.

BENDIX AVIATION, LTD.
BUREAU, CALIFORNIA

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ACTUATING CYLINDERS • RELIEF VALVES • HYDRAULIC HAND PUMPS • CUSTOM BUILT SAE'd EQUIPMENT

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VIEWPOINT



● Viewpoint is the great distinguishing characteristic among airplane builders. Bell Aircraft is building its leadership on a new viewpoint expressed in the Autocrat, in which design is dictated by foresight into aviation's needs... contrasted with the common viewpoint in which purpose is afterward found for new developments.



BELL AIRCRAFT CORPORATION, BUFFALO, NEW YORK



IN

Sales!

50 to 65 H.P.

Lycoming 50 to 65-horsepower engines achieved first place in light-plane engine sales for 1940. The preference for these "stars of the skyways" in America's civilian training planes marks a growing appreciation of the Lycoming 240000 engine's amazing economy and unfaltering reliability... the same championship characteristics which have long made Lycoming radials a favorite in the training planes of America's armed forces.

STARS OF THE SKYWAYS

Illustrated is the Lycoming 65-horsepower engine Caters four-cylinder models are available in 50, 55, 75, 100 and 115 horsepower and six-cylinder models in 150 and 175 horsepower. They are all horizontally opposed and air cooled with provision for mechanical or electric starter and accessory drive.

FREE LITERATURE

See Lycoming 50 to 175-horsepower light-plane engines may be obtained from: Lycoming Corp., Lycoming, Pa.; Rockwell and Tuller Corp., Detroit, Mich.; Lycoming Division, Corp. Aero, Lycoming, Pa.; Lycoming Corp., Lycoming, Pa.; Lycoming Corp., Lycoming, Pa.; Lycoming Corp., Lycoming, Pa.

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Contractors to the U.S. Army and Navy



YOU CAN RELY ON

LYCOMING

50 to 200 H.P.



Engines

FOR MILITARY AND CIVILIAN TRAINERS ★ FOR PRIVATE AND COMMERCIAL PLANES

AVIATION, March 1941



R.A.F. *over* U.S.A.

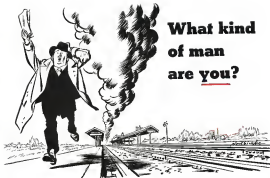
Skies over San Diego roar with test-flying of Consolidated Liberators, war camouflaged for the R. A. F.

Side by side with deliveries of U. S. Army's B24, these huge bombers roll off final assembly at Consolidated for air delivery to the front line of Democracy. Britishers say of them, "the best military airplane ever built in America".*

Men, management and money are bending every effort here to get America's first job done.

* The Aeroplane, Nov. 29, 1940.

CONSOLIDATED
Aircraft CORPORATION
ESTABLISHED 1923
SAN DIEGO, CALIFORNIA



**What kind
of man
are you?**

HAVE YOU ever skipped breakfast in order to catch the seven-o'clock?

Have you ever tossed your wife a kiss from over the left shoulder as you broke down the breakfast in a dash to the Suburban Special?

Ever spent a whole beautiful Saturday at your desk when all the time you know how gorgeous it was out on the freeway?

If you have . . . then chances are you're one of those fellows to whom business is a highly important subject. Chances are you'd readily admit that your business is just about the biggest thing in your life!

And chances are, too, you already know all about a magazine called *Business Week*.

You probably know already how packed its pages are with the exciting, critical news of business. How far it reaches to keep Management informed . . . how jealously it protects the Facts, how thoroughly it probes the Reasons.

You're sure for yourself the once-a-week job this magazine does. And it's easy for you to see that Management uses its pages in a way not shared by any other magazine, of any kind.

All of which knowledge . . . if you happen to be an advertising man . . . must come in pretty handy when you're making up your budget!

And must have had something to do with the 19% gain in advertising racked up by *Business Week* in 1940!

Business Week
ACTIVE MANAGEMENT'S MAGAZINE

WHenever resistance welding is employed in the fabrication of aircraft assemblies, you will invariably find welding tips, wheels and dies of Molybdenum being used. They have demonstrated the same superiority in this vital field as they have in the automotive ... and in every other industrial field which involves the permanent joining of similar or dissimilar metals with greater speed and for greater strength.

In many other phases of aviation activity Mallory serves with equal distinction. Many Mallory approved precision parts are standard original equipment in the products of the leading instrument and communications manufacturers. Electrical contacts of Mallory-Wells and complete control assemblies are the most widely used of any make by aircraft engine manufacturers.

In fact, there is scarcely a phase of existence . . . whether it's manufacture, operation or maintenance, to which Mallory services or products are not valuable contributions.

P. R. MALLORY & CO., Inc., INDIANAPOLIS, INDIANA

THE MALLORY

SAFETY COMPONENTS
MALLORY SERVES THE AVIATION, THE AVIATION INSTRUMENT AND THE AVIATION COMMUNICATION FIELD WITH WELDING TIPS, ELECTRICAL CONTACTS, SPECIAL ALLOYS, MAINTENANCE VIBRATORS, WIRESPACES, CONDENSERS, ROTARY SWITCHES, SINGLE AND MULTIPLE PUSH BUTTON SWITCHES, COMMUNICATIONS HARDWARE, MOTOR BATTERY CHARGERS.

PESCO

HYDRAULIC PUMPS

**meet the need for
higher pressures**

Designers of modern aircraft find Pesco engineers alert to changing trends. For example, to meet the need for higher hydraulic pressures, particularly on military planes, they find specially designed Pesco pumps setting new standards of efficiency and capacity.



New
PLUNGER HYDRAULIC PUMP

Efficient, high capacity piston pump for hydraulic systems called on to deliver consistently high pressures at all speeds. These advantages are offered —

1. Adaptable for pressures up to 5000 p.s.i.
2. Sizes from .007 cu. in. displacement upward.
3. Light weight: typical unit of 0.4 cu. in. displacement for 1500 p.s.i. weighs only 3.95 lbs.
4. High efficiency of oil speeds from 100 to 3000 r. p. m.
5. May be used as hydraulic motor.
6. Volumetric efficiencies up to 95%.



HIGH SPEED HYDRAULIC PUMP

This unit provides high capacity with extremely low weight. Pump has 0.7 cu in. displacement and weighs 2.2 lbs. Capable of pressures up to 1200 p.s.i. Recommended speeds are 2200 to 4000 r.p.m.



**HAND OPERATED
HYDRAULIC PUMP**

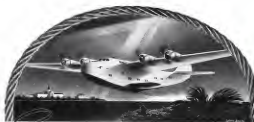
A standley hand pump — lighter in weight and more efficient in operation. Weighs 2 1/2 lbs. — providing 1 1/2 m.p.s. displacement per cycle. For pressures up to 100 p.s.i. Smaller design available for 1500 p.s.i. and more.



Other Pease Products include fuel pumps, valves and accessories, air pumps and accessories, hydraulic cylinders and control equipment and many other units built in order for unusual requirements.

PUMP ENGINEERING SERVICE CORPORATION

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Safety WITH HAZARD AIRCRAFT CONTROLS



● Hazard Aircraft Equipment is made by the people who have long clung to the slogan "In Business for Your Safety." This slogan is far more than a phrase at the Hazard plant; it is an inflexible doctrine. Nothing, whether materials or methods, is ever slighted or relaxed when safety is involved.

This means that Hazard control equipment is dependable. For instance, all Hazard control cables, whether "KORROSION" stainless steel or Hazard fused or galvanized strand and cable are pyroformed. This adds greatly to their resistance of bending fatigue—and permits the application of Hazard's TRU-LOC fitting, the 100% efficient terminal. Hazard Aircraft and Engine Hoisting Slings are equally dependable. Of course, every piece of Hazard aircraft equipment is made to strict accordance with the latest Army and Navy specifications.

For certain dependability—for safety of control operation—specify Hazard Aircraft Cable and Fittings.

HAZARD WIRE ROPE DIVISION Established 1915

AIRCRAFT DEPARTMENT, 208 Park Avenue, New York City
11101 Grand Avenue, Bldg., Detroit, Mich.
14715 Floral St., Los Angeles, Calif.



HAZARD Aircraft CONTROLS



Sentinel of Security



The
Republic

P-13

CHARGED with the defense of America's skies, the U.S. Army Air Corps has developed a swift and formidable line of pursuit aircraft.

Distinguished by its speed and fighting ability, the high-altitude Republic Aviation "Lancer" with Curtiss Electric Propellers, takes its place among those planes that are filling the exacting requirements of America's national defense.

CURTISS PROPELLER DIVISION
Curtiss-Wright Corporation • CALDWELL, N. J.

CURTISS *Electric*



Serving AIR, LAND AND SEA...

IN ALL of our defense forces of air, land, and sea, Breeze products will be found in action. Breeze equipment for aircraft, aerospace, and marine installations has long been in extensive use by both the Army and Navy.

Yet Breeze is not primarily a defense industry. It produces equipment for the airlines, for the automotive and other industries, even steel foundries for lubrication. Now the great expansion of defense production finds Breeze equipped to supply, in addition, such vital needs as radio dialing for aviation and secondary systems, emergency engine starters, cartridge engine starters, and similar accessories for defense.

Expanded production has been accomplished without deviation from the rigid quality standards which Breeze has maintained in production for the entire industry.



Radio Systems and Auxiliary Wiring •
Aircraft Instrumentation Components •
Wire Shielding Controls and Filters •
Control Switches

Automotive Engine Systems •
Aircraft Engine Components •
Aircraft Fuel Systems •
Aircraft Oil Systems •
Aircraft Lubrication

Radio Sets and Case Assembly •
Radio Case Assembly •
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Breeze
CORPORATIONS INC.
NEWARK, NEW JERSEY

Techniques that bring out the best in
Control Systems •
Control Systems •
Control Systems

Generator and Ignition Systems •
Generator and Ignition Systems •
Generator and Ignition Systems

Radio Sets and Case Assembly •
Radio Sets and Case Assembly •
Radio Sets and Case Assembly



CAN you fight motor fires that occur in flight?
Can you instantly detect an engine blaze?
Kill fire in 3 or 4 seconds? Save your car!

Airlines, U. S. Army and Navy fighting
planes have the answer. A metal ring surrounds
motor, directs fire out of existence with carbon
dioxide snow-and-gas. That's LUX protection.

By combining LUX Flame Detectors with
LUX extinguishers you can fire before it gets
started. Visible or audible alarm gives warning.
Directional valves permit discharge of LUX

gas into any motor of a multi-engine plane.
There are LUX Starter Systems for motor en-
gines on well as radio.

**AIM...
SHOOT... KILL THE FIRE**

How LUX fire extinguishers for motor protection

A lightweight fire-fighter, equipped for
airplane use, just like LUX
extinguishers at the point where
engine-fires start. It's easy to aim up
the fuselage with a LUX-LUX fire
extinguisher system.



Have you seen "Wise Men"? It describes
LUX fire protection for planes. Your
copy is free. Send for it.



Walter Kidde & Company

INCORPORATED

322 WEST STREET, BLOOMFIELD, N. J.

AVIATION, March, 1948



REPUBLIC AVIATION

POWER AT WORK . . . The rocket-like speed and fine all-round performance of Republic Aviation Pursuit-Interceptors demonstrate the value of design principles which successfully utilize the tremendous power-output of America's great air-cooled radial engines—unmatched in efficiency at high altitudes...unfaltering in service for extended intervals between overhauls. Holders of the longest single order for fighter aircraft ever awarded by the U. S. Army Air Corps, Republic Aviation Corporation is ready for today's job—and tomorrow's.

REPUBLIC AVIATION CORPORATION
FARMINGDALE, LONG ISLAND, NEW YORK, U. S. A.



Defense Procurement Program Accelerates

THE PROGRESS of the defense program has exceeded expectations in the past two months and is entering a phase of rapid acceleration. Aircraft production is forging ahead much more rapidly than was generally expected. In January, 962 military airplanes were delivered out of a total production of 1044 of all types. This is somewhat better than the equivalent made in January in January ("The Truth About the Defense Program"—T. P. Wright).

Of the 962 military airplanes approximately 30 percent were fighters and here increases reached 12 per cent late in January. A large proportion of the combat ships were, as might be expected, single-engine fighters. Production of P-40s at the Curtiss-Wright plant led the combat field, having reached 18 per cent and the extent of the Bell Aircraft Company, also in Buffalo, is also rapidly accelerating because of the increase in production of Allison engines, there are more than enough available for the needs of these two companies, and it is expected that by the close the P-40 is in quantity production, the Allison output will be sufficient to meet the needs of its three users.

Radial air-cooled engine production has also made more rapid strides than might have been expected. Both large manufacturers have passed the million-horsepower-per month mark and have been delivering engines in such quantities as to make a substantial dent in the "glide line" of airplanes previously

built and awaiting engines. Although we cannot expect any substantial contribution to engine production from the newcomers in the automotive industry before Fall or the end of the year, it is reasonable to believe that this new capacity will come along in time to assist materially in the rapid acceleration of the engine production which is just beginning and will continue throughout the year. Most important addition to the extensive group of engine producers is Chrysler, which

intends to manufacture a 2000 hp liquid cooled engine were announced as this was written.

The present expansion plans for engine capacity, however, will probably be insufficient for the new aircraft production program which involves the use of government-owned assembly plants operated by aircraft manufacturers and using parts provided by the automotive industry. Although the final details for this program have not yet been worked out, the War Depart-



15 0000 Pratt & Whitney engine. Left to right Eugene E. Wilson, president, Detroit Aircraft Corporation; A. V. D. Wilson, chief engineer; W. A. Perkins, executive chief engineer; William Lawrence, mechanical superintendent of assembly; S. H. A. Wilson, senior engineer; Donald Jack, assistant factory manager; and L. B. Kibbe, engineering manager.

Busy!

**16,000 LANDINGS AND
TAKE-OFFS A WEEK**



VIN. BARREDO, Vice president of Aviation Facilities, Inc., where the planes of every prominent aviation institution are serviced and kept in top condition

THEY PREFER TEXACO

- International selling you know it the best. It is identified with Texaco. No other oil brand.
- More Texaco are there. A solidified fleet from any other brand.
- More scheduled airline airports within the U.S. and to other countries. It follows with Texaco that all other oil brands.
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TEXACO MAKES WHAT YOU WANT TO BUY
IN THE AVIATION INDUSTRY
BETTER FUEL
BETTER LUBRICANTS
BETTER SERVICE

TEXACO Lubricants and Fuels
FOR THE AVIATION INDUSTRY

LOCATED 10 miles southeast of Fagan's Keesler's New York City Hall, Floyd Bennett Field was one of the busiest airports in the country. 10,000 landings and take-offs in a week. That's busy.

The use of this field by the Coast Guard, Naval Reserve, Naval Aviation, and several flying schools account for most of the services. There are always available, up-to-date inspection service, expert maintenance, and Texaco Lubricants and Fuels.

New Texaco Airplane Oil keeps engines clean with wear at minimum. Texaco Airplane Lubricants covers low wear-up and high power output.

More scheduled airline airports within the U.S. and to other countries. It follows with Texaco that all other oil brands.

The performance of Texaco in the aviation field has also made it preferred in the field lived in the past.

Buyers in these fields are enjoying many benefits from using Texaco Lubricants and Fuels.

Experienced aviation engineers will gladly make you in the selection of Texaco Aviation Products, available at leading airports in the 48 States. Please the nearest, or write:

The Texaco Company, Aviation Division, 115 E. 42nd St., N. Y., N. Y.



ment has announced that landing operations will soon be let for the four biennial assembly plants to be located at Omaha, Kansas City, Fort Worth, and Tulsa, and that the seventh manufacturing contract will be Martin, Douglas, North American, and Consolidated. Chrysler, Ford and General Motors will build parts and a production of 2000 medium and heavy bombers per year is expected.

ENCOURAGEMENT OF SUB-CONTRACTING is the next step to save time in the defense effort. Work in the defense contract service has been expanded in Washington and reaches out into all parts of the country through the Federal Reserve System. Each of these banks will be set up as a headquarters for exchange of views between primary contractors and possible subcontractors within its jurisdiction, and studies of subcontracting facilities available will be made. Although detailed organization of the system is still under way, it would be well for aspiring subcontractors to contact their nearest Federal Reserve Bank for information on how they can be of service in the defense program. (Further details on page 94).

MANY FALSE PROPHECIES have arisen overnight to satisfy the public's desire for knowledge on military aviation. Some have never had a close look at a modern military airplane. Many more have no adequate conception of the many unseen factors behind the airplane that make up a balanced air force. So we launched a deep well of relief when we saw the first copy of *Winged Warfare* by Major General H. H. Arnold and Colonel Ira C. Poirer (Harvard & Ivy). It goes without saying that both of these gentlemen know what it is all about.

Winged Warfare starts with the Greek historians, moves rapidly to 1914, and winds up with the latest available material on the present air force expansion program. It answers many of the questions in the confused public mind and some of those that have been troubling people close to the aviation business. It is well written and easy to read even for the layman.

One of the high spots in the chapter on Air Tactics, which presents many of the lessons learned at the present European War. Much light is thrown on the German success at Fieseler-Stuka cooperation and how it changed the ideas of mechanical warfare. The functions of both light bombers and tactical fighters are also explained. The discussion of the effective use of small, light, slow flying planes of the Fieseler-Stuka type in combined missions by the Germans in the Polish

campaign may furnish new hopes for the future utilization of this type of equipment in military service. In the chapter on Strategy, the important factors of timing, training, and power effort are emphasized. The basic mistake of the present emphasis in flying to provide a proper proportion of all language-handling aircraft in their air forces is particularly interesting and enlightening.

The separate air lanes that are discussed and a policy of evolution rather than revolution toward that end appears to be the feeling of the authors, although they feel that much more thinking and planning should be done before

the step is taken, if it is to be taken at all.

Both ideas of the question of civil aviation's part in warfare are given and a middle ground is suggested. The importance of commercial aviation in an industry concerned to meet a leveling wartime peak and peacetime valleys in production is stressed.

Last but by no means least, the reader is impressed with the magnitude of industry's expansion to expand productive capacity at a rate unprecedented in the world's history. The 30,000-plane-per-year program is discussed in detail, and the reader is given a 30,000-plane-per-year rate. And the dangers of the shadow-factory method are pointed out.

Winged Warfare is a valuable and timely contribution to military aviation literature. Its advent makes it possible for our citizens to get the truth about air power from men who are playing such a prominent part in building it for the nation.

IT PAYS TO **FLY**



"Maybe we'll have to spread some wings to keep the military load!"

These critical moments of ground contact before the plane is air-borne, and after the flight, are worthy of every bit of study and experience and manufacturing watchfulness we can give them.

BENDIX PRODUCTS DIVISION OF BENDIX AVIATION CORPORATION
South Bend, Indiana

Bendix

LANDING-GEAR EQUIPMENT

The picture above, showing a small group of men awaiting final test before shipment, aptly illustrates the widely varying capacities of these vital requirements to a "happy landing."

to be very deadly, you can quick
The Royal Canadian Air Force not
long ago sent out a call for 375 colorful
emojis to whip up hash for the
hungry hunger bounds. If you really
want to pitch for democracy, get in
there and say one for quadrennial
dash 21 you and dairy when space
is at a premium.

If you can't find the top row of letters
with an open space, you can still open
a restaurant of chopped horse and cook
up a stew of bull-on-a-stall. If you can't
tell old from green, never mind, you
can pour lead in the spigots. In the
black space after "Quicknesses," you
can write "none" and never a winky
Winky.

At the bottom, the "S" is for the
American Service, Signal Corps, com-
missioned by bull; cows who could



have read a recipe even if they had wanted to. Some of those boys didn't know their aspie from a pella ground. We hear that military gastronomy is a fine art now, but that doesn't get us compensation for our dearest discipline. Yes, if you can't even fly a jet, you can cook.

✶ **Parents: Reason in Plans** think they might be art blasé, so they read papers, play cards, talk small talk, or just sit and look bored. They have been encouraged to this by the airlines themselves, who frequently, wailed at swimming up the merrits of flight, advertise pastimes to heavy and get a seat with. Now, there's flying over or among clouds, it actually hurts. A friend isn't about to see it too, because

words are useless. There is the propeller up, hovering as fast as the earth is turning; if that wheel were rolling on the ground it would cover the distance in three or four hours. Watch your pilot. They are different. Some lift their ships off the ground smoothly. Some fly them off green hills. Some rest on clouds. Some land on water. You will tell there's no up. Once in a while you find a cocky one, who does harmless little things that we could tell you about. If you're unusually inclined, you'll notice one or another picture than the landing light making a bluish white disk of the propeller at night. Why, we could have been so close! We could have seen it clearly. So there! Remember! Take away those newspapers and bring us a field glass.

★ **THE STRONG PLOW**, says Eastern Air Lines in a news release, is 33 years of age, 2 1/2 to 30 in. tall, weighs 100 lbs., has blue eyes and brown hair. His temperament is complacent, controlled; his average pulse rate is 71 beats per minute. Medically speaking, his physical and mental health is so superior, one can view a city at a high tribute to his health of good, clean living. End quote. Hefeborn!

► If you listen close enough you'll catch it: you'll hear an episode of Alfred Lord Tennyson's "murmur self" verse. We've been running into it around every little corner lately. Just to put you on your guard, this is it—for the first and last time:

For I dip into the fount
For no basest eye could see,
Now the wastes of the world
And all the wonder that would be,
Now the heaven filled with commerce,
Aspirants of magic arts,
Pinks of the purple twilight,
Drooping down with rosy hues.

You who went through the journey

and Durbin criss back in the post-Lindbergh days, will know what we mean. Them-and-people-should-appeal-to-the-country-our-old-is-an-asterisk-of-42-years-ago, and the ghost stories were it then as a mid-air jet. It wasn't a bad story—the first Durbin story. A couple of years later, Durbin and I were in Italy, looking over the law on the island of Corsica and decided to vacation for Bedy. Having no other way to travel, the old man rapped up a couple of pairs of wings. For many days he used men. This proved unfortunate for young Durbin, who felt himself wide out, put too close to the sea, that was a bad thing. The boat was in the Mediterranean. No bad, but not good enough for a long run, even at a dollar a glow.

And neither is this dip-into-the-baggy form of Traverso's. It isn't even



prophecy in we are supposed to believe Tempson was born in 1899, twenty-one years after the Manfrotto brothers sent up their first balloon in 1778. That Tempson's entire life was contemporary with balloon ascensions. Indeed, the Freshwater, Huddersfield, had made an ascent in the New World, was named by George Washington, in 1793, sixteen years before Tempson was born. The point alone certainly leads to some doubts about the possibility that the other alleged effects of those travels to propel balloons with sails. We do hope that this little chapter is good enough to discourage a few incautious readers of the "magic sail" poem.



Black Week

Germany has no more bombers in day raids over England than she had on May 11th, 1940. These are Messerschmitt ME 110.

Warner

The German bombers by 17 day bombers, which are able to ground heavy bombs at relatively short distances but are not as effective against well-defended military objectives. All of the type that crashed yesterday over day but fall were shot down.



EFFICIENCY REPORT ON THE GERMAN AIR FORCE

Real raids on military objectives are far from effective, according to this British authority

By Lt. Col. W. Lockwood Marsh

Editor of the British publication, *Aircraft Engineering*

It may be interesting to give some impression of the effectiveness of the enemy aeroplane attacks that have been made on England since the occupation of northern France. It is hoped to succeed in making these observations as nearly objective as possible, and avoid exaggeration in either direction. So many rumors are in constant circulation that it is by no means easy to sift the reliability of any particular item of information, but by personal contacts checking several sources against each other, it is at a rate possible to arrive at a reasonable approximation of the facts.

It may be said at the outset that the German Air Force does not appear to be particularly successful in inflicting damage on points of military importance. It seems clear that it is finding the carrying out of raids on defended

and well-organized defenses a very delicate operation from those it made in the initial stages of the war, against Poland, on in the later campaigns, against Holland, Belgium and France. Poland was not in a position to offer any effective resistance to the sustained air attacks that were launched concurrently with the invasion and in an tactically short space of time every aeroplane was immobilized not carry anywhere rendered useless. Holland and Belgium were no better placed—and suffered equally under the disaster of being subjected to surprise attacks—while France, as was well known in England before the war started, had a particularly successful in inflicting damage on points of military importance. It seems clear that it is finding the carrying out of raids on defended

has by no means been borne out by the present by the attacks on Great Britain.

Generally speaking—and this is said without the intention of depreciating the quality of the German pilots who are, in many instances, undoubtedly courageous and able—their great difficulties in pressing home attacks in the face of strong opposition has been shown and where it has been attempted, it has not usually met with success. It has been officially stated, for instance, that as a day-bomber attack by Ju 89's on Croix-laux-André, which, owing to its close proximity to London is extremely well defended, every aeroplane was lost.

The composition of the whole of the Channel coast of France entirely gives Germany a magnificent chance, she would have thought, of rendering coastal traffic along the South Coast and through the "coast channel" of Dover impossible, and there is little doubt that there was every intention of achieving this. Repeated and heavy attacks (in some instances large numbers of machines being engaged) on convoys continuing along this route have, however, to put it mildly, not been overwhelmingly successful. That this is so is evidenced by the fact that after a week or so of repeated bombing attacks of this nature the aid of land artillery brought up to the coast of France was called in to take over, or at any rate, aid in this work. In one instance when this method also proved abortive, five battleships were sent over to press home the attack. They did not prove very more effective, since the convoys escaped from the channel also on basis of as with no losses in ships.

During the same period efforts were made to get Dover Harbor out of action. At the end of all this effort, the London Times was able to publish a photograph of the harbor and there showing down to be completely undamaged—which we can confirm from the testimony of an independent and completely reliable witness on this matter from this. Here again the aid of shore batteries had been called upon with some little success in causing of damage to a few houses in the town according to British official statements.

We may here suppose an interesting sidelight on the defenses in the attacks of the two countries towards the revision of the truth in regard to the results of air raids. In a broadcast on May 26 it was stated with considerable insistence that "attacks were also made on the RAF aerodrome at Manston, near Ramsgate, where considerable damage was done to buildings."

New the German News Agency of Aug. 12—its forthright before—had claimed: "The air base at Manston was reduced to ashes. The runway had been ploughed up by bombs of all calibers, from the heaviest to incendiary bombs. The hangars and other buildings have collapsed."

It is a little difficult to visualize the need for again attacking, and being able to cause "considerable damage" in an aerodrome which has already, only a fortnight before, been "reduced to ashes." It is not, in our opinion, invariably the admission of German commentators of the nature deliberately to build. It is rather the effect of serious aerodrome and operations in the German character which makes it difficult for a German to believe that the object of an attack can have been so disappointing as to be so disastrous. A characteristic example of this tendency was the belief on the aircraft carrier "Ark Royal" had been sunk when the ship had only under the influence of the hole in a ship's hull made to avoid a bomb which

dropped as close as 100 yards to the carrier. With this, which led the crew of the aeroplane continued to believe, and report, that the bomb had never and was. The danger of the attitude of mind from the point of view of the substantial of the country being attacked is that after a time one tends to disbelieve every claim made in the German reports—a confidence which may not necessarily be justifiable.

There is no doubt whatever that the aeroplane losses in German hands were not against defended positions is not proving to be a reason of extreme importance. Whether this is due, or there is some reason to suppose, to the inadequacy of the German losses slight at usually to the difficulty which undoubtedly was not fully realized in general quarters of maintaining the aeroplane sufficiently long on a straight and level course in the face of AA and hostile aeroplane attack to avoid being shot down. Surely the fact that the British aeroplane attack unquestionably more successful in hitting their actual objectives rather than to the former objectives.

With the best will in the world and the most determined effort to be made, it is impossible to do anything but stand on the defense superiority of British aeroplanes in comparison with those they have in the last 18 months. The fighter's have a most striking advantage.

(Continued on page 10)

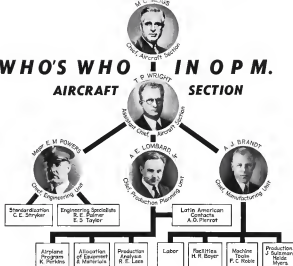
Plans of the RAF. Such like being, backed up by constant expenditure, have proved the best of their kind in the world. German planes are so much for the superior training and discipline amongst of these British pilots.



RAF pilots

WHO'S WHO IN O.P.M.

AIRCRAFT SECTION



TRANSFORMATION of the Defense Advisory Committee into the Office of Production Management as being rounded out as far as the Aircraft Section is concerned under the leadership of Merrill C. Meigs. A staff of highly qualified personnel has been attached to Washington and has taken office space in the Social Securities Building, which probably will constitute a permanent headquarters for O.P.M. The Aircraft Section is essentially a "trouble-shooting" organization set up to relieve capacity problems and coordinate shortages in various phases of the program. Details of its relationship to the Office of Production Management organization will be found on page 11. Hereafter are presented a number of thumbnail sketches of key personnel.

A publisher a private pilot with 1800 hours and a transport ticket and an able experience, Merrill C. Meigs has been at the forefront in varied fields. Born in Marshalltown, Iowa, he worked for the J. I. Case Co. of Racine, Wis., both before and after he went to the University of Chicago, and as representative of the company sold and demonstrated tractors throughout the Middle West and in South America. It was during his connection with the Case Co. that he organized the famous racing team headed by Louis Chevrolet, and then came into advertising as an executive with Lord & Thomas. In 1918 he became vice-president director of the Chicago American, and in 1928 publisher of the Chicago Herald-Examiner, now the Herald-Herald.

Associated with flying from the beginning of aviation in the U. S., Mr. Meigs was the first civilian to make a cross-country passenger flight and one of the first operators of a commercial transportation company. More recently he was a participant in the first passenger flight of the Clipper to London, and since 1937 Mr. Meigs has been flying his own plane.

If the needs of O.P.M., Aircraft Section had been known 20 years ago, and one man supposed to spend two decades preparing for its direction, his background and experience could have been no more thorough than that of Theodore Paul Wright. Many of his broad range of accomplishments are already known to readers of AVIATION and here we can only touch on the high spots.

During his attendance at MIT, he studied in the Naval Reserve Flying Corps and was commissioned an lieutenant in 1918, later becoming an lieutenant at Naval Aircraft, assigned to the Bufile and Garden City plants of Curtiss Aeroplane & Motor Co., Inc. In October, 1921, he entered the organization as executive engineer and rose rapidly to chief engineer of the airplane division in 1923. Under his direction many famous military and commercial designs were developed and an extensive research expenditure set up. In 1931 the experimental laboratory at the Garden City plant was moved to Bufile and combined with the research and production facilities quoted on under his direction. These years later he was transferred to the New York office of the Custer Wright Corporation as director of engineering and observation of the engineering policy and planning committee of the parent company.

Mr. Wright's production and engineering experience is not confined to this country, as he has traveled extensively in Europe and studied the methods used in England, Germany, Italy, and France. He has written many lectures and papers, including a series published in an American, and has made important economic studies of air transport trends. He has also done valuable work as an officer and director of aircraft standards activities of the Society of Automotive Engineers, and is a member of many associated organizations.

Chief of Planning Unit, Dr. A. E. Lombard, Jr., received his degree of Ph.D. at California Institute of Technology in 1930. Engaged for ten years (1925-35) as structural, aerodynamic, strength, and consulting engineer for Custer-Wright Corp. in Garden City, Bufile and St. Louis, he was in charge of design, construction and operation of the Bufile plant's wind tunnel.

At MIT Dr. Lombard has been assistant professor of aeronautics and mechanical engineering, and assistant director of wind tunnel research. Author of several technical papers on airplane design and aerodynamics, his thesis was written on the subject of the occurrence of flutter in aircraft. He joined the old Defense Advisory Committee last summer and did pioneer work on the present program.

To his present position as Chief of the Manufacturing Unit of Aircraft Production, Arthur J. Brandt brings a wide engineering experience gained in this country and abroad.

A Mid-Westerner, born in Walnut, Kansas, he is a graduate of the Case School of Applied Science, where he

graduated in 1911. From 1915 to 1928 his engineering activities ranged from construction of chemical and powder plants for the duPont Co. to works manager of the Stevens Tanker Div. of General Motors; modest manager of the Fisher Body Corp. and vice president in charge of operations, Pontiac Motor Co. From 1928 to 1930, Mr. Brandt has been engaged in his own consulting engineering business as president of A. J. Brandt, Inc., Detroit, Mich., and is also president of The National Tool Co., Cleveland, O.

Among the more important jobs he has handled during the past 13 years have been the layout, testing and starting in operation of the world's largest

truck plant at Moscow, Russia; reorganization of production facilities of all U. S. Baker Co. plants, including the first successful development of the conveyor system for building tanks; and reorganization of production activities and personnel of American Locomotive Co., Diesel Engine Div., Auburn, N. Y.

In his present capacity as O.P.M., Mr. Brandt will assist all aircraft and aircraft engine and parts manufacturers in speeding up production for the U. S. and Britain.

Active in military aviation since 1917, when he enlisted in the Signal Corps U. S. A. as a flying cadet, Mr. Brandt (Turn to page 13)



Merrill C. Meigs (center) discusses manufacturing problems with Charles Stryker, production manager of Ford River Rouge plant (left) and Noel Ford (right).



"I am amazed at the considerable progress being made here" said Assistant Secretary of War Robert F. Patterson (center), on his recent visit to Bufile. American's ingenuity, skill, and energy. The Secretary was the guest of President J. B. Cushman (left) Motor Supply. E. Ross (right) is Air Corps representative at Bufile American.

U. S. PREPARES ITS AIR DEFENSE

PROFITING from lessons learned in Europe, the Army's Air Defense Command has gained much valuable experience in recent months through the war games, in which "enemy" bombers try to get through to attack military and industrial objectives. Tense as the part of weeks these maneuvers have been held along the east coast last fall is upstate New York along the Canadian border, and in late Janu-

ary throughout part of New York and New England. Photographs on these pages represent these maneuvers. Swedish experience has proven that when bombers may be expected over a large area it is necessary to use civilians as an auxiliary to trained military observers, in spotting approaching aircraft. Hence one of the prime purposes in these war games has been to test how effective a civilian warning

net might be. With the help of the American Legion, thousands of private citizens have been enlisted to watch for airplanes, then rush to their telephones and dial the news to Air Defense Command headquarters. In the last recent maneuvers 10,000 observers were at some 700 observation posts. In some instances bombers got through to their objectives but were often the air defense warning net

worked so smoothly that persons interrupted them and kept off the beaches.

May later was limited to show public how progress of locating bombers is checked. Reports are limited by nature of job and number of hours.

What army of traveling bombers means. Members of the 4888 Central Postal Directory Co. march in formation.



Mr. Gen. James E. Chaney, left, shows A. T. & T. president Walter Dillard civilian observation post.



Below: A message center at Whiteford, N. Y., with enlisted men working. Each man is assigned a specific job in the center.



Location of approaching bombers is kept on a map. Several types may be many times larger than the one and is held for use in the line.



Searchlight beam from Whiteford, N. Y., shows the location of approaching bombers.



Searchlight beam from Whiteford, N. Y., shows the location of approaching bombers.

Left: Antiaircraft unit of New Coast Artillery from Fort Detrick, N. Y., attacks its target to detect an enemy bomber.

On: In the past, antiaircraft battery takes a heavy searchlight to watch for bombers.



with an idle and clean a flowings over on the vent of a high velocity bullet—the Bell P-39 Airacobra in flight.

WHY A

Rear Engine Installation

A new era of aircraft conception and design is forecast in this story of how an important step in aerodynamics was taken by Bell engineers in the design of the P-39 Airacobra

UPPLAIN as the nose on your face!" For long hours an engineer at every day, long, and for a long time the engine-fitted nose of most airplanes have been a very constant feature. The rubber bladders, bungs and usually long-tapered "nozzles" of some airplanes that have been built have been not only constantly prominent to the eye, but were generally associated with poor aerodynamic efficiency at high speed to the high drag resulting. In the parlance of the aviation industry these airplanes had "bad-on head winds"; "their noses were dirty."

The biggest reason of most of the airplanes we have seen here belong to good reason. The engine of an airplane is nearly the largest single mass of metal component. The engine must exist as most often for the installation of engines of maximum possible power has obviously not permitted the selection of smooth-faced engines. The physical limitations of materials, mechanical and thermodynamic limit the possible arrangement of airplane engines in general to one or two formulas of design—the so-called "in-line" type and the capacitor or "Radial" type. Whether of these types of engines are very small when high horsepower output is required, and the radial types have generally been the most ubiquitous at this time, due to their having the largest frontal area. Airplane engines also generally

PART I

By Robert J. Woods
Chief Design Engineer
Bell Aircraft Corp.

require much auxiliary equipment which must be closely grouped with the engine. Such auxiliary equipment adds greatly to the bulk of the engine installation and is usually the principal cause of the many bumps and bumps that are often associated with the cooling over an aircraft engine.

At Bell Aircraft Corp. at Buffalo, N. Y., has developed a new and different kind of airplane, a military single-engine fighter type with its form and clean a fuselage nose as the most of a high velocity bullet—the Bell P-39 Airacobra project now in production for the United States and British Air Services. The powerful 1150 horsepower Allison 32-cylinder in-line liquid cooled engine has been redesigned into the true streamlines of the fuselage so thoroughly and completely that to an untrained eye the actual location of the engine may not be immediately discernible. The story of how this important step forward in the art and science of aircraft design was accomplished is told here in the hope that it will herald a new era of conception

and design in which the "dirty nose" will disappear from our aircraft in favor of the "clean" entry which permits achievement of many important improvements in safety and performance.

Less than two decades ago the sight of a completely smooth entry to the nose of an airplane was not an uncommon sight. The exposed cylinder of the air cooled engines used to stick out into the streamlines quite prominently, with little nose cowling that a piece of tin wrapped around the crankcase, and the old Consolidated P-14 primary fighter with its completely unadorned nose called "Blind" was clean. The aerodynamic efficiency of these completely smooth-nosed engines, we have since learned, was almost non-existent. A few actual attempts had been made in the early days in production smooth fuselage noses on racing airplanes, but the general use of a smooth entry or smooth cowling over the engine was not prevalent.

Almost simultaneously two improvements were proposed for cooling the exposed cylinders of radial air cooled engines. One, the so-called Townsend Ring, developed in England, and the other the NACA cowling developed at the National Advisory Committee for Aeronautics Laboratories in America. Although much controversy existed at the time regarding the relative merits



"The old Consolidated P-14 primary fighter had little nose cowling that a piece of tin wrapped around the crankcase."



A few scattered attempts had been made in the early days to provide smooth fuselage noses on airplanes. This drawing shows cylinders protruding into the streamlines.



"The so-called Townsend Ring finally proved not to be the best of all at a step-up into the in the cooling problem."



"Although neither solution was perfect, the smooth NACA cowling has made a substantial improvement."

It was not the "Airacobra" that Bell first made a practical application of the other side drive installation

and illustrations of the Townsend Ring versus the NACA cowling, the Townsend Ring finally proved not to be the best of all at a step-up into the in the cooling problem. The smooth NACA cowling, although rather bulky and large, was quite an improvement as it brought with it, as a first reward, rather important increase in speed by reduction of drag. The benefits that followed in improved stability and control and finally, in spite of the early misgivings of the engine manufacturers, in improved cooling, were real advances in the science of aircraft design.

Along about the same time a definite change took place in the design of the nose of airplanes carrying liquid cooled engines. The usual flat radiator which had been installed ahead of the engine directly behind the propeller, was now placed further aft under the wings on the bottom of the fuselage in a various attempt to provide a real streamlined shape to the nose cowl of the airplane. This arrangement was used on several Army Air Corps airplanes using Curtiss D-12 and Conquest six-cylinders.

When the guns made an improved performance, stability and control and cooling better due to these changes had been established the way was clear for research and development to determine the optimum arrangement in fuselage nose design. The aircraft designers had become "nose" conscious and on time was lost in determining the optimum in reference. The airplane turned out to be our old friend the pointed nose of a high speed bullet.

The next phase of the development of nose shape was in providing cooling for radial air-cooled engines of greatly increased horsepower. The original NACA cowling designed for the J-5 and P-51 "Mustang" design was increased

considerably in just to achieve the most powerful engines and hence to develop maximum horsepower in the form of wings for taking in cooling air, cowling flaps at the exit duct and various and sundry other components. Several in-line type air cooled engines appeared on the scene, with thick leading cowling that were quite efficient. But, these engines were almost a short step removed from relatively low horsepower.

The liquid cooled in-line engines did not appreciably increase in size with increased horsepower, but on the other hand were equipped with numerous parts for the propeller and its arrangement as to provide a much smoother symmetrical streamlined nose. Propeller spacers were greatly used, and the reduction were moved off further aft to a position under or aft of the wings. The clean streamlined bullet nose was beginning to come into the picture, although the number of struts, bumps, dents, machine gun blast tubes, exhaust manifolds, etc., that the practical design efforts and standardization experts managed to get stuck on the outside were in any the least detrimental.

However, new problems began to come into the picture when, with the increase in horsepower of the engines, the weight became a greater proportion of the total weight of the airplane and severely affected the airplane balance. The engines had to be pushed back toward the wing so that the mass of the single engine airplane became outboard and began to resemble conventional and long. Also, the necessity for countable pitch and automatic type properties with their relative large movement of weight augmented the balance problem.

To add to all of these problems as weight-engineered military type aircraft, the long standing requirement for two engine use to maintain gun performance (Continued on page 10)



Seaplane or Landplane



The Yeaghi-Slatery VB-44B, one of the latest designed flying boats.

THE time is now just when doubts exist regarding the practicability of extended travel by air across the ocean. The general public and airline operators already know that it is not a matter of transportation over the vast stretches of water is safe, convenient and economical. A large demand for such transportation is inevitable. Present conditions already clearly indicate the desirability of

planes, of somewhat larger capacity than those in service, and which would be capable of crossing the Atlantic, sea-stop.

Although flying boats have been well up to the point for this service, there are some arguments concerning the desirability of using land planes instead of flying boats in the future. Numerous papers have been presented and articles written on the subject. Some of them,

in the opinion of the author, presented satisfactory conclusions. There are proposals for various radical designs which are not practical now or far in the future due to the lack of engineering and service information to prove the merits claimed.

The presence of discussions and different opinions reveal a great interest in this important subject and are very beneficial, especially new and in con-

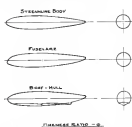


Fig. 3. Shape of the fuselage and hull(hull) in vicinity of 100 ton gross weight airplanes.

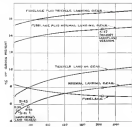


Fig. 5. Trend of weights of fuselages and attributable landing gear for airplanes. Points are from Yeaghi-Slatery designs.

for Transoceanic Service

The author believes that a properly designed flying boat of a size of 50-100 ton gross weight, besides greater safety and convenience of operation, would have an equal cruising speed, appreciably better payload or longer range at that speed, compared with a land plane of the same gross weight, power plant and volumetric capacity.

By Michael E. Gluhareff,
Chief of Design, Yeaghi-Slatery Div. United Aircraft Corp.



Michael Gluhareff

nection with the national defense of our country, where long-range airplanes are no absolute necessity, particularly flying boats, which are independent of long runways and the resultant large flying fields.

It is the object of this article to analyze objectively the reasons why a flying boat is a more practical and advantageous type of aircraft than a land plane for long-distance over-water service.

General Speculations

Obviously, the airplane, for today's day means is the ultimate achievement. Continued improvement in design, power plant and material or its materials, one type of aircraft may alter the situation tomorrow. However, in the light of present knowledge, let us consider only the airplane with which we are most likely to be faced now for making meaningful transoceanic service — airplanes on nearly the conventional design of today.

It is proper to mention here that

from a purely aerodynamic standpoint, the flying wing type is an excellent solution for the long range machine but for passenger service it is impractical.

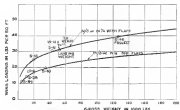


Fig. 4. Wing loading plotted against Gross Weight. Points taken from flying boats designed at the Yeaghi-Slatery Division, United Aircraft Corporation.

oil because there is not sufficient room inside. The study of a 100-ton flying wing as a trans-Atlantic sea-stop passenger carrier revealed that there was barely enough useful space to accommodate power plant and fuel if it is to be an efficient design.

Coming to present and expected demands for transoceanic air service, it seems reasonable to assume that ships of 50-100 ton gross weight will be needed to carry the payload. Safety demands the recovery of multi-engine aircraft with not less than four engines, capable of taking off under its own power and crossing the ocean sea-stop. It is hardly unreasonable at the present time to ask for speeds, throughout the entire range, in excess of 300 mph at altitudes of 5,000-10,000 ft. Such speeds should have ample room for the accommodation of passengers, mail and cargo. Statistical data suggests that these airplanes be capable of transoceanic sea-stop landings in over-cast.

For the sake of comparison of the two types of aircraft likely to be designed for such service—landplanes and flying boats—let us consider airplanes that are equal in gross weight, power plant, cruising speed, cruising altitude, payload and volumetric capacity. The value of the airplane type may then be measured by its range.

Discussion

The author and the experienced writer, which he is connected, have been working on and studying long-range airplanes of nearly all types for military and commercial use for many years. The practical experience in design and construction of large flying boats for the company indicates also, what everything is taken into consideration, the all-around practical, long range transoceanic airplane is likely to be of the flying boat type.

(Turn to page 124)

Eliminating Propeller Failures

In 1932 there were 41 propeller failures in flight. Last year there was none. Here is the dramatic story of the research that brought this improvement.

By Eric Martin

Hamilton Standard Propeller Division
United Aircraft Corp.

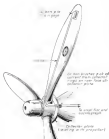


Fig. 1. Measuring vibration stresses in flight. As explained to the test, stress impulses from the point of measurement on the blade was transmitted in the secondary device to the vibrator motor. This wave from the motor, after the blade was mounted along the blade and left at the hub in a sleeve plate behind the propeller. This plate vibrates with the propeller, and current is picked up from reference wave on the rear face by means of brushes and carried by wires to the recording device.

PROPELLER FAILURES IN FLIGHT	
Year	No. of failures
1932	41
1933	31
1934	21
1935	11
1936	11
1937	11
1938	11
1939	11
1940	11
1941	11
1942	11
1943	11
1944	11
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2100	11

THE most important propeller problem since the earliest days of aviation has been the problem of safety. While any failure of the airplane structure is likely to be a serious matter, a propeller failure is usually disastrous. The safety of propellers has been derelict a matter of the utmost concern to the entire aerospace industry.

As early as 1936 when propellers were constructed entirely of wood, the problem of vibratory stress was the primary cause of propeller failure. Since the propeller is a relatively simple structure, the study stress can be calculated by well-known methods. The vibratory stresses, however, have never been subject to analysis or calculation, and have been the cause of practically all propeller failures ever since the early days of aviation. The vibratory stresses were not subject to analysis or calculation, and have been the cause of practically all propeller failures ever since the early days of aviation. The vibratory stresses were not subject to analysis or calculation, and have been the cause of practically all propeller failures ever since the early days of aviation.

The advent of the metal propeller greatly increased the problem due to the fact that the fatigue strength of metal was far greater than that of wood. In fact, the fatigue strength of metal was 25 percent to 50 percent of the ultimate strength of the material, as compared to 75 percent to 100 percent in the case of wood. Attempts were made to remove the safety of propellers by overload tests on stress motor whirling rigs at speeds up to three times the engine power for which the propeller was intended, and by an additional out of 100 hr. or more power on the engine also comparison of the wheel test. While these tests were very helpful in determining the propeller's service life, they were not adequate as a complete solution. The only satisfactory way of proving the safety of any given propeller-engine-airplane combination was to conduct actual service tests for several thousands of hours on a full-scale or more complete airplane, which might not be conducive to the safety of crew or passengers during the tests.

When a failure took place during this service test, a comparison was made and the service test repeated to see if the

failure had proved to be a propeller test. This method of ascertaining the safety factor of propellers often required one to two years' time. This situation was of course intolerable and at least some ten years ago Hamilton Standard recognized that stress tests be used for measuring the vibratory stresses of propellers in use so that the actual operating stress level in the propeller could be determined.

Accordingly Hamilton Standard embarked on an extensive program of research on stress problems which has extended over a period of years and during which time various methods of testing them were carefully investigated. Out of these investigations and study evolved the method of measuring vibratory stresses in service. This technique has been developed to such an extent that a complete survey of propeller blade tip and shank stresses throughout the entire operating speed range may be made within several hours of operating time. A complete survey of the airplane engine-propeller combination can be made in flight. The records obtained by this test method, in combination with the calculated steady propeller stresses, give a complete picture of the conditions under which the propeller is operating, and permit the engineer to predict the actual operating factor of safety within limits of error heretofore impossible. The information obtained in this way at a few hours of test time is undoubtedly of greater value in predicting and measuring propeller safety than was previously possible, even after the successful start test, engine test and service test.

Description of Method

Over the period of years from the adoption of metal propellers up to 1935, stress occurred occasionally, failure of the propeller blades and blades, which could only be explained by a repeated loading (fatigue) and unloading of the propeller material. It was well known that if a stress cycle of sufficient magnitude were repeated sufficiently

often a failure could be produced at a stress below the ultimate strength of the material. It was also known that if the stress were reduced below a certain level, no failure would result in either case when the stress were repeated.

Repeating vibration of the propeller blades caused by the power impulses from the engine were suspected as the source of the repeated stress cycles which resulted in these failures. Very approximate estimates could be made of the speed at which these blade vibrations might become critical, but said a qualitative picture of the im-

portance of these critical stresses could be obtained, the problem of propeller safety was largely a matter of opinion based by qualitative experience.

In attempting to measure propeller vibratory stresses, certain fundamental and practical problems were immediately apparent. The aerodynamic forces at a propeller tip at take-off speed may easily reach a value of 6,000 times the weight of a fully plated shaft. Light weight of the instrumenting equipment was, therefore, important. A continuous response which could be observed away from the propeller at all times was almost a

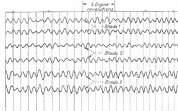


Fig. 2. Sample oscillogram showing stresses at six different points on a three-bladed propeller.



Fig. 3. Test equipment has been utilized in such a point that strain-measuring points of stress conditions at twelve points on the propeller core can be made. This shows apparatus in the test house.

SUMMARY

There is a brief description of the work on stress problems done by the Hamilton Standard Propeller Division of United Aircraft. The extensive results of the study are as follows:

1. It has definitely and substantially increased the safety of aircraft by reducing propeller failures in flight to the vanishing point.
2. It has improved the performance of aircraft by providing direct information on the weight of propellers, which was rapidly reducing critical problems.
3. It has greatly reduced the development time required on new propeller designs by eliminating very long laboratory tests for long periods of service testing previously required.
4. It has contributed materially to improved design of other aircraft parts, particularly engines, by determining various types of engine stresses, the proper corrective measures to be applied.

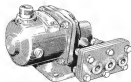
As indications of the extent to which the value of this achievement has been demonstrated, the following facts apply to themselves:

1. The Material Division of the United States Army Air Corps, the Bureau of Aeronautics of the United States Navy, and the Civil Aeronautics Board now require all types of propellers to be subjected to these vibration stress tests prior to their final approval for service use.
2. Every modern airplane-engine-propeller combination now in service on aircraft in the United States has been subjected to these tests.
3. In spite of the tremendous growth of both commercial and military aviation, the number of propeller failures has rapidly declined since that individual fact has been in fact and during the twelve months of 1946 for the first time in history, there were no propeller failures.

reducing, so that an electrical instrument was apparently required. If the measurements were to be made in service conditions of propellers, an attaching means which did not cause or damage the propeller was essential. And, perhaps, most important of all, the stress pickup must be capable of being precalibrated and anchored at the completion of the test.

The method which was finally evolved by Hamilton Standard is relatively simple yet provides a complete and accurate record of the stresses involved. It depends upon the phenomenon that certain materials increase and decrease

(Continued on page 112)



A liquid propeller anti-ice pump with electric motor, explosion-proof motor, and an electric pump with an electric motor.



The motor parts used in the anti-ice pump shown in the previous image.

Carburetor and Propeller ANTI-ICERS

WITH the successful solution of wing icing by means of rubber de-icers, the problems of propeller and carburetor ice remained. Slinger rings and anti-ice pumps have largely eliminated propeller wing icing, and alcohol injection has solved the carburetor icing problem. We are all more or less familiar with the methods of application, and interest now centers in the specific forms of equipment involved, and the factors governing their design.

Consider propeller icing. We know that a mixture of alcohol-glycerine fed to the propeller by a slinger ring spreads over the surface of the propeller blades in a more or less effective pattern, reduces the adhesion of the ice as it forms and strengthened force due to ice. Slinger rings are more

or less standardized. The liquid used has been standardized. The amount of liquid required falls within certain definite limits.

The last remaining variable in the anti-ice pump design derives largely in the slinger ring. There is every brand and variety of pump on the market; single outlet, double outlet, four outlets, diaphragm type, gear type, even inverted vaneless types have been used.

Reliable pumps that operate under all conditions are a "must" for any anti-icing system.

By David Gregg, Chief Research Engineer, Eclipse Aviation

What is the best type of anti-ice pump and why? That is a difficult question and no equally definite answer is available. Let us consider the basic ice forms on the propeller under many different atmospheric conditions and there are many different types of ice formations. The pilot cannot read an instrument and say, "I need 2 cc. of liquid per hour per propeller" or "I need 4 cc. . . ." He still judges the amount of liquid required by feel and by sight.

When he first uses the anti-ice pump he turns it to a high rate of flow until the initial ice formation is removed, then gradually reduces the rate, keeping the flow just above the point where the ice sloughs away from the root of the blades and engine vibration has been reduced. The flow that accomplishes these results differs for different ice conditions. As the pilot judges the amount of liquid required by the feel or sound of ice removal from one propeller, it is essential that all other propellers receive approximately the same amount of liquid in order to remove effectively their ice. The pilot must

have some definite way of knowing that this result is being accomplished. That is the first requirement.

The second requirement follows. There must be some effective control by which the amount of liquid required can be varied. Several parts of a slinger operator and accurate measurement of the amount of liquid used under varying weather conditions indicate maximum limits of from 3 to 8 cc. per hour per propeller, the higher flow being sent when the anti-ice pump is first turned on to remove accumulated ice. An average of approximately 5 cc. per hour per propeller is used under ordinary ice conditions.

The third requirement is based on the characteristics of the liquid used. We know that an alcohol-glycerine solution may evaporate in the pump and lines when they result unused, leaving behind a sticky glycerine deposit that collects in low spots in the system and blocks the outlet lines, or clogs the opening parts of the pump. When the anti-ice pump is placed in operation a must have sufficient power to overcome initial friction of the powered parts and develop enough pressure to force any deposit of glycerine from the lines. It must be capable of making starting and continuous operation at temperatures normally below the freezing point. It, therefore, is essential that the maximum power available to drive an anti-ice pump be largely in excess of the actual operating requirements otherwise the pump may not function.

As alcohol is volatile, the liquid in the tank will gradually absorb water and the resulting mixture is far more corrosive than the fresh solution. The materials used in the pump and pump assembly accordingly be selected to prevent this corrosion.

Then there are the four basic requirements of an anti-ice pump: 1. Uniform distribution of flow. 2. Control of rate of flow. 3. Adequate reserve power and pressure. 4. Freedom from corrosion.

Uniform flow distribution in several propellers can be secured by a single outlet pump and means of accurately dividing this flow, or by a multiple outlet pump with a separate outlet for each propeller. Various combinations and arrangements are possible. Consider the normal installation as an airplane and the equipment involved is shown in the following table. Since the length of tubing required has been based on the assumed distance from the pilot's cockpit to the center of discharge at the wing tip, the diameter of the lines goes as the cube of the engine nozzle, and the distance from the base of the engine nozzle to the slinger ring. These required distances are particularly liable to occur in actual practice due to bend-

ing in the tubing, so that the estimated tube length is, at best, too short.

Several facts are apparent from a study of this chart. First it is obvious that the use of liquid flowmeters, such as a float in a tapered glass tube, increases the required length of tubing from two to ten times, while an electric flowmeter has no effect on the tube length. Next, it is interesting to note that the estimated length of tubing for uniform distribution in a 4-engine airplane is nearly the same as that for 2-engine airplanes, when two, instead of four, engine pumps are used, and that the arrangement gives the maximum tube length. Assuming then that a two-outlet pump provides the most desirable protection and still affords the simplest service facilities, what type of pump is best? As far as the drive goes, an electric motor is directly indicated and this choice has been generally universal. Now let the pump itself. The single

outlet propeller anti-ice pump resulted from the collaboration of several groups intercepting propeller ice removal. Information in this time favored a dual discharge type pump, having several outlets of these units have been sent and have given good service. As first, guaranteeing of the motor itself, most difficulty, but a new type of double outlet pump, with two outlets, has been developed. Accumulated research and service experience led to the development of the gear type anti-ice pump with an outlet in gear or side, and two outlet parts. For a two-outlet gear pump, two arrangements are possible, two sets of discharge pumps driven by a single motor, or a single three-outlet pump. The latter type elements are gear, and can use of balances as well as several bearings—on a relatively more accurate than the double pump. As the required output is low, the gear and bearing must be made in very close tolerances of the pump is to provide uniform flow from one outlet under widely varying pressure characteristics. Laboratory and service testing points in maximum and minimum, between the inlet of the gear and the housing, at 1000-1500 to provide proper flow. Such elements require intensive liquid the range of ordinary production methods, and involve adequate tooling and thorough inspection.

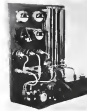
Such a pump is fabricated with well, both assembled and disassembled. Term and weight have been reduced to a minimum. The unit is completely explosion-proof, and includes an explosion-proof AN type discharge pump. The dry weight of this pump is equal to the 10 to 15 lb. of alcohol-glycerine solution, and even the gear work are wet, a vacuum of approximately 20 in. of mercury can be maintained. The discharge pressure in the pump is made through the installed tubing and is normally 5 to 10 lb. per sq. in., but the pump is capable of building pressure of from 300-600 lb. per sq. in. if necessary, to remove unyielding slugs of glycerine from the lines. Though the manufacturing process must be closely controlled is a pump of this type, reverse requirements are not entirely eliminated. Aside from changing the gear brushes or partially in oil and, there are no parts that should require repair or adjustment is a service life beyond that of the assembly in which the unit is to be installed. The uniform distribution of flow possible with a pump of the type described here is shown by the output curve, with one outlet at zero pounds back pressure, the other at 15 lb. per sq. in. back pressure.

With distribution of flow taken care of by proper design and manufacturing methods, the rate of flow can be accurately controlled.

(Turn to page 128)



Pressure drop (lb. per sq. in.) versus flow rate (cc. per hour) for various pump configurations. The graph shows that a single outlet pump has a higher pressure drop than a double outlet pump, and that the pressure drop increases with flow rate.



Flow rate (cc. per hour) versus distance from cockpit (feet) for various pump configurations. The graph shows that the flow rate decreases as the distance from the cockpit increases, and that a double outlet pump maintains a higher flow rate than a single outlet pump.

Anti-Ice Installation for 4-Engine Airplanes

Type of Pump	No. Req'd	Location	No. Tanks	Location	Type of Flowmeter	No. Req'd	Foot of Tubing Req'd
2 outlets	1	Eng Nozzle	1	Eng Nozzle	Electric	1	40
4 outlets	1	Eng Nozzle	1	Eng Nozzle	Electric	1	100
2 outlets	1	Engine	1	Engine	Electric	1	40
Single	1	Engine	1	Engine	Electric	1	100
Single	1	Eng Nozzle	1	Eng Nozzle	Liquid	4	160

1 cc. of alcohol-glycerine solution per hour per inch of tubing.

Anti-Ice Installation for 2-Engine Airplanes

Type of Pump	No. Req'd	Location	No. Tanks	Location	Type of Flowmeter	No. Req'd	Foot of Tubing Req'd
2 outlets	1	Eng Nozzle	1	Eng Nozzle	Electric	1	40
4 outlets	1	Eng Nozzle	1	Eng Nozzle	Electric	1	100
Single	1	Engine	1	Engine	Liquid	2	70
Single	1	Eng Nozzle	1	Eng Nozzle	Liquid	2	81

Institute of the Aeronautical Sciences

Each year the value of the annual meeting of the Institute of the Aeronautical Sciences is increased. It represents a cross-section of the industry's technical developments—a few of the important papers are abstracted in the following.

ONCE again the annual meeting of the Institute of the Aeronautical Sciences has been held and once again there were presented papers that give a representative view of what research and technical work is being carried on throughout the industry. Presided by the Honors Night Dinner, on Jan. 25, the meetings lasted for three days, with sessions in the morning, afternoon and at the evening.

The Honors Night dinner had more "boom-bust" of the industry per square inch than have ever before been collected in one place. Frank M. Caldwell, vice Director of Research for the United Aircraft Corp., was inaugurated as president of the Institute on this evening year, succeeding Major James H. Doolittle. Learning highlights were the talks by Major Caldwell on that last year's contributions to the I Ae S and by Griffith Brewer, distinguished president of The Royal Aeronautical Society, who recounted accounts of early aviation.

With this as a good start the next day got under way with some 400 technical papers presented. Although open door, not present in many of all of the papers given, the following abstracts give a cross-section of what important subjects were covered.

Air-Cooled vs. Liquid-Cooled Aircraft

By John S. Lee
United Aircraft Corp.

There are many reasons for choosing an air-cooled or a liquid-cooled power plant. They may be technical, economic, or practical. They may depend upon the equipment of certain factories, or the collective experience of a group of engineers, or the previous equipment of an air base. The present paper is confined to the technical reasons.

The basic drag of a streamlined object is first considered. . . . The curve of drag coefficient vs. freestream velocity is a reminder that practically speaking drag is not caused by frontal area alone, but is largely caused by skin friction. It is a point fact, a streamline object has no drag at all. . . . It is clear that when we increase the diameter of a fuselage

to take a larger engine the increased drag due to the increase of frontal area, other things being equal, is an interesting by-product of frontal area and fuselage, rather a side effect upon the wing. This increase in drag is very much smaller than the increase in area, but is sufficient to cause us to favor a compact liquid-cooled engine.

Morrell (J. W. Morrell, "Cooling of Aircraft Engines," R&M 1464, August, 1935) and others have shown that if air is admitted to a passage at high speed, expanded through a diffuser to a condition of low velocity and high pressure, and then heated, the heat energy imparted to it may be recovered as thrust, if the air is discharged rearward through a nozzle or jet. This is the basis of the "direct" radiator of the liquid-cooled engine. It is also the air-cooled engine itself.

The only difference between the radiator and the air-cooled engine is that the radiator can be fitted with duct work without consideration of the resistance of the duct and hence engineers favor this a relative simple change to make. The important point is not whether the air-cooled or the liquid-cooled engine has the lower cooling drag, but that the cooling drag of both is being rapidly reduced, and may ultimately become zero.

Any system of heat energy recovery from the engine requires, however, from the engine, ultimately, to the cooling air, that transfer to be efficient requires that the heat be available at the highest possible temperature. Below the high temperature, air-cooled cylinders are intrinsically more effective for heat energy recovery than the lower temperature coolant radiators. It should be noted also that only a maximum amount of cooling air should be wasted. This is where the "direct" radiator has had an advantage in the past. The bulk of the air-cooled engine have not been good enough improvement in the internal

able at the highest possible temperature. Below the high temperature, air-cooled cylinders are intrinsically more effective for heat energy recovery than the lower temperature coolant radiators. It should be noted also that only a maximum amount of cooling air should be wasted. This is where the "direct" radiator has had an advantage in the past. The bulk of the air-cooled engine have not been good enough improvement in the internal



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Sciences

able at the highest possible temperature. Below the high temperature, air-cooled cylinders are intrinsically more effective for heat energy recovery than the lower temperature coolant radiators. It should be noted also that only a maximum amount of cooling air should be wasted. This is where the "direct" radiator has had an advantage in the past. The bulk of the air-cooled engine have not been good enough improvement in the internal



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cooling and heating of air-cooled engines at high speeds. . . . Throughout the remainder of this paper the cooling power for both air-cooled and liquid-cooled engines has been arbitrarily taken as 3 percent of the engine rated output. This value represents a conservative average from Pratt & Whitney turbine models.

In any discussion of air-cooled and liquid-cooled engines, the question of the relative heat consumption always comes up. It is recognized by all that the heat consumption under full throttle conditions of the air-cooled engine is substantially poorer than for the liquid-



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cooled engine. However, no engine, whether turbine or conventional, runs at wide open throttle for more than a very small fraction of its total time. Even the piston plane was spent the majority of its time looking for its owner, or getting from its base to the probable zone of operation. The significant fuel consumption, then, is the cruising fuel consumption. In studying the comparative weights of air-cooled and liquid-cooled installations, the actual weights of a considerable group of engines and their installations were re-evaluated. This resulted in a series of data, much of it modified, but in all cases too extensive to be presented in this paper. . . . Whenever possible, actual weight



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values were used. It will be seen that some of the weight values varied over rather wide limits. The liquid-cooled engines, for instance, have a range of about 15 percent. This is because some engines included two-stage gas-driven superchargers while others relied upon turbo superchargers. In all cases, the individual weights were pooled with due consideration and with every effort to make them strictly comparable.

The evidence presented thus far indicates that the cooling power and the heat consumption of the air-cooled and liquid-cooled engines may be taken as follows.



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relatively equal, but that the liquid-cooled installation has a certain advantage in drag and the air-cooled has a corresponding advantage in weight. . . .

Using the data presented in the first part of this paper, various groups of airplanes have been tabulated and pre-

sented of air-cooled and liquid-cooled engines compared in such groups. Comparison of the engines on one or two points, others are modifications of present engines; still others are only on paper. . . .

We are concerned in comparing airplanes on the basis of equal power and pay little attention to the fact that the liquid-cooled airplane, being heavier, is also the larger. . . . We are presently concerned not with the power of the airplane but with its performance, its cost, and the difficulty of holding it.



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We must include in that performance not only air speed, but in climbing, landing and maneuvering ability.

We have not discussed availability, reliability, or engine cost. These are difficult questions in hard-to-guess territory. A very good qualitative case can be made for the superiority of the air-cooled engine as regards availability and reliability. A bullet through the cooling duct or the buffer, for instance, will not stop an engine.

In this study it has been clearly brought out that, on the basis of equal weight, equal diameter, and equal manufacturing characteristics, the liquid-cooled installation can equal the speed of the liquid-cooled, and surpass it in climb, climb, and climb. When range and weight-carrying ability are the primary considerations, the air-cooled has every advantage.

Propeller Design Problems of High-Speed Airplanes

By R. S. Doolittle
Lockheed Aircraft Corp.

THE problem of propeller design for high-speed airplanes is the familiar one of compromising conflicting experimental data by means of inadequate theory. Airplane designers have actually flown at speeds in excess of 400 m.p.h. and are being designed for higher speeds, yet, to the author's knowledge, there is available not a single reliable test on a propeller at forward speeds of this magnitude. Therefore, the only resource is in the substitution of

(To be continued)



Applying Automotive Methods to Aircraft Production

Part II of an article describing what can be accomplished in planning projects and processes and the various elements necessary in achieving high industrial efficiency

PART I of this article, printed in the January, 1951, issue of AVIATION (p. 42), discussed mass production and suggested four objectives by which such methods could be accomplished: Product Engineering, Tool Engineering, Cost Analysis and Production Control. In the concluding part there is offered the final three of these objectives in mass production in the aircraft industry.

Tool Engineering is greatly influenced by quantity and the basic design of the part. Properly planned and executed tooling contributes much toward the efficiency of the mass production principle particularly:

- (A) Properly planned operation sequences
- (B) Simple load-proof jig and fixture design
- (C) Gauging and locating points common to mating parts

- (D) Non-interfering work heights and positions
- (E) Reasonable application of time and motion studies
- (F) Providing adequate inspection tools

The advantages and necessity of proper tooling cannot be over emphasized since the manufacture of inter-

changeable parts is the indispensable requisite precedent to mass production. Therefore, in the production of increased quantities of aircraft it is inefficient to merely respond proportionately to the constant manufacturing involved. Better tooling requires added expenditures since the product, as influenced by mass demand, must be produced more quickly and more economically. In this regard, advantages can be derived by recognizing increased respect for the automobile industry's attitude and dependence on effective tooling.

The inspection man, although involved in its application (by quality requirements) is of more importance than is generally conceded, for it is one of the most essential factors in controlling variations in the product. Inspection expedites production, whereas for example, the exclusive and ignorant automobile inspection lack. Therefore, it is

imperative that provision should be made in the airplane tooling budget for inspection equipment. Good inspection tools inspire greater confidence in inspection results.

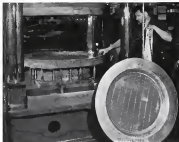
The final, "Accurate Cost Analysis and Regulation" is more than keeping books to record expenditures. The automobile industry in mass concerns and the modern car would be impossible without this consideration. Accurate estimates are necessary for reliable cost predictions and these in sufficient provision and outlaying of aircraft manufacturing expenses due to be of material assistance in compiling useful

cost-like precision with which cars are produced is significant evidence of its effectiveness. Another element of such control that has had practically no recognition in the aircraft industry is the application of economies for size (the economic cycle) or at it is more generally known the mass economical number of parts to make in a setup. Hereafter, there has been limited opportunity to apply this science in aircraft but new circumstances are much more favorable to its adoption.

Material handling can be regarded as part of production control and it is a very much neglected phase of manufac-

turing. Its function is more than organizing and moving material through the plant. In fact, an efficient plant layout is planned on the basis of proper routing of material between the process stations, assembly departments, storage depots, etc. Considerable care is taken of material handling and planned layout studies provide means to the best solution of the problem. In the automobile industry there are different solutions with regard to sales, customer's desires and requirements, trend of the era, etc. Furthermore, there is this important difference: the automobile industry flourishes under mass advantageous conditions because this type of product can be directed toward a mass sales production. In addition, the experimental models will finally go into

(Turn to page 122)



Showing the need in continued welded down dies. Read and is made that the being to choose only because of which steel plate. These down dies can be broken so many parts by merely setting the material over the punch. By doing of the down heads material achieve proper shape.



less welding of exhaust stacks has reduced to a minimum the unacceptability involved in the welding of such parts. Gas welding is difficult to control on a consistent scale of linear inches or mill is decreased.



The trend is to lower gas welding open area like is specified in order down dies. Generally however area is now or used here required about 30 minutes of labor with a total investment of 5 dollars. On the other hand, the area can be produced on a total investment of \$25.



The trend is to lower gas welding open area, particularly on location lines. In which the welds have been substituted. A superior product is achieved, also a more profitable rate of production.

ICE-FREE CARBURETORS FOR LIGHTPLANES

Here is one very practical solution to a problem
that is often serious for lightplane operators

By L. V. Kimball

Motor Laboratories
New Haven, Conn.

DURING the last two years there has been a great increase in aviation flying, much of which is done in Class 1 and small Class 2 airplanes. Although carburetor heat has been available on practically all of these planes there have still been many complaints about carburetor ice in almost tropical climates which could be readily diagnosed as carburetor ice.

Symptoms

Many new pilots are so yet unfamiliar with the symptoms of carburetor ice, and, for this reason, it seems proper to explain them. The ice is usually formed in the carburetor at a fairly slow rate, depending on the outside air conditions, and is characterized by a drop in power or falling off of rpm. Sometimes this is accompanied by a roughness of the motor caused by ice around the venturi of the carburetor which makes the motor very rich.

When sufficient carburetor heat is available for take off and cruising conditions the carburetor may only ice up on the approach before landing in which case the pilot finds that on opening the throttle the engine either stalls or has practically no power. This is embarrassing if it is found necessary to go around again or run the throttle to correct under shooting the field.

Cause

Most of the smaller ships are equipped with carburetors which are subject to ice formed by refrigeration. This ice is formed from moisture removed from the air by the latest test of refrigeration of the fuel line. This phenomenon can be demonstrated by leaving the fuel in a thermometer. As the liquid evaporates, the temperature will drop. The temperature drop through the carburetor is about 40 deg. but varies with different arrangements. This

causes the moisture particles to be condensed out of the charge, and under several conditions these particles will cling so pure to the carburetor. The more common ice accumulations are found around the throttle shaft or butterfly valve. This is dangerous as it gradually closes off the passage, thereby reducing the power.

The second most common formation is found around the jet which sprays the gasoline into the air stream. This ice dangerous as it increases the reaction on the jet and makes the mixture very rich. This may continue until the engine will cease to operate. The ice that formed is particularly dangerous as it decreases the available carburetor heat in two ways, first by



A typical engine installation in a lightplane showing the position of the carburetor directly below the cockpit with the air intake in front.

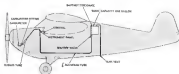


Diagram of the installation of a tank for Ethyl Alcohol or Avial in relation to carburetor, preventing ice.

reducing the power and second by covering the mixture.

Effect of Weather Conditions

It is possible, to a great extent, to predict icing conditions although these predictions are not reliable. Carburetor ice may be expected in temperatures of less than 50 deg. where the dew point is less than 5 deg. below the prevailing temperature. To what extent the available carburetor heat will prevent the formation of ice is largely a matter of experiment, and varies greatly with the equipment and weather conditions. During actual precipitation the amount of carburetor required is somewhat greater. It is assumed that most of the light plane operation is done during fairly clear weather and thus very little flying through overcast conditions is necessary or possible.

The Cure

One positive method of ice elimination is suggested here which is required in engine operation where the carburetor heat available is sufficient to cause a 40 deg. temperature rise in the intake air. This makes use of Ethyl Alcohol or Avial instead of the carburetor air intake. The installation consists of a supply tank of about 1 gal. capacity mounted inside the ship at some point where it can be filled conveniently and sufficiently high above the tank.

(Turn to page 122)



Continued from page 121

History Repeats itself

DURING the first World War Goodyear biplane tires were extensively used on American military aircraft—because even in those days they represented years of progressive development. Since then Goodyear has paced aviation progress with many major new contributions, headed by the Airwheel and the Hydraulic Disc Brake. So it is not surprising that today, as America rears in the air, Goodyear again is taking the lead in supplying not only tires, but tubes, wheels and landing gear for all types of fighting ships. For Goodyear, through long cooperation with the industry, was ready with proved and tested wheel equipment to meet the existing demands of modern military aircraft. Perhaps our engineers could serve you, too. Write: Aeronautics Department, Goodyear, Akron, Ohio, or Los Angeles, California.



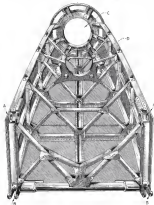
THE ORIENTED NAME IN RUBBER
GOODYEAR

ON YOUR NEW DIP GOODYEAR AIRWHEEL TIRES, TUBES, WHEELS AND BRAKES

Goodyear-T. W. The Goodyear Tire & Rubber Company

AVIATION SKETCH BOOK OF DESIGN DETAIL

One of the features of the Duplester Mustang D-31 monoplane is the hydraulically operated engine control for rotating and unloading the position of the propeller blades. The gear action of the engine is located at "A" with the engine mounted in front of the cockpit and the camshaft drive shaft in the propeller assembly through "B" in "C". During take-off, camshaft connected to "B", and hydraulically operated valve the engine, hence the propeller, for increased lift-off performance. The engine is a Junkers Jumo 223.



Shown above is the leading gear of the Duplester D-31. With a slip of this also large wheels are mounted on the front and rear of the main leading gear shaft to take the torque inside. The mounting shaft is at the upper left taking the wheel to operate the center of the slip.



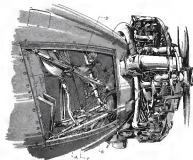
The leading gear of the Duplester D-31 is shown above. To indicate the size, the gear "A" is the height of an average man. Notice the shaft leading off the top.



The Duplester engine leading gear of the Duplester D-31. The large wheel of "A", when rotated, operates the center of the slip shaft, the wheel of "B" is geared up into the slip shaft at "C".



General drawing of the engine of a Vought V-101 engine is shown above. The primary gear is shown at "A", in which is the main gear "B". "C" is an auxiliary gear driving power for the pump located just forward of B. The camshaft for the engine valve pushed is at "D". This is on the forward side, with the main camshaft for the intake valve located just to the rear. "E" is a camshaft valve.



A Vought V-101 monoplane engine installed in a Fairchild trainer 1948. "A" is the oil line running from the oil tank in the engine. "B" is the return of the running oil through "C" in the top of the tank. "C" is the lower section of the engine section.



The primary section of the Vought V-101 has a bearing front over with closed type ribs. It is both symmetrically and aerodynamically balanced. Substituted, solid ball bearings are used in the lower "A". Control for the adjustable bearing ribs the rib is at "B".

Stainless Steel Fabrication

By Frank M. Smith

General Manager, Stout Aircraft Corporation

WIPER stainless steel entering more and more into use in the building aircraft today, methods of fabricating and working it are becoming of great interest.

A check of the research laboratories of the various steel companies, and of many prominent industrial concerns working on stainless steel reveals, however, a scarcity of data regarding the fabricating of high tensile austenitic stainless in rolled and formed shapes.

It has been found that methods of rolling and forming familiar to the majority of manufacturers handling stainless steel in the tensile, utilized in the automotive and structural trades have had to be radically altered with the use of stainless, having a tensile over 100,000 lb. p.s.i. Attempts to force the material harder to the design point, resulting in failure through cracking or bending especially in the design joints.

Figure 1 illustrates an improper design for a rib or longeron section where forming to a radius is necessary. A flat top and square corners on the compressed section of the member will assist in failure, while a properly designed section, as shown in Fig. 2, will allow for accurate and satisfactory forming.

For forming light angles in a curve, the most satisfactory results can be obtained by crowning as illustrated in Fig. 3, as the variance in hardness and tendency to spring back make close bending or wiping too inaccurate without constant alteration of the jig, dies or force.

The use of spot welding instead of riveting connections much more exact and accurate joining as proper spot welds are dependent on laying sections both for strength of weld and smooth-

ness of surface. Fig. 4 shows a spot welded structure fabricated on a proper jig, while Fig. 5 shows the results of spot welding over an inaccurate jig. In order to eliminate irregular conditions caused by spot welding this data is however page treatment, the skin should not only be tightly drawn over the structure, but spacing should be done by the dimensional method. That is, the first spots ought to be located at corners and center of the surface to be welded and subsequent spots at the centers of the welded sections progressively throughout. (Fig. 6).

To eliminate the annoying and disturbing noise occasioned by the so-called "oil-popping" of flat surfaces in the structure, rearranging of such surfaces should be resorted to wherever possible. Fig. 6 illustrates a spot weld with the Welder from which is especially critical in "oil-popping" in the wing ribs between vertically, while Fig. 7 shows a cross section with corrugated web where "oil-popping" or noisy leakage is usually eliminated. Both of these angles have been stressed beyond the elastic limit diagonally.



Fig. 1. Improper design for rib or longeron section where bending is a must as in recovery. (Shown vertical)

Fig. 2. A properly designed rib or longeron section. (Top surface)



Fig. 3. Complete solution to the most satisfactory results for forming light angles in a curve



Fig. 4. A spot welded structure fabricated on a proper jig.



Fig. 5. Results of spot welding over an inaccurate jig.



Fig. 6. A spot weld with the Welder from which is especially critical in "oil-popping" in the wings between vertically.



Fig. 7. Cross section with corrugated web where "oil-popping" or noisy leakage is eliminated.

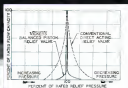
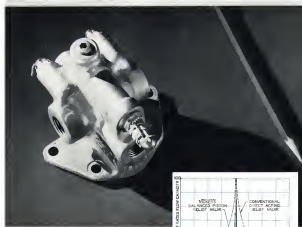


BALANCED PISTON TYPE RELIEF VALVE

Greater Accuracy—Independent of Flow Rate

A hydraulically loaded and balanced piston takes the place of the customary spring-loaded direct-acting relief mechanism. The exclusive and patented Vickers construction means more sensitive operation and greater accuracy throughout the wide pressure range. Variation in pressure from cracking point of valve to its maximum rated capacity is negligible (see chart below). Compact design, installation directly in pressure line, and simple adjustment are other features of this Vickers Relief Valve... precision engineered for the aircraft industry by America's largest manufacturer of power hydraulic devices.

VICKERS Incorporated 1462 CARMAN BLVD., DETROIT, MICHIGAN



IT ISN'T IN THE BOOK



The Crowell Trainair

A new on-the-ground training machine has been developed for teaching beginning students how to coordinate flight controls

A NEW device called the Crowell Trainair has just been put on the market for the purpose of introducing primary flight students with the basic theory of airplane controls. With the consent of the Tech Trainer Society established for teaching instrument flight instructors, the production of an on-the-ground trainer for beginning students is well worth considering.

The new trainer has been designed by John E. Crowell, a pilot of many years' experience. He has flown all sorts of airplanes under all sorts of conditions. He has built airplanes, harnessed in fumes, done aerobics for years, and has taught scores of students—including two Chinese cooks—*to fly*. The trainer which is now on the market is the result of several years of intensive work. Each part was engineered to perform so that the overall effect to the student is that of piloting an airplane in flight.

"We don't expect to teach students to fly in the trainer," said Mr. Crowell.

"Obviously you can't learn to fly on the ground. The purpose of the trainer is to teach students to coordinate the movements of their hands and feet with the attitude they want in the airplane. It is a student's job if he or she is of instruction on the trainer before he gets into the air. He can do what he instructor tells him to do without conscious thought. If the instructor says, 'Get your nose up!' the student's reaction is prompt. When the student is told to do a climbing left turn, he knows how to be on the movement of the stick and rudder. Students who have got in time on our trainer before beginning actual flight instruction feel at home in an airplane. They sit relaxed, grasp the stick lightly, and are able to absorb instruction rapidly."

Details of the Crowell Trainair are seen in the accompanying photograph. The machine is extremely sensitive. It is balanced so delicately that moving the stick forward, or back, whether or not the propeller is in operation, lowers

or lifts the nose. Likewise, movement of the stick from side to side changes the horizontal axis. However, as in an airplane, when the stick is moved from side to side in conjunction with the rudder bar, the action is more rapid. Movement of the stick affects the balance of the weight beneath the pilot's seat.

The motor at the base of the trainer drives a 12-in. wooden propeller. This is by no means a toy, as it produces a real thrust. When the trainer is simulating straight flight, the propeller turns at a horizontal plane. When the left end of the rudder bar is pushed far to the left, the propeller plane is tilted to the left, and the pull of the propeller moves the nose to the left. It right turns the propeller is tilted to the right and draws the nose in that direction.

Students are taught to maneuver the airplane and make precise turns by using a ray spotlight at a wood panel, on which is painted a series of flight paths. Beginning students have trouble in keeping the spotlight on the board. After several lessons they are able to see it where they intend to.

The first lesson is in keeping the spot on the board. The next is in moving the spot from side to side on the horizontal bar by use of the rudder bar, keeping the stick in a central position. Then the student learns to move the spot up and down the vertical line by using the stick, at the same time keeping the rudder bar level. Following this, he learns to coordinate the hand and foot controls by tracing the diagonal paths. The test of the machine is that an experienced pilot can immediately put the spotlight exactly where he wants it.

There are several auxiliary instruction aids on the trainer. The first is the set of safety belts. Wipe the machine

(Turn to Page 122)



John E. Crowell demonstrates his training machine to a group of students at the Academy of Aeronautics. The flight panel, board is at rear left.



Skill FOR AMERICA'S DEFENSE

Recruits in the making—quick to learn, eager to work—much of the future security of America depends on these young men who are building their careers around the sound lesson of coordinated hand and mind.

The skill of Wright's thousands of "old timers," painstakingly acquired through many years of service to the aviation industry, is being transferred into young America to give greater strength and new vitality to industrial progress and to National Defense.

Wright Aeronautical Corporation, Fairport, New Jersey
A Division of Curtiss-Wright Corporation



Consistent with its training program, Wright has increased its production to more than 1,100,000 A.P. per month, with double this figure in 1945.

WRIGHT
Aircraft ENGINES



Wally Timm Aero-Craft

A new trainer designed for secondary training



APROVED by the CAA for secondary CPTP student work, the new Wally Timm Aero-Craft advanced trainer is now in production. Designed to assist the needs of civilian flying schools, the Aero-Craft is particularly suitable for cross-country flight instruction and for landings. Construction is of conventional metal type, with steel tube fuselage and wood wing structure fabric covered throughout, providing for minimum maintenance costs. Powered with a Kinner B-5 engine of 305 hp, the plane has a cruising speed of 155 hr., a service ceiling of 12,000 ft., take-off time at one level, fully loaded, of only seven seconds, and initial climb of 1,100 ft. per minute. Cruising speed is 115 m.p.h. and top speed 160 m.p.h. is the speed indicated, high rate of climb, high service ceiling, and relatively long training stages which adapt this plane as well to all types of cross-

country flight instruction. The stability of the plane is right itself automatically from any position in also a factor in evidence instruction. The plane will also be suitable directly with a Kinner U2-40 engine for advanced flight instruction, or for experienced pilot progression, and can be obtained with side-by-side seating instead of the standard tandem arrangement. Fuselage, wing ribs, and tail group of the Aero-Craft are of steel tube construction fabric covered. The wings employ solid spruce spars and steel type spruce ribs, fabric covered. Upper wing panels attach to the main wings by simple pin fittings and are braced to the fuselage by hinges by means of compression struts. Landing gear is fixed, of rigid beam construction, with compression struts across the main attachment point on the front spar of the wing web, to the fuselage upper longeron.



Span of the Aero-Craft is 35 ft. 5 1/2 in. length 29 ft. 6 1/2 in. and height 7 ft. 2 1/2 in. Wing area is 175 sq ft. and gross weight 2,200 lb. The plane is manufactured in Glendale, Calif., by Wally Timm, Inc., a project of one of the two Timm brothers, Wally and Otto, pioneer flight instructors, plane designers and manufacturers.



They're Rolling at Martin's!—

Rolling in swiftly mounting numbers! The U. S. Army's new Martin B-26's—world's "toughest" bombers (forgiveness)—rolling off America's most efficient equipment for production of bombardment aircraft, Next, (backcross) Martin 167 Attack Bombers for Britain's R. A. F. (remember Toronto?) and the great PBM Patrol Bombers—battle crissers of the air, for the U. S. Navy. Defense is on the wing!

Martin

AIRCRAFT

Builders of Dependable Aircraft Since 1899





The new 1941 Piper coupe.

NEWLY DESIGNED PIPER COUPE

WHILE the external design of the Piper Coupe remains much the same as last year, many improvements in design details and equipment have been included in the 1941 version. Most significant announcements is the fact that the Coupe (as well as the Cruiser) has been equipped with an engine starter as a standard item.



How to work the starter. Twenty to thirty easy turns does the job—then push in.



An under-dashed view of the starter mechanism. The clock end at the bottom (see arrow) goes in the left and end is wound up by the user and clock.

The starter is available in an optional and dual, and semi-optional, but power is derived from a shock cord, cranking through the handle in the end post. To start the engine the pilot needs a removable handle in the instrument panel, twenty to thirty turns, then by pulling the starter forward (see photograph and drawing). It is engaged with the engine, giving it two complete revolutions. Incidentally the whole starter mechanism weighs only ten pounds.

The interior of the ship has been completely reworked. As standard equipment a custom built radio by RCA is mounted on the right side of the instrument panel, with the only part visible being the dial hand and the two tuning knobs. A custom built water meter is available for installing in one of the two oil-fuel compartments in either side of the instrument panel. The tachometer, oil and temperature gauges are one integral instrument while the compass by Corval, is mounted above the panel. Altimeter and air speed gauges complete the group of standard instruments.

For additional electrical power a wind driven generator is included and mounted between the landing gear. This apparatus was custom built for the coupe, with a charge indicator on the instrument panel as well as a rheostat to control the rate of charge. Navigation lights and battery are also included as standard equipment.

Performance has been increased with the installation of the Continental 75 hp motor. It cruises near 100 mph, and lands less than 40 mph. Cruising range is 500 miles with a gas tank of 20 gallons.



The new instrument panel showing the radio at the lower right, instrument cluster, and starter and compass at the top.



Starting device without optional installation in the new Piper.



THE CROSS-ROAD'S ANSWER TO AIRCRAFT BUILDERS—

Monomold... plastic-bonded molded veneer... is the smart way to build aircraft and aircraft parts. *Monomold* smooths out production problems. Aircraft manufacturers are invited to confer with our engineers toward converting metal components for fabrication by the proved *Monomold* process.



BRISTOL AIRCRAFT CORPORATION BRISTOL, VIRGINIA

Manufacturers of Molded Aircraft Parts under the Vield Process

三菱MC20型旅客機

THE MITSUBISHI MC-20 JAPANESE TRANSPORT

With the current interest in commercial versus military priority in aircraft it is interesting to note what developments in design are taking place in Japan

By Paul H. Wilkinson,

Consultant Diesel Aviation

A LITTLE OVER 10 years in the passenger-carrying airplane, and by Japan's standards, the Mitsubishi MC-20 is a modern aircraft. It is the first of a new series of aircraft designed by the Mitsubishi Aircraft Company, Ltd., which has been a pioneer in the design and development of aircraft for the Japanese market. The MC-20 is a twin-engine, low-wing, single-aisle transport aircraft. It is designed to carry 12 passengers and has a maximum speed of 250 m.p.h. The aircraft is built in Japan and is expected to be a success in the Japanese market.

Presently the Mitsubishi MC-20 has been designed for long-range transport of passengers and cargo. It is a low-wing, single-aisle transport aircraft. It is designed to carry 12 passengers and has a maximum speed of 250 m.p.h. The aircraft is built in Japan and is expected to be a success in the Japanese market.

In construction the Mitsubishi MC-20 consists of a fuselage, wings, tail, and landing gear. The fuselage is 25 ft 6 in. long and 10 ft 6 in. high. The wings are 31 ft 6 in. long and 10 ft 6 in. high. The tail is 10 ft 6 in. high. The landing gear is 10 ft 6 in. high.

The MC-20 is a twin-engine, low-wing, single-aisle transport aircraft. It is designed to carry 12 passengers and has a maximum speed of 250 m.p.h. The aircraft is built in Japan and is expected to be a success in the Japanese market.

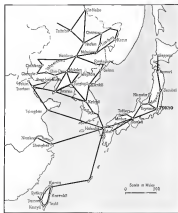
The wings are of all-metal construction and are built in three sections. The outer section is attached to the fuselage and the inner section is attached to the fuselage. The wings are 31 ft 6 in. long and 10 ft 6 in. high. The tail is 10 ft 6 in. high.



Transporter from view of new Mitsubishi MC-20 12-passenger airplane

The wing span of the Mitsubishi MC-20 is 31 ft 6 in. and its overall length is 25 ft 6 in. Its height is 10 ft 6 in. The tail area is 10 ft 6 in. The wing area is 10 ft 6 in. The wing chord is 10 ft 6 in. The wing tip is 10 ft 6 in. The wing root is 10 ft 6 in. The wing tip is 10 ft 6 in. The wing root is 10 ft 6 in.

The power plants consist of two 14-cylinder Mitsubishi R5M air-cooled radial engines which have a take-off rating of 1,000 hp, each and a normal rating of 850 hp. The engines are supercharged and give the airplane a good performance at a cruising altitude of 7,000 ft. The aircraft is built in Japan and is expected to be a success in the Japanese market.



TOYOTA
Flying
EQUIPMENT

The map of the left shows the extensive routes now being flown by Japan Airways, Nippon Airways, and the Japan Air Lines Co. and the Japan Air Lines Co. All these lines are controlled by the Japanese Bureau of Civil Aviation. The map shows the extensive routes now being flown by Japan Airways, Nippon Airways, and the Japan Air Lines Co. and the Japan Air Lines Co. All these lines are controlled by the Japanese Bureau of Civil Aviation.

routes as far north as Sapporo on the island of Hokkaido, west to Matsuyama, Fukuoka and Osaka, and south to the island of Taiwan (Formosa).

Most of the equipment used on the main air routes consists of Douglas DC-3 pre-engineered airliners, Single-engine Doerflinger and other American and European airplanes and are on the local routes. As an example of the time used by using air transportation, it takes 2 days by boat from Tokyo to Kyoto (Seoul) in Taiwan (Korea) but only 3 hours by air. A later route by air will connect Tokyo to Osaka in 2 hours over the 250 miles separating the two cities in less than 50 minutes. In time it is hoped to have an air route between Tokyo and Osaka which will save considerably time between Japan and the United States.

The empty weight of the Mitsubishi MC-20 is 11,000 lb. and its gross weight is 18,000 lb. The aircraft is built in Japan and is expected to be a success in the Japanese market.

The aircraft is built in Japan and is expected to be a success in the Japanese market.

Considerable progress has been made in civil aviation in Japan during the past few years. Part of this progress can be attributed to Japan's policy of keeping up with the times and developing civil aviation in the West. Part of it is due to the facilities in the Orient and the need for rapid transportation of supplies and mail.

Japan Airways, or Nippon Airways Company, operates in connection with the Mitsubishi Aircraft Company and the Japan Airways Company an extensive network of airlines, serving over the Japanese Empire, Manchukuo and North China. All these companies are under the control of the Japanese Bureau of Civil Aviation. Scheduling from Tokyo does not include



Specifications

Wing span	74 ft 6 in
Overall length	25 ft 6 in
Height	10 ft 6 in
Wing area	780 sq ft
Wing empty	11,000 lb
Loaded weight	18,000 lb
Maximum speed	250 m.p.h.
Cruising speed	195 m.p.h.
Stall speed	1,000 ft/min
Service ceiling	10,000 ft
Top engine power	1,000 hp
Wing loading	24 lb/sq ft
Power loading	0.8 lb/hp



The Stinson Voyager

A 90 Horsepower Three-Place Light Plane



The bridle wing sticks on the Stinson Voyager are designed to make the plane spin-proof and together with the wing flaps to achieve landing speed.



The windshield is re-designed for better visibility. The instrument panel has been relocated at the bottom center of the instrument panel and is easily reached from either side seat.

THE STINSON VOYAGER, a new 90-hp three-place airplane for the private owner has been announced by Stinson Aircraft, Division of Vultee Aircraft, Inc. The new plane, equipped with electric starter and generator, is an improved design of the famous Stinson 108, which has been loaded for its safety record during the last two years. Much of this is due to its built-in ailerons and flaps, which have been retained on the new Voyager model for 1941.

The Voyager has a top speed of 115 miles an hour. It is a most powerful model, then for 90 and standard equipment includes a 98 hp Franklin engine, hydraulic brakes, parking brake, wheel pants, detachable tailwheel and landing gear. The engine develops 90 hp at 2500 r.p.m. and runs a large diameter propeller. Take-off was with full load for the Voyager is reduced to 580 feet while the rate of climb has been increased 28 per cent up to 610 feet a minute. Service ceiling is 11,000 feet.

The new instrument panel has provision for radio installation while all control buttons and switches have been rearranged for ease of operation. The interior of the ship is done with luxurious Lindbergh broadcloth. The cabin doors have been installed for greater cabin quarters while the contour of the windshield has been changed to minimize distortion. The new master switch is of the honey-duty knife type and has been relocated in the center.

EQUIPMENT

90 hp "Franklin" Engine
Dual Magneto Ignition
Automatic Valve Lubrication
Hydraulic Valve Tappets
Camberator Motor
Metal Taper Propeller
Starter Battery, 6 volt
Semi-automatic Shock Absorbers
Hydraulic Brakes
Parking Brake
Full Barrel Tail Wheel
Full and Needle Bearing Controls
Flaps, S.A.C.A. Induced Twist
30-Gallon Fuel Tank
Dual Wheel Controls

INSTRUMENTS: All instruments linked. Compass, airspeed indicator, altimeter, vacuum fuelometer, oil pressure and temperature gauges, electric fuel gauge.

WEIGHTS

Empty weight (including oil) 940 lb.
Gross weight 1,100 lb.
Fuel capacity 30 gal. (40 gal. max.)
Luggage and Extra 20 lb.
Gross Weight 3,615 lb.
*Subject to plus or minus 2% variation.

PERFORMANCE

Full Load

Maximum Speed at Sea Level 115 m.p.h.
Landing Roll 150 ft.
Rate of Climb 600 ft./min.
Service Ceiling 11,000 ft.
Turned Run at Sea Level 530 ft.
*Subject to plus or minus 2% variation.



THE FLEETWINGS KET-12 fabricated from Stainless Steel, offers a new source of supply to National Defense Training.

In the construction of Stainless Steel, time-consuming grinding is replaced by spot and seam welding, making a superiorly adaptable in mass production.

Fleetwings is also manufacturing, for the Aviation Industry, Stainless Steel and Aluminum Alloy Tons and Movable Barrels, Hydraulic Valve Equipment.

Special Parts and Fabrics, also including Precision Machinework and dependable workmanship.

Fleetwings' greatly enlarged facilities are geared to meet the demands for dependable mass production delivery schedules.



FLEETWINGS
INCORPORATED

NEW YORK, N. Y. DENVER, CO. TAMPA, FLA.

LEADERSHIP

in Speed!



SPEED—the very essence of the growth of air travel. Speed—often the vital difference between life and death in a fighter. Speed—that allows a peace time transport to be transformed into a famous bomber.

"Lockheed engineered" stands for that kind of speed. And it's good business when you realize that greater speed permits longer flights during daylight...cuts working time of flying personnel...allows more revenue miles per flying hour...and more miles between overhauls.

But just speed is not enough! It's speed with dependability...speed with all 'round performance...and speed with economy that has made Lockheed commercial transports, the nation's fastest, famous the world over. And now from that great reservoir of experience in "engineering for speed" has come the Lockheed P38, a fighter in that distinguished 900 mile per hour class...and a capable defender of democracy.



Speed with safety is a Lockheed event! That's why the most effective, high-altitude device known to modern aviation...the Fowler Flap, was adopted and developed, then installed on that Lockheed transport and military airplanes. The above (this twice) shows one of the flaps in extended position for landing.

It can't take a Lockheed to beat a Lockheed! In 1936 a modified Lockheed 14 shattered all records for transport airplanes, by flying across the continent in 10 hours, 33 minutes and 59 seconds. On November 2, 1940 a newer Lockheed, the Lodestar, clipped 1 hour, 6 minutes and 20 seconds from this unbeatable record.

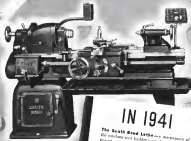
LOCKHEED AIRCRAFT CORPORATION • BUREAKE CALIFORNIA, U.S.A.

LOOK TO *Lockheed* FOR LEADERSHIP





The Maryland Lottery—disputed and under
 Enclave by Henry Mendez, Esq. Considered the
 first printed work containing letters from the
 who all words in the same words have been developed



The South Bend Letter—a masterpiece of the machine and ladder art. A teacher took Edward's notes, reeling previous letter—designed and built in three previous days shop require means for picture, speed and in color.

IT IS an international tradition that the United States leads the world in the production of fine machinery. This enviable American accomplishment may be attributed to efficient management, capable engineering and modern machine tools. We take pride in presenting the new Series "S" South Bend Lathes for 1941—our contribution to American Industry—a modern machine tool.

Aviation—
RADIO

The audio frequency amplifier meets the requirements of voice intelligibility, 300 to 3500 cps within plus or minus 2 db, distortion less than 10 per cent, 0.3 watts audio input across 100 ohms driving impedance to produce 100 mW into 100 ohms. The circuit is a single stage of beta-generi triode scheme employing a 6J4 grid-coupled triode, two 20K μ f capacitors, two 250 μ h inductors, and a 6VS audio amplifier. The 6J4 triode is also used as a heterodyne oscillator for the associated receiver indicated by a center on the panel, which is connected by a rotary switch to indicate no external source and current; current, p.a. grid current, p.a. plate current, modulation plate current, filament voltage and receiver high volt-

The workable receiver, mounted in the same housing with the transmitter, offers ten channels tuned automatically, by the motor drive, with the transmitter channel selector. The tuned circuits are contained in an easily-reversible rotating turret. The sensitivity is two microvolts at 4-10 Mc/s; noise level is 12 dBmV. Selectivity is 3 dB down at 41 kc, 50 dB down at 35 kc. I-F and audio responses are all better than 60 dB. The audio characteristics are approximately the same as those of the transmitter, plus or minus 3 dB. The 1000 Mc/s power output of the 800 Mc/s driver is available from the 800-Mc/s output.

An over-voltage limiting device placed in the initial π -d stage permits operation with high inputs.

New U-H-F Airport Transmitters

Developed primarily for airport traffic control, but useful in a variety of non-communication situations, this 100-watt transmitter is controlled by the new Radio Receiver Corp. of New York Type 435 and 437A are 50-watt versions of the unit, covering the range from 118 to 132 and 140 to 144 Mc, respectively, with frequency stability at plus or minus .001 per cent. These transmitters employ a 676 crystal oscillator, two 6N7's and an ALE as frequency multipliers, and an 829 as final amplifier. A 655C acts as output audio amplifier followed by a 6L5 modulator. A 6AL5 diode acts as a generator table for metering from the power line. An output of 60 cps, 130-150 volts. The audio frequency response is the (from 100 to 4000 cps, with 3 per cent distortion).

The power output of the type 428 is 125 watts, over the range 125 to 142

Ariace-Bendix Airline
Transmitter-Receiver

The Bendix Radio Corp. of Baltimore has released details on the model RTA-61 airline transmitter and receiver designed cooperatively by airline engineers working through Aeronautical Radio, Inc. This equipment is of particular interest because, under the terms of the contract between Bendix and Arco, engineers of both organizations consulted cooperatively with airline engineers to produce the final design. The transmitter has a 100-watt output.



Readily available in paperback for \$14.95



Top view of the STM. I



Harvey Wells induction reactor

Especially designed for light aircraft use, a new radio receiver has just been placed on the market by Harvey-Wells Communications Inc., of Southfield, Mich. Said to incorporate the latest advances in aircraft radio engineering, the design is based successfully on actual requirements of private aviation as found through long experience of Harvey-Wells personnel. This compact model, Type NR-3-A, 4-tube radio weighs only 9 lb. complete with heavy duty handset, battery case, cables and headphones. The receiver cabinet contains only 4x6x12 in. and is furnished with Universal shock mountings so that it may be mounted in any number of positions in the ship.—*Aviation*, March, 1941

A new 100-psi positive pressure pump has been announced by Eastern Engineering Co., New Haven, Conn., makers of a growing line of submersible pumps. Useful for any application in which location or tools and containers make well pumping essential, the new pump, designated Eastern Model "U/P", has a maximum pressure of 20 ft. psi and a maximum volume of 5 gal. per min. Of rotary type, it is 6 1/2x2 1/2 in. and weighs 25 lb. Power is supplied by a Universal flat-coated motor operating at 4 or 6 in. 1/4 light weight and small size make it particularly adaptable to all limitations regarding the pumping of thin liquids where weight and space control must be kept at a minimum.—*Aviation*, March, 1941

One of the most interesting of recent developments for speeding up watch production is a new process called Photo-Sensitized Metal made by Republic Engineering Products, Inc., 460 Lexington Ave., N.Y.C. The new, patented, aluminum-base metal is said to be easier and more convenient than paper or cloth, in the shop, and ideal for making exact copies of photographic plates and inspection sheets. The usual steps of development and etching following the processing of ordinary film is recommended. Distortion, due to unequal expansion and contraction commonly present in sensitized film, paper or tracing cloth, is, reportedly, eliminated. Runs up to 10,000 ft. can be supplied on special order.—*Aviation*, March, 1941

Turners and chronographs are always useful in aviation and when an instrument performs just important functions, it assumes the nature of indispensability. The new double-lens Vitelectron, Style No. 220/DT, recently introduced by John Kroeck & Co., 29 W. 47th St., N.Y.C., can be used as a timer, watch, tachometer, and a altimeter in various outdoor or light and sound. Equipped with a 45 sec. register, the new Vitelectron model offers a time-out feature. Pressure on one button consecutively starts, stops and again starts the sweep second hand. Pressure on the other button brings the sweep hand back to zero. The mechanism is a 13-jewel Gold movement.—*Aviation*, March, 1941

A bright future seems to be in store for plastics, which is rapidly becoming a major industry. Latest development is the perfection of a new process for making plastic molding, strip and tubing rolls in continuous lengths. Detroit Masco Corp., Detroit, Mich., in announcing the new method, report that their extruded plastics are obtainable in a wide range of colors in either opaque, translucent or transparent form. Advantages claimed for the Detroit Masco process are economy, smooth finished finish, low heat conductivity, and imperviousness to changing climatic conditions.—*Aviation*, March, 1941

Two new additions to its "Multi-size" drill line—sizes B0 and B—have been announced by Ingersoll Rand Co., Philadelphia, Pa. B. Extremely light, ranging from 14 to 25 lb., numerous attachments can be furnished to adapt these tools



Eastern Model VT Hinge Pump



Photo Copy in Aluminum



Double Lens Vitelectron



Drill Masco extruded plastic



Ingersoll Rand "Multi-size" drill



Vertical Drilling Machine



Small Platform Scale



Hand Tool



Hand Tool



Hand Tool



Hand Tool



Hand Tool



Hand Tool



Hand Tool



Hand Tool

for light wire drilling, not running, close-quarter drilling, wire brushing, sanding, etc. Straight, lever handle, or pistol-grip handles are available.—*Aviation*, March, 1941

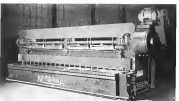
The hammer, get the varieties in National Defense. Soder Precision Tool Co., Jackson, Mich., are manufacturing model SPB11112, made of a series of precision planning hammers. A high speed machine for finishing sheet structural shapes, the hammer also makes an efficient riveter for light covering by installing the proper shaped dies. Having a speed varying from 4- to 2000 revolutions per min, an angle can be adjusted from a better work to a heavy heavy enough to shape and form 18 gage steel. A foot control permits the operator to start and stop the hammer instantly, power and speed control valves are located directly over the cylinder. An important feature lies in the fact that the hammer breaks and die support blocks can be raised or lowered to meet operation requirements.—*Aviation*, March, 1941

How to get into a tight corner is illustrated by The Pate spread production drills made by Searp-on Tools Corp., Kenosha, Wis. These light, compact, all ball-bearing tools of 5 to 7 in. drilling capacity have an overall length of 2 1/2 in. and a body diameter of 3/8 in. A grip handle is centered below the body of the drill for easy handling. Helical flutes are precision-machined for easy operation and long life, making them essential for the 4 or 6 in. to 60 cycles, with a ball load speed of 2000 r.p.m. for the 5 to 7 in. drill and 1200 for the 3 in. drill. Special speeds and ratings are also available.—*Aviation*, March, 1941

A light and simple hydraulic system for the operation of auxiliary equipment, such as lifts and loading gear, is manufactured by the Duplex Equipment Corp., Long Island City, N.Y. Called the Duplex "Lander" engine driven pump, the unit is designed to incorporate its own pressure regulator and can develop high operating pressures, a combination which is said to make it of particular value for portland ships.—*Aviation*, March, 1941

Newest arrival in the family of streamline continuous Blawiegan Machines, made by L. F. Blawiegan Co., Chicago, Ill., is model "25-10". The combination consists of a model "25" feeder in conjunction with a model "10" washer and dryer. The "25-10" produces uniform, continuous prints with clean, sharp lines at an exceptionally low cost per sq. ft.—*Aviation*, March, 1941

Addition of a new power squaring shear, No. KL-18, having an 18-inch cutting length and capacity of 5 in. mild steel is announced by Niagara Machine & Tool Works, Buffalo, N.Y. Equipped with a quick-setting, self-mechanizing ball-bearing, parallel block gage, adjustable in increments of 1/16 inch, the shear is controlled by means of a 15-point clutch operating in a half of a sec. All gear drives are mounted on anti-friction bearings, and enclosed in a light case. Single operating, independent spring pressure foot hold-down provides a smooth and firm grip on sheets of varying length.—*Aviation*, March, 1941



Power Squaring Shear

Kelly Field to San Angelo, Texas, is being sent. Another detachment will be approximately 500 reinforced men and one aircraft. The 50th Wing has a total strength of about 1,000 officers and enlisted men.

NAA Liquidates Debt Improves News Letter

National Aeronautics Assn., under the leadership of Col. G. E. Foyent Langer and Col. G. H. Wilson, is taking an increasingly active part in the public affairs of the nation. The association's interest in receiving consideration as all the federal and other agencies can receive with interest and flying. News and better advertisement to print into the regular Washington News Letter, which is now edited by Kendall Hoyt, an experienced Washington journalist and aviation writer. Since June 1949 the NAA has paid \$2,500 in debts and has a well-stated balance in the bank. All employees have had increased salary increases of 10 to 30 percent. Mail postage per month increased also from June 1949 to 1948. Membership increased 20 percent.

PFA Plans Peace Day

"P Day" is Peace Day, as the Pacific Flying Assn. puts it, and that day will bring trouble, the officers believe. A string of declassified photos will come, a mail-palmed market for supplies, but they will meet up with as much lawlessness, racketeering, and organized crime as they will find in the average photo and after two or three more, even more trouble. The association believes that a complete support system, for the air force only, will add to their troubles. But it proposes an air day on the policy (1) A practical "United Front" with other interested associations; (2) support for a "Civilian Air Reserve"; (3) creation of a "P Day" plan for service flying; (4) endorsement of the PFA Bulletin as an adjunct to the three above objectives.

27,037 Hours Afloat

Latest reports on pilot training at Randolph Field state that 650 flying cadets are getting 70 hr limit each in the basic basic training course. Flying time limits at previous schools when Randolph Field planes averaged almost 4 million miles in 27,037 hr. during the month.

Calling Names

W. A. Patterson, chief of the Air Force, is expected to visit the base in 1949. Patterson is expected to visit the base in 1949. Patterson is expected to visit the base in 1949. Patterson is expected to visit the base in 1949.

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I. Ae. S. Honors Given at New York Meeting



GLENN I. MARTIN received the distinguished Donald Guggenheim Medal for contributions to aeronautical development and the production of many types of high performance aircraft.



MRS. BELLA G. LANDAUER of New York City who has generously sponsored the Institute of the Aeronautical Sciences with a new collection of aeronautical art and models.

C. H. SPENCER, American Airlines, received the Guggenheim Medal for his contributions to the development of the Douglas C-47, the world's most successful transport aircraft.

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Flying Fortresses.

GREATER THAN EVER

● Possessing a Boeing tradition, and long before national defense assumed the vital importance it has today, Boeing developed that mighty monarch of defense—the 4-engine Flying Fortress, peak of the U. S. Army Air Corps. The basic idea of the Flying Fortress was to be in advance of its time that today's vision of the airplane—the world's fastest long-range bomber—is the B-29 Superfortress—each model an improvement on its predecessor. Now being delivered to the Air Corps is a fleet of B-29 Flying Fortresses, new under construction in the expanding Boeing plant in a fleet of B-29 models—incorporating all greater advancements. Thus, at a time of national emergency, Boeing, with unequalled experience in building 4-engine aircraft, is ready with the proven design, background, equipment, and confidence to speed the delivery of new Flying Fortresses—America's standard bearer of leadership in military aircraft.

Boeing has always built tomorrow's airplanes today!

Boeing

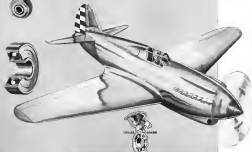
AIRCRAFT COMPANY
SEATTLE, WASHINGTON

DR. HUGH L. GAYSON, Chief of the Mechanics and Plans Division of the Bureau of Standards, awarded the Boeing Aircraft Award for contribution to aeronautics. Other winners were pictured in last month's issue.

FEDERAL

Aircraft BEARINGS

QUALITY and PRECISION ball bearings are essential to the National Defense Program. The FEDERAL plant -- working to capacity -- is producing its full quota of fine ball bearings to meet the constantly increasing demands of the aircraft industry.



THE FEDERAL BEARINGS CO., INC.

Makers of Fine Ball Bearings

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Two Federal Finance Plans for Defense Plants Aircraft Manufacturers Seem Sky of Ownership

Washington (AVIATION DEFENSE)—Defense Plant Corp., an RFC subsidiary, is moving its operations forward as the advocate of financing plant expansion—displacing the Emergency Plant Facilities method developed by the Defense Committee. Outstanding examples are Ford's \$100,000,000 plant for Pratt & Whitney engines. Usually announced as EFP contracts, these are now being handled by Defense Plant.

The two plans are similar in that under both the Army or Navy pays for the plant and five years, and the plant, and will then sell the plant to the manufacturer if he wants it. The EFP contract, however, provides that the plant be initially financed by bank loans, that the manufacturer hold title during the five-year repayment period, and that the government make a few contracts to meet the 48 weekly payments (which means that weekly appropriations must be set aside for the program).

Under the RFC arrangement, Defense plant loans held during the five-year period, and now the money to build from RFC. The manufacturer runs the plant as if a new plant loan and upland financing, even RFC is more willing than a private bank, Army or Navy need not send money before weekly give to the Defense Committee to appropriate for the installment payments as they come due.

For this reason Army and Navy prefer RFC to EFP financing. The latter also has the benefit of freedom which might mean that money can come from the Treasury.

Major aircraft plant expansion jobs are being handled by Defense Plant Corp. (includes Bendix Aviation, \$25,000,000; Consolidated Aircraft, \$14,475,000; BSA, \$10,000,000; North American Aviation, \$10,000,000; Wright Aeronautical, \$10,000,000).

Three Airship Bases

Construction of three inflation-draw airship bases is planned by the Navy because Congress prohibits the Army in the future from adding Navy appropriations.

plans", in the opinion of speed, and, Col. James Turner, who proposes that the government buy up or lease all available land, earth, gravel, slivers, and here private firms to form an aviation home defense corps.

He contends that such plants could be effectively converted to military use and made capable of dropping a 100-lb. or two 10-lb. bombs from 1,000-1,500 ft. that production of these little ships could easily be stepped up to 200,000.

COL. HOWARD H. EDWARDS, newly appointed commander of Fleet of Southern Fleet, who came 1,000 miles of flying time in his big ship, the Grizzly, the machine as landing.

Latin Air Bases

Sen. Admiral John W. Gravelle, who headed the joint Army-Navy board that selected base sites in British Atlantic islands, stepped for two years of military or naval bases, revealed that the United States expects soon to reach agreement with Great Britain for American air and naval bases. He said we are ready.

FIRST NORTHROP PATROL BOMBER at start of flight from Little Ferry, N.J., in General with a 1000 hp. Wright, has a normal range of 1145 miles.

Cylinder Engine

President E. T. Keller of Chrysler Corp. announces the company's engineers have developed a 12-cylinder V-type liquid-cooled engine capable of turning out nearly 2000 hp. power. It was reported that the engine would use special high-octane fuel rated at 110 octane. The engine usually runs at 1000 to 1200.

Napier Spher

British Napier Spher engine is being considered for US production by Army and GPO. The Napier Spher is a 12-cylinder, 1000-hp. engine with low tanks of six cylinders. It has a low fuel consumption, but the engine will have less than 1000 hp. It was reported that the engine is liquid-cooled, but this was not certain. The high power rating is obtained by mechanical supercharger due to continuously high speed at 2800 rpm.

Army and Defense officials are considering the new engine project that show promise, but in each case they are taking into account the large problems of long delays that are sure to follow.

1000 Young Men

Secretary Knox of the Navy wants 1000 young unmarried men, age 20 to 27, with an interest in aviation or naval aviation. Successful applicants will be selected in Navy Reserve for training in the United States. Successful applicants will be appointed aviation cadets and assigned to the training centers. Navy will operate three group facilities: Pensacola and Annapolis, Pa., and Corpus Christi, Tex. Candidates may apply at any of the 118 Navy recruiting stations in the U.S. There are 12 Naval aviation reserve stations.

George B. Warner is now vice president, charge of engineering for Federal Aircraft. He formerly had technical jobs with Westinghouse, Capt. A. J. Foster, senior president of Macgregor and Kresner, is in charge of the Federal Aircraft, which will build glider-bomber ships.

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AVIATION, March, 1942

AVIATION, March, 1942

11

Your BEST INSURANCE Against

Speed-killing "Doubtful" Screws



Guarantee OF HIGHEST QUALITY

To insure satisfactory performance, Parker-Kalon Fastening Devices are made in leading speedshops covering dimensional and physical characteristics which are checked at every stage of manufacture, and passed upon by Parker-Kalon's Quality Control Laboratory. Thus, this greatest price factor in any way replacement will be made without charge.

PARKER-KALON CORPORATION
NEW YORK, N. Y. U.S.A.

A Guarantee that's backed by Parker-Kalon's Unique Quality-Control!



Only a handful of "doubtful" screws may lower the quality of a full day's output of a product. Besides, consider how much time is lost in locating properly cut slow production and lower up costs. That's why Parker-Kalon's famous "doubtful" screws—screws that look all right but of which a percentage fail to make satisfactory fastenings—is vitally important to you. This Guarantee is your insurance that the Parker-

Kalon Screws you use less cost the vital tests imposed by the Parker-Kalon Quality-Control Laboratory. From raw material through the various processes of manufacturing, inspection and tests guard against "doubtful" screws.

Specify Parker-Kalon and get Fastening Devices you can depend on to live up to the performance promises made for them. Parker-Kalon Corporation, 180-184 York St., New York, N. Y.

SOLE ONLY THROUGH RECOGNIZED DISTRIBUTORS

Quality-Controlled PARKER-KALON Fastening Devices

Subcontracting Favored to Speed Production Organization of Pools Spreads Employment

Washington (AVIATION)—Subcontracting as a means of defense contracts is rapidly becoming accepted here as the answer for expanded production that standardization of products tends to slow this year. However, subcontracting is the current emphasis will go further than that that short-term program.

Larger scale examples so far is the plan to have subcontracting plants make engine parts on subcontract. These are extremely slow in getting underway, the plan has included the point where a fairly detailed program has been worked out by which the big three—Ford, Chrysler, and GM—would make parts for the same 500 engines a month, then select sub-contractors for some of the work. This will double the number of those firms able to experience, and there is already talk of drawing the smaller firms into a subcontracting program on engine and power plants. Meanwhile Douglas has independently made contracts with 2000 fully equipped engine and power plants.

Behind the push for forming out are two basic motives. Growing emphasis in the administration on the necessity of faster cash deliveries to Britain leads to a desire to utilize every manufacturing capacity of the country—even at the expense of the greater economic security and lower costs which can be achieved by the slower process of building new plants.

A second objective is to spread the employment resulting from defense orders over the country rather than concentrate it in a few armaments base towns. It is pointed out that 98 percent of the prime contracts in the here come to 110 firms. It was this which led Hillman's administration to take up the idea last fall and out there. Goals to working on it.

Goals is still at work expanding the list of small firms which can subcontract as a test and passed out the work among their members. But meanwhile the services and OPN have asked on the side as a prebidding device. And Navy has now urged their contractors to subcontract as much work as possible, and QPM's production division has stressed a Defense Contract Service to give technical help in this.

Helping up Defense Contract



BUICK WILL BUILD A \$10,000,000 aircraft engine plant at Melrose Park, Chicago to manufacture B-26 engines.

Sign. New plant will be under government contract for production of B-26 1200 hp engines and will employ around 10,000 men with an estimated annual payroll of \$17,000,000. Plans call for more than 1,000,000 sq. ft. of manufacturing space in which all parts and subassemblies will be produced and engines assembled. In addition there will be 24 test cells for engine testing, power house, personnel and administrative buildings. Army contracts for aviation engines and parts will total about \$10,000,000.

Progress Aircraft graduates in building a new \$11,180,000 plant, government-owned, at South Plain, N. J., for manufacturing engine parts.

Recent Awards

- Harry Engelhardt**
- Boeing Aviation Corp., Detroit**—\$1,343,036 for aircraft work.
- Brewster Aeronautical Corp.**
- General Motors Corp., Detroit**—\$1,071,320 for Pratt & Whitney engines and parts.
- Continental Motors Corp., Inc.**—\$1,010,110 for engine and parts.
- Air Controls, Inc., Chicago, N. J.**—\$64,104 for cylinders.
- Boeing Aviation Corp., South Plain, N. J.**—\$11,180,000 for aircraft engine parts.
- Westinghouse Electrical Instrument Corp., Newark, N. J.**—\$14,000 for dynamometer indicators.
- Paychick Aviation Corp., Jamaica, N. Y.**—\$107,400 for pistons.
- Boeing Aviation Corp., Idaho, N. Y.**—\$118,775 for engine assembly.
- Yale & Towne Mfg. Co., Stamford, Conn.**—\$151,604 for fuel pumps.

PRODUCTION MODELS of the Bell P-39 are now coming off the assembly line at Buffalo. Three pilots from Republic Field, who never before had seen this ship, take delivery and in a few minutes were doing take-off and landing flying.



AIR CHUTE

Symbol of Safety



Standard equipment of all government air fleets and used by commercial companies and private fleet operators... the only parachute to today reflect the Symbol of Safety.

The Famous "SPARTAN" School Prefers "IRVING"



Only the Irving Air Chute Co. has been awarded the highest contract award in the world for the production of parachutes for the U.S. Army, Navy, and Air Force. This award is given to the manufacturer of the best parachute in the world.

Only Irving gives you the "Symbol of Safety."

Our Engineers are at Your Service

IRVING AIR CHUTE CO., Inc.

1670 JEFFERSON AVE., BUFFALO, N. Y.

Factories at: Buffalo, N.Y.; Glendale, Calif.; Fort Lee, Canada; Lancaster, England; Redwood, California; London, England; and other countries.

AVIATION ABROAD

On Schedule

By "Rita"

Last month the columns South Daily extensively on the fact that for the moment the Pan American Airways Service was having no future plans on large landplanes, and that an announcement could be expected shortly that a number of large landplanes in the 30-ton class had been ordered. Upon his return from London on the first survey flight of the new Westbound route of the Trans-Atlantic Clippers, Mr. Juan Trippe made a statement which bears out the correctness of this column's prediction, except on fact that each plane had been ordered, but that for definite reasons he was not in a position to give further details.

Interesting air trails, it should be noted, a new plane, in which we shall see landplanes crossing the ocean at high altitudes at far greater speed and with a greater degree of regularity than can be attained at the moment with flying boats. Undoubtedly the experience of the British, who are experimenting to obtain T.W.A.'s big Boeing Stratojets for a continuous ferry and diplomatic line between England and Canada, will be invaluable, and prove that landplane service of the ocean is very well possible.

In 1932, when the first Stratojets were in construction, there were plans that Pan American Airways would use three on a line to England, and possibly to Australia, when at that time the only line facilities for American planes were available at England. The war, and the tremendous development of the Latin American routes made it necessary that these airplanes, when they were finally delivered, were used on service to Buenos Aires and the Canal Zone, where need for faster, better connections was urgent.

In Canada, the C.P.R. is steadily acquiring more and more of the mail, independent air services in the Northwest Territory, will apparently plans to eventually establish an alternate service to the Ocean via Alaska and Siberia. Such a route would thereby form the last link in the network of all English airlines around the world, a goal toward which British lines have been working for years, and which was almost some years ago by American refusal to let a foreign airline use Honolulu as a base.

At the same time, Northwest Airlines is going ahead with its plans to connect the Midwest with Europe, a service which has long been needed. They propose a 340-hour service from St. Paul and Minneapolis to Stockholm, via Fargo, Edmonton and White Horse, thereby making it possible to reach Alaska within 24 hours from New York.

The British are now operating 95% of their pre-war air mileage, by the latest reports, maintaining connections between London and the Empire via London, where the R.E.M. DC-8s, under charter to the British, are now a familiar sight. From there, the service swings the South and then East across the water of Africa in Egypt, where connections are made with lines down South Africa, the Near and Far East and Australia.

American Export Airlines' transatlantic project involved a network where the Roman Agreements committee struck out \$1,250,000 from the Post Office bill, providing for mail payments. The argument was that the cost of new AEA service, as compared with proposed additional FAA trips, was not justified. But the cut still was to be passed on by the Senate so this was written. Meanwhile it was assumed that if AEA was denied the mail subsidy for any considerable period, Company would concentrate on its Latin American projects.

Belgium to marry, the Italian airline LATI has been maintaining a landplane service between Rome and Rio de Janeiro during the past year and a quarter. This involves new transatlantic stoppages, including connections between London and Italy.



THAT EXTRA SOMETHING

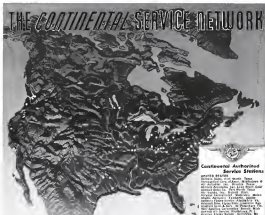
• Distinction in an individual may be one of a number of things... refinement, character, breeding, poise. In a Dumore motor it's the combination of a lot of little extras... refinements in design and construction... care and inspection in manufacture... that impart extra hours of productive power. That Dumore motors contribute extra value to business machines, medical devices, household appliances, portable electric tools or other equipment they are called upon to power. In some instances Dumore furnishes complete units... in others, where a manufacturer makes his own bearings, Dumore "mated" motor parts are supplied, for they, too, have "extra power hours" built into them. So, you see, however peculiar your power problem may be, if it's a fractional horsepower motor you need, Dumore engineers can solve it. Write for facts... no obligation.

THE DUMORE COMPANY • Dept. 381-E • RACINE, WIS.

*Illustrated on page 8 Motor Parts, 1/2 H.P. of 2,000 r.p.m. in operation at 100 volts, 540 cycles.

DUMORE MOTORS





Supplying Authorized Service to Red Seal Owners Everywhere

Merely to drive to the hum of a Continental Red Seal Engine is to sense the reliability of its driving power. This helps to explain why buyers of planes specify Red Seal Power, to study an exhaust, that more of these famous engines are powering light aircraft, than all other makes combined. There are thousands in daily use and you will find them in *Aviation* — Culver Cadet — Jivacop — Interstate Cadet — Luscombe — Piper Cub — Powerfield — Reerain — Stinson — Taylorcraft — Boniflow — Welch.

Dependable Red Seal Power is constantly being made more significant to the expanding sector of aircraft owners through a service network that is conveniently on call throughout the continent. The reliability of this service is a perfect complement to the dependability of Continental Power.

Red Seal Power is Dependable Power — on the job — EVERYWHERE.

Continental Motors Corporation
Aircraft Engine Division
MUSKEGON, MICHIGAN

Continental Authorized Service Stations

Continental Motors Corp. has a nationwide network of authorized service stations for its Red Seal engines. These stations are located in all major cities and are staffed by trained mechanics who are familiar with Continental engines. The network is designed to provide prompt and efficient service to owners of Continental-powered aircraft. The stations are located in all major cities and are staffed by trained mechanics who are familiar with Continental engines. The network is designed to provide prompt and efficient service to owners of Continental-powered aircraft.

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In recognition of his services as Administrator of Civil Aeronautics, COL. DONALD H. CONNELLY has been promoted to the rank of Brigadier General, GSC. Connelly is also a member of the N.A.A.A.



EARL M. KALTWASSER, made general manager of Illinois Division of Vultee Aircraft, has been identified with Vultee products, and will direct operations of the Vultee, Moline, plant.



Turning out Altimeters for the Army is SAMUEL S. BRADLEY, who is president of the Altimeter Co., but he is also a newly elected director of Sterling Engine Co., builders of the new Altimeter engine.



To help speed plane production, LUTHERAL has placed DR. V. M. KRAVCHAK, metallurgist, at the head of its new laboratory. Dr. KRAVCHAK came from the Soviet Union.



At Bendix Aviation Corp., VINCENT ESCH (above) was installed president, and Charles Marcus vice president in charge of aviation operations. Board of directors was re-elected in its entirety.



All the Edward's of Canada President Division, R. ELMER DIXON (above) succeeds George C. Rodgers as sales manager of the Mott, formerly 1934-1935. The company is now president of the Mott, formerly 1934-1935. The company is now president of the Mott, formerly 1934-1935.



As partner of Burns & Co., R. ELMER DIXON (above) succeeds George C. Rodgers as sales manager of the Mott, formerly 1934-1935. The company is now president of the Mott, formerly 1934-1935.



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Newly appointed controller of American Aircraft Corp. is O. S. GOODWIN, formerly with S. D. Lusk, and Co., during the World War he was chief accountant on several government projects.



Chrysler Corporation's Warren Ave. plant, which will be used for the manufacture of airplane parts, will have as operating manager O. A. BARNES, former operating manager of the Dodge Main factory.



Additions to the sales staff of Lear Corp. is TOM STANLEY, for 11 years superintendent of aircraft parts and maintenance in the United Air Lines. He will specialize in sales of air-line equipment.



To better coordinate its operating activities, United Air Lines has appointed to its eastern and western sales offices, appointing RUSSEL P. ARNOLD and HOMER J. MERCHANT sales men and western sales managers respectively. ARNOLD was formerly with the United Air Lines, and MERCHANT was formerly with the United Air Lines.



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Houghton Wins Lesley Award

The newly established Robert M. Lesley Award, in honor of Capt. Robert M. Lesley, killed during an air raid in Norway, while attached to the U. S. Legion as a military ob-



server, was presented to Harry G. Houghton, Jr. (above), Assistant Professor of Meteorology at R. I. T. in 1958. Houghton, awarded by W. H. Goddard, developed a method for clearing fog over a limited local area by use of a sodium chloride rocket, and has recently extended his research into the exact nature of atmospheric condensation processes in general. Presentation on behalf of the Institute of the Aeronautical Sciences was made on March 21st by Capt. F. W. Roubalievitch, Chief of the Weather Bureau.

Funds for Research

A budgeted fund of \$400,000 for NACA had been reported by the House Appropriations Committee of Congress as this was written and was headed for quick passage with the Independent Office bill. This sum, for fiscal 1961, is \$1,000,000 more than the 1961 appropriation, nearly half of which is to provide for additional personnel. The new personnel are mostly with the new laboratory at Moffett Field, San Francisco, Calif. The three new laboratory will not be in operation for a year or more. NACA research is ranked as one of the most vital functions of the Government, with no such money being spent on air craft, and with air war being barely a constant in design for performance.



Ultra-High Frequency

Looking forward to state-of-the-art in installing 10 experimental instruments along the New York-Chicago route. Night will be in operation by mid-1960, and there will be seven-armed equipment has already been tested at Indianapolis, Philadelphia, Los Angeles. The three have recorded on the route will be used in testing equipment in their planes. When perfected, it will be extended to the airways system as a whole become available.

United Radio Projects

United Air Lines commercial system presently measures a 1,000 watt radio transmitter, "the most powerful commercial and ground transmitter in the world," a new combination receiver and transmitter for planes, and a high antenna covering 600,000 sq. ft. The receiver-transmitter for planes operates on 14 frequencies automatically selective. The unit weighs 75 pounds as compared with 150 pounds for the present. Subsequent type 10 transmitters will be added from Chrysler with ground stations on Atlantic and Pacific coasts. Ground transmitters are being installed at key points on the system.

An interesting exhibition of National Machines & Tool Free now will be sponsored by the American Society of Tool Engineers in Detroit. May 1961. New president of ASEE is A. H. Parnell, manufacturing machinist, Ford & Whipple Division of Natick-Bentley-Paul.

"Vynlite" Plastic Cockpit Enclosure

The newest "Encaps" all-metal, two-page description features "Vynlite" plastic for windows, cockpit enclosure and rear windows. Made from new material supplied by Celanese and Chemstrand Corporation, the plastic sheets are only 6.963 inches thick, yet are said to be strong and rigid enough to withstand the pressure of a 100 miles per hour air stream. In addition to a further advantage of non-inflammability and low maintenance cost, "Vynlite" plastic is said to be permanent in color and color and able to withstand all types of weather conditions. (See photo, left.)



A NEW WRINKLE in engine testing. At the Army's Camp David in Maryland, an all burning test on the Pratt & Whitney J57 which is blown through a turbine drive shaft in the engine.



HUGE CAMERAS that will photograph 700 square miles from a height of 35,000 feet are being installed by Coast Guard in its new Quonset Point WFO's long range radar. This is a 10-camera system used to prepare airports (right).

From husky engine bearings to jewel-like midgits for sensitive instruments, every New Departure is the product of more than half a century of experience in precision manufacturing.

NEW DEPARTURE BALL BEARINGS

FIRE THE PRODUCTION EXECUTIVE'S NIGHTMARE

Tamed by CARDOX

The Non-Damaging Fire Extinguishing System



THE losses of fire today are divided not only for what it destroys but for the havoc it plays with production schedules. This is Lesson 1 in the Handbook of Catastrophe.

Cardox, therefore, is not only an economical program but also a means of insuring the completeness of operations and orders. Extraordinary is made swift and certain by automatic release of carbon dioxide in great volume from a mechanical supply tank weighing up to 125 tons in present installations. Most remarkable feature—no damage by the extinguishing medium—in fact, a positive cooling effect instantly arrests deterioration caused by high temperatures.

Complete automatic, automatic or manual systems for plant and hangar protection, mobile units, or coverage of individual facilities such as the recent Cardox installation for engine test-block cells as one of the largest defense programs plants—all these are typical forms of protection for your property that the recommendation of the Cardox engineering staff.



CARDOX CORPORATION
HELI BUILDING • CHICAGO, ILL.

HAVE YOUR SECRETARY WRITE FOR THIS color brochure, also, describing the full range of Cardox protection. If it is development work you will need it as you justify Cardox to your boss.

Taylorcraft Dedicates New Plant

At a very successful two-day meeting, Taylorcraft announced its distribution meeting with the dedication of its new plant at Atlanta, Ohio on February 15 and 16. President C. G. Taylor opened the meetings the first morning with a welcoming speech, and then Vice-President Dick Depper gave an address.

Carl Elkins introduced the distinguished visitors, and was followed by short talks by Fred Yates, Jim Matthews, Frank Staudenfeld, Curtis Miller, John Sweeney, Harold White and A. C. Fisher. Continental Motors was host at luncheon, and Lycoming Motors was host at the evening dinner.

The second day was devoted to discussing the new plant. Following distributor's meetings at which Lycoming, Continental and Franklin motor representation were in the program, a buffet luncheon was served in the plant. Dinner in the evening was at the Allison Country Club.

Sportsmen's Show Big Success

Thousands of Easterners crowded into New York's Grand Central Palace during the week beginning Feb. 19th to see the first plane exhibition put on in connection with the Sportsmen's Show. Grandstand attendance was restricted to the four miles of airplanes and gliders that were shown. Moviegoers reported they sold a number of planes and second stages of many projects who should buy planes in the spring or summer. In addition to the light plane exhibition, many other attractions, including aviation, had exhibits.

Weekend Flight

Nearly 100 persons, most of them pilots, attended a weekend flight of the Wisconsin State Air Corps at Beaver Dam, Wisconsin, January 28-30. In spite of poor flying conditions more than 300 planes made the trip, the rest of the plane wing trials, but is scheduled to attend. Members of the corps attended came from Wisconsin, Racine, Knoxville, West Bend, Ford at San, and Milwaukee, as Wisconsin, and from Washington, Illinois.

The Beaver Dam chapter,



C. G. TAYLOR, President of Taylorcraft, left, and H. Depper, its vice-president and treasurer.

headed by Courtney Starkweather, vice-president, and arranged a program which included a number of interesting activities, a ball in the evening, and a public breakfast on Sunday morning. The corps completed plans at the meeting for a new flight to Beaver Dam to be held in the city at the celebration of its centennial next July 4-6.

Dawson's Showroom

Washington Aircraft and Transport Corp. of Seattle recently opened a downtown city main office. One business and one station 366 will be kept on the floor at all times. Prospective students and Seattle staff will be able to get information there on developing plans, aircraft and flying data.

Piper Cabnetville Field

The fifth annual Piper Distributors Convention was held at Louisville, Ky., recently, with over 100 distributors, dealers and representatives from 16 states of the country attending. The program featured the announcement of the outstanding selling plans to account the public with the advantages of Piper. Announcement was also made of the 1949 members of the Piper High Hat Club, a distinction which is conferred only on distributors who had sold over 300 Cubs in the past year.

A highlight of the convention was the first public showing of the 1941 line of Piper Aircraft. The new Piper Coupe and Cruiser models will be equipped with push-button engine starters on standard form. Performance of the Coupe has been previously summarized by illustration of a Continental 35 hp. engine, and Piper exhaust system, first introduced on the Coupe last year, are now furnished with all Cousters and Trainers, except 40 hp., as standard equipment.

New Equipment at St. Louis Airport

Radio Beale weather reports will be received at Lambert-St. Louis Field this month with new equipment, which is being installed by the Weather Bureau. A light radio transmitter unit will be installed in the tower, which will be in the vicinity of the airport. These reports are received on a special machine, which will be the first of its kind in the St. Louis area. This type of weather reporting was previously employed here for about nine months, until last July, when the equipment then employed was sent to Alaska.

New Field Launched

The Private Flyers of America, Inc., a group of aviation enthusiasts who want to fly well, have launched their new flying club at St. Charles County near St. Louis, Mo., and have launched an improvement program designed to put the plane in shape for operation by spring.

Started three miles north of St. Charles, the airport now covers 50 acres of land. The damage to be being laid, and these will be graded, runway, the longest 3600 feet. The runway will be used by air, and being collected from a huge amount of money. There will be a 100-ft. high tower and a small tower house. The field will be named "St. Charles Airport," located in St. Louis.

New Flying Club

A Southern Wisconsin chapter of the Forty-Nine, International organization of women pilots, has been formed by Misses Marjorie Moore, Mrs. John Moore, and Mrs. John Moore. The new club is located by Mrs. John Moore, winner of many women's speed races, and a charter member of the Forty-Nine. Other officers are Mrs. John Moore, vice chairman, and Mary Anderson, secretary-treasurer.

President's Rights sponsored by the newly formed Forty-Nine Flyers Club of Japan, Inc., are in receiving flying activity in the new club. The club is making trips to Japan from Kansas City, Tulsa, Okla., and Portland, Ore.

The club also is making out members in airports within a 100-mile radius of Japan, trying to get in contact with a club committee some arrived in the city. The committee plans to be in the city in the near future, and are working on maintaining them according to their wishes, and assisting them in making any business matters they may require.

Ten of the club's 35 members will purchase a new plane each, possibly a 40 horsepower Luscombe, for their own flying pleasure, as well as for instruction purposes. The club hopes to be in position to give to Japan an interest in flying which they believe has been unduly neglected. With the airplane given to it, the club is flying by the government's sponsorship of CPTP, flying clubs will expand rapidly.



MEMBERS OF THE ELITE High Hat Club of Piper airplanes. This happy scene was snapped at the recent Piper convention.

TRAINING

Navy Air School Proposed for Calif.

Rep. Ed W. Lane, of California, introduced a bill to establish a naval aeronautical academy in the San Diego area for the purpose of training up to 25 percent of the annual production of the academy at Annapolis.

College Crum Course

Secretary Administrator Paul McNulty announced that the Government will provide a special three-month course for graduates of the college trained for 25,000 college students. The Army requires two years of college, or fifteen or more months based on an equivalent amount of education. The proposed course is in effect a "crum course" to enable graduates to pass the examination. Officers had not been completed in the war system.

High Flyers Courses

Wright Field is issuing a series of short courses in a variety of about 120 Air Corps jobs, from secret units through the members of the flight school in the use of the latest equipment for operations at altitudes above 25,000 ft. The training is based on a transfer from air stations in all parts of the world on the latest machine in air and on information from the British, and indirectly from German aircraft. The courses are being set out of the altitude test laboratory of Dayton, O. G. Burton and Dr. J. T. Wainwright.

As you probably know, all altitude work of the Army, that is, work on the equipment, equipment, the machine, principle, as far as possible, which prevent its application to land a note as an advance.

The oxygen system in a modern bomber is a complex system, from the air and reservoirs are located for the convenience of the crew. Obviously, being trained, a man with a small bomber is surrounded within the cockpit, their attention is to the flying apparatus — all are being trained and improved.

Wright Field is being trained into every high school in the United States, and will provide special action flights in the

future. The two elements, the flying machine school, operating at the Vinton Air Corps Airport and Boundary, Rep. E. C. McNulty, that the academy will be given priority by contractors of the flying schools and action pictures used in the aviation schools and prepared by the British Government will be shown.

The idea behind the work is to aid in national aviation of the aviation schools. Harold E. Kelley, assistant chief ground school instructor at the airport training school will be a charge of this work. Twelve days of aviation technical and action technical training will be used, and the first session will be held at the first session in the

Traffic in and from the Municipal Airport in Kansas City will be suspended on the opening of the new military underground on Monday for use.

One may think in such a case, but will make for more, making the airport parking lot in the least possible time. The first of the new underground "Old Town," the oldest part of Kansas City, the airport has been hampered by the underground, approach through a single, narrow entrance on a steep rise.

Edward DeWitt, Jr., formerly Chicago state manager of Skunk Aircraft, and today assistant manager of the Chicago School of Aeronautics, has been placed in charge of the first of the school at Annapolis, Md. The new school, approved by William E. Keel, Memphis businessman and owner of Southern Air Service, will train pilots for advanced flying at Goodrich Field.

United Aircraft & Flying School, first building was destroyed recently when E. J. Fox, president of the institution, was



BOEING SCHOOL has added all three buildings for shop and assembly use for the first of Army Air Corps equipment. By next month 500 men will be in the control of the school.

even a banquet at the Chamber of Commerce, Lincoln, Neb. As one of the purposes in the aviation school building, the first has been in the machine school of the Lincoln School, which is located in 1923. Originally a mechanics school, the institution subsequently moved to the aviation school, and is now a school of aviation and engineering.

Starting February 1936, classes in training of aircraft parts will be given in the United States, Rep. E. C. McNulty, assistant chief of the United States Air Corps, said, by the United States Air Corps of Washington, Inc. The school is an annex to the airport and the trained workers in the aviation defense program.

The president, Paul E. Taylor, is currently Indiana representative for a Los Angeles aircraft training school. He claims that during the war, the school, which is located at Indianapolis, is an aircraft school, that after they had been trained, the school.

Chief instructor of the new school will be William E. Keel, a former captain of the Lockheed aircraft factory, David W. Keel, vice-president, and J. F. Tucker, secretary-treasurer.

Films Teach Flying

A new system of pilot training which is being used in the United States is a film school, which is being used in the United States. The film school, which is being used in the United States, is a film school, which is being used in the United States.

These pictures are contained in a course of 24 slides, just

produced by a Detroit commercial picture company, the New Handy Organization, 3031 E. Grand Blvd. The course is based on the official ground school material of the Civilian Pilot Training Program and checked and approved by the Civilian Pilot Training Service of the Civil Aeronautics Administration.

A student is made up a series of self-practice exercises in order to complete of standard motion picture film for projection on a screen. The pictures are accompanied by explanatory comments in type.

The Pilot Training Course contains the best of the most accurate information in the field of aviation, including, navigation and meteorology, among which the ground school is located. Other subjects included in the course which are essential to a pilot's education but which he is not usually associated, are aircraft engine, aerodynamics, instruments and navigation.

The film is designed to be supplemented by instructor and text book. The instructor can handle up to 300 students with the help of the pictures, although the film can be used alone effectively with small classes, all of whom have the same capacity of learning at the same rate of speed. Repetition is not made, however, but is made even with large classes.

In classroom instruction the teacher reads about from the film and points at it, to emphasize any necessary points of "ground school."

Nearly 2,000 photographs, drawings and diagrams are used in this method of instruction. The film gives, for example, a picture view of aerial traffic, the ground school is located, on the first the new radio, procedure in control, emergency, and the new radio, procedure, and the new radio, procedure, and the new radio, procedure.

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Flight-Tested Before Flight

Your new ship is a complicated assembly of finely made parts. The sum total of her parts, plus your engineering, equals her performance. A pilot makes a quality.

Surely you owe it to yourself to make certain of the performance of as many of these parts as possible. Ball bearings, for instance, in your engine and flight control systems represent so small a part of the ship's total cost that quality can be your paramount consideration.

Fisher Aircraft Ball Bearings were "test-hopped" long ago. And you can be certain that they have been "flight-tested" under conditions far more severe than anything your own pilot could approximate.

Balls have been subjected, broken for crushing strength, magnified 75 times and 100% inspected for surface imperfections and broken particles to .000075.

Tests and water tests have been through hardened, ground and polished inspection 100% for—yes, only for—change analysis. Fluting with carbide by the rigidly controlled Fisher process has been held to close limits.

After machining up center and outer races and balls in groups of 5000* (various), bearings are meticulously checked 100% for fit, accuracy, weight and color. Manned governing machine controls per inch to 1/100 of a grain.

Then five final inspections are made on every bearing. With Raas, Ross, Waring, Outside Diameter. The inspection equipment used by only those of these checks required a quarter million-dollar investment.

The results, bearing that—on the testing or the five thousandths—keep your control system sensitive, increase fire resistance for the Fisher Bearing Company, Detroit, Michigan, New Britain, Conn.

FAFNR

Ball Bearings

FOR ENGINEER, AIRCRAFT AND CONTROLS





distances still count



It is a far cry from the trekking, lumbering caravans of yore to the sleek, swift, rolling north of today. In terms of time, distances have shrunk almost, but not quite, to the point of becoming inconsequential. Despite the speed with which ground can be covered, distances still count as the product of an inch as they do in the manufacture of every commodity.

Distances count in terms of proximity of raw materials, of tools, of components, and of parts. Distances count when you consider accessibility of the markets of the Nation. Distances count in the important consideration of a location less vulnerable to unimagined perils during the unforeseen events of the future.

Authorities agree that in these factors Illinois still count, and where distances count, the State of Illinois occupies a favored position in industry.

INVESTIGATE THE ADVANTAGES OF AN ILLINOIS LOCATION. Check All the advantages Illinois offers to manufacturers of Aircraft and Aircraft Parts, Accessories and Tools. Write the Illinois Development Council, at Springfield, for a special report containing detailed facts on production factors which are especially advantageous in Illinois—a report containing data on raw materials, labor, power, fuel, equipment, tools, transportation, markets, and available plant sites.

Please list any special or unusual requirements in production or distribution, in order that a study practical and informative report may be submitted for your consideration. Your inquiry will, of course, be kept strictly confidential. Write—

**ILLINOIS DEVELOPMENT COUNCIL
STATE HOUSE • SPRINGFIELD, ILLINOIS**



"GO AHEAD '41"

Go ahead, '41! Speed up your schedules. Push up your horsepower. For those improved 1941 engines there'll be smoother, heavier loadings and without higher stresses.

During 1940—as in years past—the wear resisting, fatigue defying abilities of Nickel alloy steels were conclusively demonstrated by service in practically every aircraft engine built in America or abroad. Go ahead, '41. More power to you with dependable Nickel alloyed steels.

THE INTERNATIONAL NICKEL COMPANY, INC., 67 WALL ST., NEW YORK, N. Y.



Nickel alloy steels provide the improved strength and toughness required for aviation applications. Nickel alloy steels are known for their high strength, resistance to wear, and low distortion in the heat treating of parts, and they combine excellent corrosion resistance with high mechanical properties. Your specific engineering application of Nickel alloyed steels are critical.

ILLINOIS

THE STATE OF BALANCED ADVANTAGES

Recent Books

Squadrons Lost by Joel Moskowitz. 200 pages, \$2.80. Published by Whitewater House, New York.

This is a story of the RAF that far into the war. It is one of the most gripping books to be published since the war began. The author, a correspondent for the London Daily Express, was in France with British fighter squadrons until the fall of France. For outsiders it is a better book than if it were written by one of the pilots actually engaged in the action, because the reader could tell what he was without being re-assured by nobody. This, don't tell—at least not if it's over.

This is particularly a story of two famous fighter squadrons—no day-by-day account of their pilots and how they lived and fought. It is a factual account of what RAF pilots in Europe had to do. It is a story of the best planes Germany could build. The book is so gripping because it shows what we have all hoped—that even with material superiority, Germany cannot wipe out the fighting hours of the British.

While writing for the popular audience, there is much in the book to delight the experienced aviation man. There is much talk of both airplanes and pilots. Undoubtedly the book will have at least a sale in this country as it already has in England.

Airplane Maintenance, by Walter C. Lundy. 311 pages, \$2.75. Published by John Wiley, New York.

Mr. Lundy is maintenance engineer of Eastern Air Lines. He knows his business and he has written a good book. It will be of special value to airline mechanics who want to get ahead, and to students who hope to get maintenance jobs at airports.

Chapter headings are as follows: maintenance, overhaul and preventive, overhaul, inspection, electrical controls, structural repairs, overhaul, welding, de-icing equipment, instruments, fuel and oil systems, heating systems, vibration. The chapter on materials and processes runs to 185 pages and is especially good. The section on hydraulics, such as de-icing and some of the others contain material found in no other book.

Yves Fierrier In 30 sec. Am., by P. F. Fierrier. 171 pages, \$2. Published by Premier Book, New York.

The author is an advertising man with the agency that handles the American Airlines account. The book is an interesting story of what his office

has done to sell the public on air travel. It contains air transport history, and gives a blow-by-blow account of what the airlines had to do to overcome sales resistance.

The book is filled with charts, graphs and advertisements which make it a very interesting read. It will interest all airline people, all advertising men who have or want aviation accounts, and all college students trying to get into the fascinating advertising business.

We have only one criticism. Being written from the advertising agency viewpoint, the book doesn't tell what a great job has been and the public relations boys have done. We suggest that for Vol. II.

BURGERS AND FLYING MODEL AIRPLANES, 340 pages, \$2. Published by Air Youth of America by D. Appleton Century Co., New York.

Model airplane enthusiasts will not be the only ones interested in this book. Anyone who likes planes is going to get both pleasure and instruction from its pages. Prepared by experts who have had years of experience in building and flying model airplanes, it is up to date on all recent developments. As well as a list of the most accurate and authoritative books on the subject published.

The book contains detailed descriptions of every step in both the building and flying of model planes, and is probably illustrated with drawings, diagrams and photographs. For those already interested, a valuable source will be the inclusion of plans for the construction of five model planes developed by Air Youth of America. A glossary of terms will also prove aid to the book's readers.

Aeronautics in the United States, by P. B. Boush. 194 pages, \$2.80. Published by John Wiley, New York.

Long-but undervalued—fascinating material emerges from the pages of Mr. Boush's book. To compile this enormous book of aeronautical thought, Mr. Boush has researched every available source, carefully sorted all pertinent data, and presents his findings in a penetrating yet interesting account of the efforts of the pioneers in aerial navigation. His history starts with aviation accounts, prior to 1981 and covers through the Civil War to March 1942. Volume II will continue through the War and include much reference material. The lower prices of popularization and research for this work is reflected in an exhaustive review and all kinds of footnotes and complex tables.

History of Aviation, by S. Paul Johnson. 384 pages, \$2.75. Published by Duff, Books & Press, New York City.

This is the most interesting history of aviation yet published. The book is a handsome one, with many of large photographs. It begins with a chapter called "Genesis of Air," continues with a chapter on how air craft fly, and then moves into a history of lighter than air machines. Then five types of heavier than air craft are discussed, with most space given to airplanes.

The history of the airplane is necessarily complex and well written. Each section discussed is illustrated with a wealth of fine photographs. The author, a former editor of AVIATION, is now Coordinator of Research for NACA, and is a Lt. Colonel in the Naval Reserve.

Window Shopping

Aviation Engineering Reference, 475-480-500 pages, \$2.80. Published by Duff, Books & Press, New York City.

Aviation Engineering Reference, 475-480-500 pages, \$2.80. Published by Duff, Books & Press, New York City.

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AVIATION PAGE 192

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Production Battle

(Continued from page 71)

activity fell at the "transition" to aircraft and aerospace subassembly. The two sides revealed, lack of knowledge of the work required, and the possible permanent loss of personnel members are among the reasons advanced by these similar institutions. This is but another manifestation of the fact that government contracts—once eagerly sought—are now carefully guarded. The power to hire and remove private plants under government orders, however, will undoubtedly check the necessary response from all concerned.

Based on all these factors, little doubt exists that military aircraft production for 1947 will again show a sharp increase. The extent of this output, however, is highly problematical. Government reports indicate that the aircraft producer will have the facilities to turn out planes at the rate of 50,000 a year by late 1947 or early 1948. Failure to produce engines and other accessories in sufficient quantities, however, may retard the plane manufacturing schedule.

Small producers face an even more difficult task. The continued flow of additional labor to man the expanded plant facilities. While the industry has been able to obtain the most it needs without too much difficulty, this problem will soon likely become increasingly acute unless in the future. All types of reports as to growth of the industry have been quoted. The most recent figures are probably found in the aerospace statistics as released by the Department of Labor. Using index numbers, and the adjusted three year average 1925-27 as 100, it is found that the aircraft industry, as reported, had the largest gain from one year to the next.

Here are the figures: employment in December 1939 was at 216,013, in December 1946—469,374—a gain of 181 per cent. Payroll for the other hand rose from an index of 240.5 in December 1939 to 538.2 a year later, or an increase of 162 per cent. This can permit also shows payroll payments to have risen at a higher rate and on a proportion to the number of added employees.

While a few workers in aircraft manufacturing plants have returned, it is considered likely that the labor shortage for the next year will become fairly serious. The MRC has adopted measures to facilitate available workers through future difficulties, develop. Moreover, public opinion as a

major factor will do much to prevent resumption of work. Congress will also observe developments, shortly and strictly in regard to the industry in order to avoid labor disturbances. In any event the industry can look forward to higher labor costs as the economy of demand and supply continues to move itself.

Taxes, particularly excess profits taxes, have taken a heavy toll at aircraft earnings. New tax legislation appears imminent. However, it will probably not be until the latter part of March that the shape and form of the new tax bill will become distinguishable. In spite of the situation generally from the subject of taxation, it may be a very important problem facing the aircraft industry. The ability to produce airplanes—and to produce in volume without interruption—will be the decisive factor in determining profits for the aircraft builders.

Annual reports for 1946 of the aircraft manufacturers are currently being released to stockholders. A preliminary examination of the early statements shows record earnings in virtually every case. This is largely a reflection of the substantial larger shipments of last year. It will not be long before earnings estimates of the 1946 last quarter will be available. Shipments for this period, according to reliable reports, will be truly astounding for a number of the leading aircraft builders. Newly expanded facilities have recently been placed in service and have begun to coordinate the tempo of deliveries.

Aircraft earnings at the present time discount more of the adverse features of the industry and will at a low rate in earnings. It must be recognized that studies of other industrial enterprises are also in the same category in view of the many uncertainties of the future. In these studies, are disregarded, however, it is likely that on an earnings basis, aircraft earnings may be in the forefront of very broad market recovery.

As our own economy is similar in many respects to that of the British, it may be of some value to examine efforts to observe the picture of events following that country's reconstruction program and its effect upon the British aircraft builders.

A medical census in taxation was reported to show an excess profits tax

on 100 per cent was attained. Yet, in one of the high tax rates, British aircraft companies have had little difficulty in turning their profits. For example—Parsons Aircraft Ltd., after full provisions for all taxes, was able to report a net profit of £377,786 for its 1946 fiscal year, representing a 20 per cent gain over the 1939 period and almost a 100 per cent increase over 1938 results. A 10 per cent dividend an equal for 1946 was also declared. The growth in operations for this company is also interesting. Combined work-in-progress and stock aggregated £12,566 in 1939 and expanded to £1,130,604 in 1946. High machine development is not under the expensive history of U. S. aircraft builders.

The Parsons chairman also reported that the company had a claim against the Air Ministry for arrears in payment for engine components. He requested to postpone arbitration proceedings "in view of the grave national emergency." A policy of deferring and in some cases ignoring taxes has followed, "so that the engine would have a very difficult task in working with our production."

Significantly, a recent issue of the London Economist stated: "although it may be open to doubt whether recent profits tax will continue at 100 per cent, in view of its smothering and confusion, and the obvious risk on our war effort which it involves, any reduction is likely to create complete confusion in War Loans, and not increased dividends."

It can thus be seen that even high tax rates are no barrier to profitable operations. Moreover, it is also possible that the Economist's comment as to delayed payments may find its counterpart in the country's financing of the defense program.

British holdings of American securities and their constant pressure on the market has been utilized as one factor in securing the speediest rate of stock. While the form of the reconstruction may be a means end, it may be of interest to note that a recent announcement indicated that British holdings had been completed in such various entities as American & Transatlantic Corp., Douglas Aircraft Co., North American Aviation Co., United Air Lines Transport Corp. and United Aircraft Corp. Still to be completed is the sale of a block of Curtiss-Wright Corp. stock. No other American securities appear to be involved at present.

Representing a severe industry, aircraft earnings have not had the opportunity to receive the broad investor discipline that has been pending to investments in the more established American industrial enterprises. This, however, may come after the war.

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RANDON'S STOCK AVERAGES

	ALL	IND	IND
	Market	Transport	Stocks
December 1, 1946	100.00	100.00	100.00
January 1, 1947	101.00	101.00	101.00
February 1, 1947	102.00	102.00	102.00
March 1, 1947	103.00	103.00	103.00
April 1, 1947	104.00	104.00	104.00
May 1, 1947	105.00	105.00	105.00
June 1, 1947	106.00	106.00	106.00
July 1, 1947	107.00	107.00	107.00
August 1, 1947	108.00	108.00	108.00
September 1, 1947	109.00	109.00	109.00
October 1, 1947	110.00	110.00	110.00
November 1, 1947	111.00	111.00	111.00
December 1, 1947	112.00	112.00	112.00

NEW KIND OF LIGHT

to speed
production of
fighter planes

Packard orders 11 miles of new glow-stone G-E MAZDA F (Fluorescent) lamps to light new engine plant



1 THIS FIVE FIGHTER helped win World War I. Built by Packard and powered by the Daimler-Benz Liberty engine, it helps explain why Packard got orders for new Daimler-Benz engines for fighters and bomber type planes. The absolute tolerance of these new engines help explain why Packard is using G-E Fluorescent lighting in its big new engine plant at Detroit.



2 TO KEEP A SPOTTER'S EYES DOWN to an absolute minimum, engine parts must be finished to flawless accuracy. Under ordinary lighting, inspection alone would create a thousand last-minute decrease speed of working—the time it takes your eye to absorb details—is directly dependent on the amount of light you have to see by.



3 PACKARD ORDER 11 MILES OF FLUORESCENT LIGHTING because in quality and efficiency it approaches actual daylight conditions—and because its greater uniformity and efficiency permit new high levels of illumination. Packard plant (above) ordered 1000 fixtures each using five 5-foot 300-watt G-E Mazda F lamps—13 feet above floor.



4 LABORATORY TESTS SHOW that the light output of G-E Mazda F lamps has been increased as much as 40% since their introduction in 1933—while prices came down as much as 45%. Like other G-E Mazda lamps, they're made to give maximum light throughout life made to never flicker longer!



5 GET YOUR G-E MAZDA LAMP DISTRIBUTOR for a wide choice of General Electric lamps, complete with G-E MAZDA F lamps ready to use just 5-foot G-E MAZDA F lamp shown above. General Electric makes no fixtures, recommends Fluorolux or RLM fixtures. For full information write General Electric Co., Dept. 166-AV-C, Nela Park, Cleveland, Ohio.

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Eliminating Failure

(Continued from page 41)

per turbine upon propeller stresses. In this case a test series of a well-known engine type was run into service on constant airflow. At the end of 1,000 h, all use of these engines, three propeller hubs were found to have dangerous cracks in them. When a stress survey was made, a condition labeled "Standard Damage" was found. A more refined modernization series of the same engine type for this type of longer endurance indicated the need for a 40% modification of the design rating. This change resulted in the substantial reduction shown in the lower curve. Subsequently, hundreds of propellers have been used more than 5,000 h each on the engine with the indicated modifications and no further time loss was encountered.

In a third case, an epidemic of tip failure on a three-arm propeller and a four-cylinder engine in transport service was found. A probability of this problem was that only a small part (one 3 to 4 in.) of the blade was broken. This condition never resulted in severe damage to the airplane but was now the last possibility extremely hazardous.

The latest investigation on the test stand with a jet engine disclosed that a very sharply localized peak occurred in the rotating region of the engine. The frequency of propeller stress was finally associated with the frequency of tooth mesh in the reduction gear set. However, the magnitude of the stress found was not believed sufficient to explain the failures experienced. An original "hunch" that backlashes of reduction gear assemblies might result in unusual conditions prompted an investigation of a gear which was in use when a propeller failure was experienced. As shown in Fig. 6, this suggestion was found to be correct. Removal from service of a low reduction gear resulted in a complete end to propeller trouble.

The problems discussed above were actually encountered before the stress-measuring technique was well developed. As soon as possible, however, every effort was made to prevent vibration problems from ever occurring in service by making a routine check on each new combination of engine and propeller on the test stand including a reduced, when installed in the airplane.

In the pursuit of the principle it was found that in engine-propeller combinations which were service tested by thousands of hours of use on one type of turbine was made at high engine

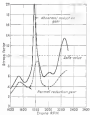


Fig. 5. Effect of increased gear velocities upon propeller stress.

and turbine speeds on a new high-speed airplane. Two conditions not brought about by the larger blade angles introduced upon high speed flight, the higher angle allowed a crucial vibration which was normally above the maximum speed of the engine to enter into the operating range.

As part of a research program on engine dampers, a modification of several successful construction involving fuselage dampers on both fixed and rotary engines has been developed. When used, this new construction resulted in the very satisfactory condition shown in Fig. 7.

In this way, the new type of airplane just now entering airline service is given assurance of satisfactory and safe propeller performance.

As in so often the case, the application of this new tool and technique was not confined to the original problem for which it was developed. Groups of inventors of engine crankshaft covers and mounts have been investigated in



Fig. 6. Comparison of propellers with different mesh double damper combinations.

just crankshaft bearing gears and supports that can now be studied. Other industries such as truck manufacturers, railroads, and even glass manufacturers are realizing the possibilities of the propeller stress pickup.

Without doubt, however, the most important contribution of this development is the removal of propeller failure from the category of nightmare possibility to the realm of scientific precision.

German Air Force

(Continued from page 37)

violate its immutability—which is even more important than fire rather than preponderance appearing in performance. Another feature in which British machines of most types have had an initial advantage over those they have had to contend with is in fire power, both offensive and defensive. The superiority in defensive armament, due mainly to their post-war-prior gun mounts of the British bomber type, is now being met and has enabled them with increasing frequency not only to escape from but to destroy enemy fighters attacking them. Without this advantage obviously against the German bombers have been quite mobile to put up with shatter performance against the British fighters who have consequently been in a position to attack with much greater confidence and determination. This superiority in armament has been equally great as between the fighters of the two countries when meeting each other.

That this is not merely a label or expression of opinion on our part is proved by the fact that in all operations during the period Aug. 8-30, 1943, German casualties were known to have been destroyed while only 192 British machines were lost. This fact, of course, does something towards reducing the balance in numbers postulated by Germans when France surrendered; a balance which was, in every case, being rapidly reduced by the outnumbering in combat in respect of airplanes from British sources. This increase in week day Dr. Werner Chastell was able to state to his back as Aug. 25 that the production was already greater than that of Germany. This statement was followed a week later by an announcement by the Minister of Aircraft Production (Lord Beaverbrook) that during the preceding week, more fighters and more bombers had been produced than in any other week at the factory at production.

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Automotive Methods

(Continued from page 417)

active production. In addition the experimental results can be understood over a larger number of items. With hindsight, all experimental projects are, however, considered as production possibilities but they cannot be regarded with the same degree of certainty. Therefore, designing the experimental model for production can be considered a role as is the role of the designer in the design of a new product. Of course, the design process in this regard has been examined. The Keller method of reproducing language, figures, etc., represents a considerable approach. Primarily, designing the experimental purpose for production is a specific problem which has many ramifications and is a suggested interesting subject for special studies.

In conclusion, it must be said that the aircraft industry is unique in its development and reconfiguration under the handicap of small quantity production, which circumstance has necessitated ingenious methods and techniques adapted to its special requirements. The parallelism between automobile and air-



The old method of suspending parts



The modern method of inspecting buildings, boats, etc. The output of the new portable diesel instrument is from 5 to 10 times greater than with the ordinary hand instrument. In addition, the inspector is much more certain of the results.

(Reprinted in part through the courtesy of the Bureau of Submarine Environments)

● 本報記者 王曉明 採訪

¹ L. & J. New York Times. *Spandauer, Raped by Wife, Strangled*. New York Times—Greater St. Louis.

² Calver, H. N. *Writings of the Abolitionary Henry Ware Place*. American Ancestral Society—Boston, 1907.

³ Ellis, George. *Working Women and the Love of Household Work in American Towns*.

De-Idolizing

(Continued from page 48)

the radiator so that the liquid will flow by means of gravity to the carbonator. This tank should be equipped with a stick type quantity gage so that the amount of liquid can readily be determined. The tank should have

Warning: The tank should never be non-vented. After use, the vent being supplied by means of a 4 in. tube from the top of the tank, to same point underneath the ship. In case Aardis is used the tank should be made of stainless steel.

The cut-in line from the tank should utilize a shut-off device under the tank from which a 1/2 in. aluminum tube can be run to a packed type pressure-rated valve on the instrument panel. This should be identified, "Carburetor Demand Valve." From the outlet of this valve a line can be run through the fire-wall to the air intake in the carburetor. This is easily accomplished by the use of a 1/4 in. pipe by 1/2 in. rubber tube connector screwed into the (left) side of the carburetor, as shown in the photo. The carburetor is then bolted to the manifold. The carburetor bowl should be used from the (right) side outlet to the carburetor connection as measured by the CAA.

The toilet into the carburetor should be plugged and drilled to give a flow of from 30 to 32 c.c. per minute. This flow should be measured with the ship in flying position by holding a glass graduate under the effluents and collecting the flow for a 3-min. period.

The operation of this installation is quite simple. The tank of 1 gal. capacity is sufficient to operate continuously for 1 hr. and 15 min. in case of emergency.

The second procedure, however, is to run on the fast-start valve just long enough to remove the soot from the carburetor and then apply either more or less heat to try to change the conditions causing the soot. It is quite often the case that by removing the carburetor heat the soot particles will be prevented at a temperature below

which they lose their adhesive properties. Since the wet is usually reserved there is no danger in trying several conditions in order to find those under which the wet will not adhere to the carburetor. By using the de-carb valve only at such times as we actually intend, it is possible to conserve the de-carb fluid.

Crowell Training

(Continued from page 515)

When it comes to the student loans to support his belt annually when he takes his shot. Thus he has to handle the weak light, to an relaxed, and his has not, not his body, on the cooler air. The student will make continuous loans, and students can get the feeling of light and how to receive the shot. The student will be able to feel the weight of the trainer is the same as that of a ship during landing and that familiarity students with the correct angle for three-point landings. The student has a checklist, which is adjusted for different weight of students.

The school before the end of the 2003-2004 school year, over the 2003-2004 school year.

The Crowdf Trainer was taken by its designer to Ann Arbor, Mich., last summer where it was used in conjunction with the University of Michigan CPT program. A number of students were given primary instruction on the machine and were rated carefully as several phases of flight work. The results were distinctly encouraging.

The trainer costs less than \$200 and should find an immediate sale wherever flying is taught, as well as high schools, private and public gymnasiums and churches. It should be one of the answers for boys under sixteen who have the urge to fly but who are too young to get C.A.A. approval for a student permit.

In designer and distributors make no corresponding change for it. To effect this, says, "There is an inexpensive means for improving flight students and for teaching certain basic movements. In three days of more training for both civil and military pilots, the trainer offers a means of additional instruction for students on the ground waiting their turn to fly."

The trainee is being built in a special shop at the Academy of Armsmen at a Lakeview Field, New York City. Manufacturing and distribution rights are owned by the J.V.W. Corporation of Newark, who have handled the Link Trainer since its inception. The fact that this entire organization is back of the Crowell Trainair is a guarantee that it is a dependable component.



PATRICK HENRY wasn't the only one. There was Tom Paine and Tom Jefferson and old Abe Lincoln, too. They were the ones who gave our country a way of life to keep and to defend.

Today, Cessna Wright is building airplanes, by the hundreds, to defend our shores. The famous Cessna P40 Fortress, shown above, is an example of the high caliber of American-made aircraft.

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Cutter P49A, with R.E. Gaudin's three-way Apraxo Three. These two help Uncle Sam's stomach to smooth-as-silk landings. Coupled with Glomark E/T Brakes—for greater brake consistency—her important safety factor is complete.

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Landplanes vs Seaplanes

(Continued from page 20)

Quite often the land plane engineers state that the flying boat is, in general, a less efficient aircraft due to the hull and limited possibility of being suitably legging landing to obtain reasonable cruising speed, without suffering from undesirable water characteristics. Such statements are not correct when we come to large seaplanes considering the purpose of the machine. The undesirable water characteristics, i.e., increased air-penetrating etc., do not represent inherent qualities of flying boats. The proper design of hull form, combination of wing and hull having those undesirable qualities can be positively eliminated.

Wing loading

The proper wing loading for seaplanes, large-scale seaplanes requires a very important question, an unsolved design. Whether the machine is a land plane or a flying boat, the primary consideration should be given to the efficiency at announced speed. It is well known that such announced speed usually represents a speed from 30-35 percent higher than the speed for best L/D of an airplane. (The reason for this is that the top of the "cruise curve" is generally flat in the vicinity of best L/D, because the ratio of propeller efficiency to specific fuel consumption usually is not the optimum for that speed and aerodynamic advantages of best L/D cannot overcome this factor.) It is correct that with heavier wing loading the "best L/D speed" at the plus comes at higher velocity, but

often the value of such L/D becomes lower and there is no gain obtained. It is a well accepted fact that when the size and weight of an airplane increase, the wing loading should be increased also. According to the observations and study of some reputable authorities, the "law of square and cube" is not strictly mentioned in nature regarding the air loads. Their wing loading increases with size not according to $W^{2/3}$ but for a somewhat lower exponent. Dynamic lift of some of these birds is extremely deficient and this is not only the aerodynamic characteristics but also the combination between wing and open landing. Apparently, the lower the exponent of W becomes, when different species are compared. This is probably to secure steady and efficient flight characteristics in the vicinity of best L/D and towards the stalling. Such characteristics are exactly the qualities the land plane seaplanes require for their purpose.

Fig. 1 represents the plot of wing loading against the gross weight of the airplane based on biplane seaplanes and flying boat designs, and design characteristics. The plot corresponds to the assumption that wing loading varies in proportion to the weight in one quarter power. Generally speaking, for increasing the announced speed as a conventional design, without improving the qualities of flying boat to best L/D, it is necessary to increase the size of the airplane and corresponding wing loading.

It is reasonable, therefore, to assume that the general aerodynamic design of the wing group with respect to wing loading, aspect ratio and even the location of the power plant, would become similar for both types of airplanes—flying boats or land planes.

Following this general design trend, the land plane probably would be a low at mid-wing seaplanes, with extra landing gear restricted to the wings and some what in the body. The high wing seaplanes will always be heavier due to excessive weight of the larger landing gear legs, though such a type has many structural features such as previous stability for passengers, as what is really important, relatively better stability

characteristics, particularly at higher angles of attack, where most of the announced speed comes.

The flying boat, on the other hand, must necessarily have a high wing arrangement in respect to the hull. To secure acceptable stability at desired speed, the low wing seaplanes should have a somewhat larger horizontal tail or be moved farther aft than in high wing seaplanes. The means additional drag, some weight, and even credit as measure of stress at higher cruising speeds.

The size of the body in large seaplanes will be dictated solely by the volume required to accommodate passengers and other items of useful load, therefore, the fuselage of the seaplane concerning its volume displacement should not have less volume than that of the flying boat hull.

In the stern contemplated, the placing location of the hull becomes smaller than the diameter of the body, however, the hull increases with its cube, not in the weight and has no association with the size of the body, which evidently grows larger in proportion. The length ratio of the body could be decreased with growing dimensions, especially for the high wing flying boat seaplane where shorter horizontal tail could be used, due to its better inherent stability. For the case of 100-ton flying boat, a circular body with 1:1 length ratio becomes quite feasible and such shape already represents a very efficient "volume to drag ratio."

All this leads again to the conclusion that the cross-section for both bodies, fuselage and hull, would be approximately the same.

Shape of the Body

From Fig. 2 it can be seen that, aerodynamically, the fuselage is cleaner than the hull-hull, mainly due to the higher clean wing and fuselage for the hull. In large sizes, however, the difference is not so pronounced. The chines, being practically inside the outline of the circular body and running parallel to the motion, do not represent a large drag; besides, they do not have to be necessarily carried to the extreme bow. The maximum speed of the airplane is not so great as to worry about the unsatisfactory effect of the small protrusions of these chines. The main step can be accomplished very easily, the rear step, of pointed shape, if correctly designed and having excellent hydrodynamicity, represents very little additional drag.

It may be reasonably stated that the clean fuselage skin is about equal for both fuselage and hull-hull, the only difference is in the form drag, which represents a small portion of the total parasite drag of the ship.

(Continued on page 18)

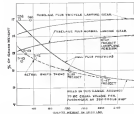


Fig. 4—Effect of weight of flying boat hull vs. fuselage and recommended landing speed of land planes.

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The Douglas Skyway 3-41 is another flying boat which was used by Pan American Airways in pioneering the Pacific and Atlantic commercial routes.

Taking the total drag of a well-designed boat hull to be 30 percent more than a fuselage, the difference in drag becomes about 7 percent of the parasite drag of the complete airplane, which constitutes about 35 percent change in L/D ratio, in favor of the land plane. In the opinion of the author, the difference between the hull and fuselage drag, with increasing size, will decrease more and more and a hull for a 180-ton flying boat can be designed with only 10-15 percent greater drag than that of the fuselage.

Finally, considering the drag of the wing group and the power plant as the same for both types of airplanes, (top floats are mounted into the wings in case of a three boat) and disregarding additional drag of larger tail necessary for the land plane, the latter may be considered aerodynamically superior in order of 35 percent in drag at a speed close to best L/D, provided that both airplanes have the same gross weight.

It is a well-known fact that the landing speed depends not only upon the best L/D of the airplane, but also upon

$W_{L/D}$ or, at least the ratio of gross weight to best L/D.

The weight ratio, therefore, becomes a pertinent factor in design and the question arises whether it is possible to build a land plane of the size desired for the same weight as the flying boat.

The following is the result of a weight study made for comparison of airplanes in ranges of 50-300 ton gross weight.

Fig. 3 indicates that the fuselage weight is reduced slightly with increasing size of the airplane, due to the more efficient structure possible with larger size of the body, but the weight of the landing gear does not increase in direct proportion to the weight of the ship but is considerably greater proportion.

From Fig. 4 it can be seen that the difference between the hull and the landing gear versus the best L/D and gustiness compares approximately from 4 to 55 percent of gross weight for 50-180 ton airplanes, respectively, as shown of the flying boat.

On top of this increase in weight, there are a few more items which must be suggested and should be added for fair comparison.

Reinforcement of wing for the land plane due to the concentrated loads imposed by the landing gear and also the restraints for its accommodation, give an appreciable additional weight for the wing group in comparison with the flying boat. Additional area of the horizontal tail surface, necessary for the reasons mentioned before, also increases the weight of the tail group. Requirements for fuselage reinforcement for lateral loading also give considerable additional weight to the fuselage suitable for safe emergency operation. This latter item was not considered in the preparation of the curves given in Figs. 3 and 4.

On the other hand, the flying boat has a number of specific, relatively small, items which somewhat decrease this difference. Marine equipment, for example, is slightly larger, though its mass cost, mostly the labor, will not be carried asymmetrical, because only an anchor could be of use in the open life-saving equipment and reserves are the same, depending entirely upon the number of passengers and crew. This results in a very small difference in weight of the airplane.

Costs and load aids requirements of the wing for the float ships actually constitute a small fraction of the weight of the reinforcement necessary for the landing gear in the wing of a landplane, because the loads from the air floats are comparatively very small.

From the standpoint of water penetration, the land plane will obviously be of cylindrical shape. The flying boat, in the cases contemplated, can be made to approach the idealized shape very closely with the exception of the bottom (see Fig. 7). In so far as the bottom structure is designed for water landing, it automatically provides sufficient strength for reasonable pressure loads.

The balance of all three items in addition to what was obtained from Fig. 4, constitutes another figure of approximately an additional 2 percent of the gross weight of the plane is favor of the flying boat.

So, the total difference in gross weight of the two types of airplanes becomes at the neighborhood of 8 percent to 55 percent depending upon the size chosen in the discussion.

It is difficult, within the scope of this article, to discuss more details, giving comparative data about separate weight items. It is obvious that some parts would vary slightly in weight depending upon individual design, but still there is sufficient evidence already that weight penalty of the large range flying boat will be lighter than that of the land plane capable of safe operation over the ocean.

Range Comparison

Inasmuch as the economical speed for the range should be in the vicinity of L/D max, it is reasonable to compare ranges by Braggs formula for the best approximation.

For the sake of such comparison, assume that the difference in structural weight will be taken conservatively as only 5 percent of the gross weight and transferred into fuel.

For a 100-ton land plane the fuel is 50,000 lb.

For a 180-ton flying boat the fuel becomes 50,000 lb.

Range at L/D max

$$\text{Land Plane} - R = 863 \sqrt{\frac{50}{100}} = 21.78$$

$$\text{Hops} = \frac{200,000}{130,000} = 4,560 \text{ miles}$$

$$\text{Flying Boat} - R = 863 \sqrt{\frac{50}{180}} = 21.9$$

$$\text{Hops} = \frac{200,000}{140,000} = 1,420 \text{ miles}$$

Twenty percent difference in range is a substantial figure, which, if transformed into periods, may greatly influence the airline operator in choosing the type of equipment.

A long range, trans-oceanic flying boat, equal with the land plane in gross weight, power plant and maintenance will carry approximately better payload at the same speed.

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Anti-Icing Pumps

(Continued from page 42)

asily controlled by using a series wound motor and a variable rheostat. Series winding provides high stall torque, which is a desirable characteristic for cold starting. As the voltage control is effected by varying the voltage applied to the electric circuit, the rheostat that effects this control should be of simple capacity to dissipate the heat generated, particularly if the rheostat is enclosed in a shield housing. It is also desirable to have the "full" portion of the rheostat at the maximum resistance value, so that the first action of the rheostat provides full line voltage to the motor for cold starting and to provide the power necessary to operate the pump with gummed deposits or partially plugged lines. Control is clearly tied in with reserve power and cold starting. By using a cordless rheostat gear reduction between the motor and pump, with the motor operating at high speeds for maximum output, there is a large reserve of power for starting, and relatively low speeds and very little output is used at the maximum normally required for sea service.

Construction

Laboratory tests were made to determine the corrosive effects of the alcohol-glycerine solution on a variety of materials, and on dissimilar materials in close contact. The test specimens were alternately immersed in the alcohol-glycerine solution and then exposed to the air, this process being repeated continuously for several months. Based on these tests, stainless steel alloys were selected for all pump parts exposed to the alcohol-glycerine, and as a further precaution all of these parts are satisfactorily stress treated and have the same heat treatment. As a result, this type of pump is almost entirely free from corrosion and can be left standing for months in contact with the alcohol-glycerine and still start instantly and operate properly when the rheostat is turned on.

Performance

Flowmeters are not essential to the control of a properly made anti-ice pump, but are a very definite assistance in adjusting the flow on four-engine airplanes, and in adjusting the amount of liquid being used.

Before the introduction of the two-piston gear pump, several types of flowmeters were in use. Flow-type flowmeters made it necessary to bring

the alcohol-glycerine solution (an inflammable liquid) into the pitot's or anemometer's nozzles, which is certainly an undesirable. Flowmeters of the float type can be accurately calibrated for size given all of conditions, but their readings vary widely with changes in temperature and viscosity, and diagrams in the anti-ice equipment. Laboratory tests on various tests of this type indicate resistance as high as 100 percent to the flow with temperature changes such as are expected in the course of an ordinary flight.

Electric Flowmeter

To overcome these difficulties, the electric type of flowmeter was developed. The use of the flowmeter presupposes an accurately accurate gear type pump and a dependable, non-capacity electric motor. With a unit of this type, the discharge from the pump varies directly with speed, and the pump speed in turn varies directly with the applied voltage at the motor. It is, therefore, possible to use these facts as a basis for designing an electric flowmeter which is more accurate than the liquid type and is entirely unaffected by temperature, or viscosity changes in the liquid used. Electric flowmeters weigh only a few inches and, of course, an liquid need be carried into the main body of the airplane. With these flowmeters on a four-engine airplane, the pilot adjusts the flow from the left-hand dual pump to give proper air removal from the left external engine, then the reading on the flowmeter, and then adjusts the right-hand dual pump to give the same reading.

Carburizer Facilities

Carburizer line has probably caused even more trouble than propeller ice, and has resulted in severe loss of engine power and resulting engine landings. Due to the reduction in temperature when gasoline is vaporized, carburizer icing can occur with air temperatures 20 to 30 deg. F. above conditions producing propeller or wing ice, and this icing occurs more frequently. Removing the pilot air is one solution, providing sufficient heat can be supplied, but is open to the objection of horsepower losses due to lower inlet air density and lower possible supercharger pressure due to limitation of cylinder compressors. Various chemicals have been injected into the carburizer air intake both in pre-war aircraft, and to serve as an anti-ice permitting the

normal use of a lower grade of fuel. None of these has been overly satisfactory and at the present time the injection of pure alcohol seems to produce the best results. Numerous laboratory and refrigerated tests indicate that a great deal more alcohol is necessary to remove an accumulation of ice than is necessary to prevent subsequent ice formation.

Pump Problems

As a result of these tests an initial flow of approximately 85 pounds per cubic foot per hour is required for about 30 sec. The flow is then reduced to 20 lb. per cubic foot per hour, and continued as long as wing conditions permit. These flows, amounting roughly to 30 and 12 g. per hour, are six times the maximum required capacity of an anti-ice pump, and the pumping of pure alcohol introduces problems that



Carburizer anti-ice pump, which uses the same expansion pump used normally in the propeller anti-ice pump check assembly.

differ from the handling of the alcohol-glycerine solution. It is obvious that a well-designed propeller anti-ice pump has altogether too small a delivery to be used as a carburizer anti-ice pump, and, consequently, that any propeller anti-ice pump that can be used for carburizer anti-icing has been extensively designed for its original use.

It is therefore necessary to start at the beginning to design the new carburizer anti-ice unit. Handling pure alcohol only, there is no lubrication available, and pump seals and materials must be used in accordance with operation understood. As there will be no gumming or clogging of walls with pure alcohol, the maximum pressure that such a pump must produce is only 50 lb. per sq. in., and even 5 lb. is probably sufficient. Corrosion is even more severe, as there is no glycerine film to afford some measure of protection.

A study of these conditions leads to the design of a vane type pump having a single rotor, and an extensive housing designed on each side of the vertical center line of the rotor. A complete pump cycle occurs in 360 deg., and two independent discharge strokes each

complete rotor revolution. There is always one blade between any two adjacent ports, and adequate clearance is provided between the rotor and housing to prevent the passage of residual trapped fluid. With this arrangement the distance of flow between the vanes is a very accurate, wide undercut an average clearance of only 3 percent, and is a close nearly uniform by change in clearance.

By a simple change in vane gear ratio, exactly the same electric motor, expansion-pool housing, and mounting fixture can be used for either the propeller or the carburizer anti-ice pump, enabling in very definite production savings.

Control of the carburizer anti-ice pump is by means of a float rheostat and a three position switch, having first a manually engaged position, then an "off" position, and lastly a "full" position. The manually engaged control applies full line voltage to the motor, and the pump that delivers at a rate of approximately 85 lb. per cubic foot per hour. This rate is maintained for 30 sec. to remove any ice accumulation, and the switch then moved to the "full" position. Here a rheostat is cut into the motor circuit, and the resistance has been selected or adjusted to the particular installation to give a flow of 20 lb. per cubic foot per hour.

With the replacement of pure alcohol for carburizer anti-icing it is also possible to use the same liquid supply as removed from the carburizer. A simple line line, and three-way valve can be used for this control. In order to avoid the necessity of two separate tanks, one for alcohol-glycerine and one for the alcohol-glycerine solution, the latter has been made in one straight stroke for propeller anti-icing. It is not as effective as the glycerine solution, does not cling to an spread over the blade surface as well, and is definitely less effective in removing ice. More liquid is required to effect the same ice removal. As a minimum for future possible use with straight alcohol, the carburizer pump described here has been thoroughly tested with this liquid, and is satisfactory for use with pure alcohol.

Adjustable vane down, propeller and carburizer anti-icing, together have largely eliminated the hazards of ice formation. Further development is needed in scope, and includes tests operating with air from the ice protection system, such as attachable and detachable taking the gas controls or similar removable resistance, flexible propeller spacers, ball to detect air or water from the carburizer air source, and various types of windshield anti-icers. These are now under development or test and will be followed with interest.

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I.A.S. Meeting

(Continued from page 41)

propeller performance from initial lift-in and data.

This session includes a brief review of well-known compressibility effects on airfoil characteristics. Above the critical speed the lift at constant angle of attack drops sharply, and the incidence angle. Below the critical speed, the increase in lift coefficient with speed at constant angle of attack is obscured through greater slope of the lift curve, the angle for zero lift is moving unchanged and the shock wave is formed.

The most predominant compressibility effect is the increase in C_p , which is the surface increase in C_p , which is



DR. BAILEY THOMAS

caused by the dissipation of energy as heat in the shock wave and by the associated separation of flow from the surface is provided by a more gradual rise. This rise is due to the fact that the pressure involved are beginning to be of sufficient magnitude that the compressibility of the air is causing an appreciable change in temperature.

It will be shown that high-speed propellers must operate with considerable portions of their blades under the critical Mach's Number, so that the drag variation above this speed is of primary importance. (The fact that the NACA data show a sudden drag increase at the critical speed at all angles of attack, is of primary im-



ARTHUR P. TUTTLE

portance. The possibility of flight above the critical speed depends, of course, only upon the rigidity and strength of the wing structure and its speed.

One striking characteristic of all the NACA compressibility data is that as maximum drag coefficient is shown near the critical speed, it will prove to be of interest to determine the extent to which the compressibility effect on the drag of propeller airfoils could be controlled.

Practical propeller design for 300 mph must compromise between optimum speed and section lift coefficient.



FREDERICK C. O'NEIL

Any effort to decrease the lift coefficient by increasing either the rpm or the diameter will, of course, only result in increased section speeds, so that no gain will be realized. The number of blades and the blade width can be increased, provided the thickness ratio is maintained. Without increasing the section speeds, this will reduce the lift coefficient.

While flight tests would furnish an immediate answer to the question of the utility of the airfoil data, a propeller development program would probably require wind tunnel tests at high forward speeds. Wind tunnel results become available which can give data of 300 mph at working section densities corresponding to about 26,000 ft., the advantages of full-scale compressibility effects become a single factor.

Advanced Physics and Aerodynamics

By Thomas D. Ferry
Development Staff
Research Institute & Chemical Co.,
Pittsburgh, Pa.

The use of plywood in aircraft during the first World War is briefly sketched, and the early increases in its use, two of these increases, are outlined. It is shown that while plywood may be more inflammable than metal, the real fire hazard of an airplane is the gasoline; that the construction materials have very little influence on the personal safety factor. The development of the earlier glass

have been overcome by the greater flexibility of the more advanced.

Two types of resin adhesives are described, the phenolic formaldehyde type for plywood sheets and molded shapes that can be polymerized with adequate



CHARLES A. CRAMER

heat (300 deg F.) and the more formaldehyde group that can be set at relatively low temperatures (70 deg F.), which, however, can be accelerated by heating the changed units at 130 deg. at 140 deg. F. It is pointed out that the phenolic resins are fully cured, while those of urea are highly water resistant. Both are relatively immune to fungus growth.

High density plywood, frequently called super-plywood or improved wood, is described as the product of a process that increases the strength and durability of wood, com-



JOHN A. WENGERT

posed to the addition of alloys improves the results.

The recently developed method of bonding wood assemblies by diffusion is outlined, and its advantages as heat penetration into large wood units is recognized.

A number of solutions to aircraft design problems are suggested, realizing how the characteristics qualities of wood may be employed in super-ply aircraft performance. The interrelationships between wood and metal require distinct alterations in fundamental design, rather than the mere substitution of one material for the other.

The status of government specifications is outlined, with suggestions as to changes that are likely to occur. A number of useful tables are included. (This is page 130)

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L.A.S. Meeting

(Continued from page 131)

concerning the AEC-3 fundamentals also concerns working values, within the range of the most common aircraft physical construction.

The bibliography lists the more important publications and periodicals that are useful to students in the field of aircraft physical design, the author points out that plywood, with the double advantages of recent developments, presents many new and relatively undeveloped possibilities in the aircraft industry.

The Omnidirectional Radio Range A New Navigation Instrument

By David S. C. Lusk
RCA Mfg. Co., Camden

When there is little or no visibility, radio is the only known means by which an aircraft can get direct information as to its position over the earth's



WILLIAM H. McCULL

surface. . . One may choose a system of radio guidance from three main types: directional beacon transmission on the ground, called radio ranges,



STANLEY JONES

direction-finding receivers on the ground and direction-finding receivers carried by aircraft, called radio compass.

Present radio ranges, each lying on four fixed radio beams over the ground, have the accuracy and reliability obtainable only with precise ground equipment and, because of this and of their great convenience in use, they

have become the standard instrument for guidance in the country.

Ground direction-finders, which have never come into general use in the United States, may also be precision instruments and, giving direct position information to ground personnel, relieve the pilot of much navigational work. But, they can serve only one ship at a time.

Because any form of radio guidance can be disrupted by interference, everyone is now looking hopefully toward the almost universal use of automatic radio



ARTHUR H. SPENCER

of the self-contained ultra high frequencies, where the natural state is troublesome or impossible now in use is almost completely absent. Just now, there seems no hope of a fully satisfactory ultra high frequency omnidirectional radio



HAROLD H. HARGRAVE

compass, and the desirability of ground direction finders in this country at a mean navigational aid will remain undetermined.



BRUCE H. McFARLANE



JOHN H. ANDERSON

We have developed a radio guidance system which gives just the same information as does the ground direction finder, but gives it directly to the pilot. Each aircraft tuned to a given ground transmitter of the new type receives,



ED VON RABAN

automatically and continuously, a direct indication of its aircraft's true direction (true and, if desired, an additional, magnified indication of its deviation from any chosen course). These indications are entirely independent of the loadings of receiving craft. Any trans-



E. A. SPARH

mitter of aircraft may be served at once, with no transmission from there to cause interference or reveal their position to hostile observers. Because the new device lays down, in effect, as many straight, fully developed radial courses as may be desired, in any desired direction, we have called it an "omnidirectional" radio range.

The ultra high frequency antenna and radio receiver on aircraft using the omnidirectional range need only be nearest ground equipment.

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QUICKEN—EASIER—LESS EXPENSIVE IN

"Hallowell" STEEL BENCHES



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1300
STYLES AND SIZES
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LEGS
Sturdy steel
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never warps, gets wobbly.
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keep steel top in balance
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sets firm and steady, just
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as you wish.



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arranged? If there
are crowded spots
to get through
just remove nuts
and bolts—take
the bench apart—
rearrange and re-
assemble. No need
to when needed
But don't try this
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wooden bench—
you'll have a pile
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Typical above fuselage U-6-S Stainless Steel. In high area involves stainless permits greatest savings in weight with increased strength.

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(Below) Elevator for low cost training plane of U-6-S Stainless at a price competitive with other materials.



FIGHTING planes take a beating—and dangers have been that stainless steel gives them rugged surfaces, elevation, radiuses and color control surfaces. Each part must be strong—but they can also be made cheaply as is shown by the use of stainless in the lowest-priced planes.

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Other Sperry products include the Gyro-Horizon, Directional Gyro, Gyro-Plot, and Automatic Radio Direction Finder

Who's Who in OPM.

(Continued from Page 22)

M. Powers has had 24 years of varied flying service.

Born at LeRoy, Illinois, September 4, 1882, May Powers took a special engineering course at New York University. In 1916 he was graduated from flying school and commissioned a first lieutenant in the Aviation Section Signal Corps Reserve, and served as flying instructor until 1919 when he was honorably discharged. Assigned to a commission of 2nd lieutenant in the Air Corps Regular Army in 1920, he became a captain March 2, 1935. His service with the Army included: Engineering Department, San Antonio air depot, Chief of Engineering Dept., Philippine air depot, Chief of Shops Branch, Chief of Power Plant Branch, Chief of Equipment Branch, and Assistant Chief of Engineering Section at Wright Field, and from 1938-41 was Chief, Engineering Section, and Technical Executive to Chief, Materiel Division A.C.

In the first half of 1942 Mr. Powers was sent all the principal aircraft and aircraft engine factories in England and the Continent, and upon his return prepared a complete report for the Chief of Air Corps.

With eleven years' diplomatic service in South America, and having acted as representative for several leading American aircraft manufacturers, with headquarters at Buenos Aires, A. Odean Powers is qualified for the post of Foreign Contact Officer for Latin America Aviation Affairs of the Production Plant Division.

A Pennsylvanian, he was born in Philadelphia, September 29, 1900 and received his primary and secondary school education in Westfield, D. C. At Georgetown University he majored in economics and received a degree of B.B.S. from Georgetown Foreign Service School.

In 1923 Powers was appointed to the Foreign Service and assigned to duty at American Trade Commission in the U. S. Embassy, Rio de Janeiro, following a year of exploration throughout the Amazon Valley. Subsequently he was promoted to Trade Commissioner, American Commercial Attache, and later Commercial Attache at the Foreign Commerce Service.

An indefinite leave of absence has been granted him by the aircraft manufacturer he has represented, since his

resignation from the diplomatic service in 1934, in order that he might serve in his present capacity.

In charge of the important standardization work is Carlisle E. Saylor who has done much work on international standards for the S. A. E.

Specializing in automotive engine design, he was graduated from the University of Michigan with the degree of B.S. in M. E. and worked as Coordinated Motors' Laboratory in Detroit. His months later, in January, 1917, he moved over to Curtiss Aeroplane & Motor Co. in Buffalo and became Assistant Specification Engineer, a position he held until joining the U. S. N. Aviation Section in June, 1918. He acted as relieving the Navy Aviation Mechanics School at Great Lakes Training Station near Chicago as an instructor. After leaving the service in 1919, he returned to the Pacific Coast and spent several years in electrical and mechanical engineering work. In 1927 he was partner in an architectural consulting office with W. J. Waterhouse, and later worked with Worthington dynasty of Douglas Aircraft, Vulture Aircraft Co., and was chief engineer of the Curtiss-Wright Technical Institute.

One year Mr. Saylor came to New York with the SAE as Technical Staff Representative in charge of structural activities, particularly aeronautical conferences, and worked in the Aeronautics Branch of the Dept. of Commerce at Washington for several months before his appointment to OPM.

In taking up his duties as Chief, Airplane Delivery Program, Airplane Production Section, Harold Parkes is no novice from American Airlines for a six-year period of six months, beginning February 1, 1941.

Trained at Washington University, St. Louis, Mo., he graduated with the degree of B.S. in M. E. From June, 1908, to June, 1920, he was with the Curtiss-Robertson Manufacturing Co., Robinson, Mo., in the hardware department as foreman. Later, in 1920, he became project engineer, Curtiss-Wright Corp., St. Louis Aircraft Div., where he continued until 1940 when he went over to American Airlines, New York City, as research engineer.

Among his duties assigned to him will be the establishment of airplane

delivery schedules for Army-Mary-Bush and other customers in conjunction with representatives of such customers and of the manufacturers.

In charge of the Propeller Division, R. E. Palmer came to OPM from Licensing Division, Autolite Mfg. Corp., where he was sales manager since 1936.

A graduate with a B.S. in Aeronautical Engineering from the University of Michigan, Mr. Palmer first worked with Vortice Aircraft Corp., Detroit, as structural and aerodynamic engineer. Later he became technical chief engineer at Stinson Aircraft, Wyand, Michigan, and design engineer for Curtiss Aeroplane Division, and from 1931-36, manager of the Curtiss Propeller Div. at Buffalo.

Robert E. Law, Chief of Aircraft Production Analysis Division, has been with the Detroit-Corpus since June, 1940, coordinating data and preparing special reports and recommendations of policy.

With fourteen years of engineering experience in aviation, Mr. Law, prior to his present position, was Chief, Aircraft Section, and manager, U. S. Maritime Commission, Aircraft Office, Agents and Airports Div., W. F. A., and Assistant Design Administrator, W. F. A., in charge of air transport, commercial aviation and other transportation codes. He was also with Packard and Kellogg Aircraft Company, and Waco Aircraft Co. as sales manager. Earlier activities involved design and domestic market analysis, research and sales management of various automobile products.

As Chief of Technical, Aeronautical Section of the Office of Production Management, Harold R. Boyer will be Assistant to Arthur J. Rowley, with duties of interpretation and recommendation of applications for Certificates of Necessity, Defense Plant contracts and Emergency Plant, Defense for aircraft plant expansion.

A graduate of M.I.T. (B.S., 1922), Mr. Boyer was born in Springfield, O., in 1899. In June 1923 he was with the Standard Motor Co., Detroit, and in 1923 he became construction engineer at the Fisher Body Corp., where he remained until 1923. He then took the position of works engineer with the Power Motor Div. of General Motors in Pontiac, Mich., and from 1929 until taking up his duties at the OPM has been president and general manager of the Allen Corp., Detroit, makers of ventilating appliances.

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says Chief Pilot Alfred Lake of Lewis School of Aeronautics . . .

In five months' time, Lewis School of Aeronautics has trained 31 aviators students . . . a solid contribution toward the purposes for which the CPTF was inaugurated.

Chief Pilot Lake says there never was as many still weeks from watching slow rolls, steep rolls and vertical maneuvers as there have been since the school got their Waco CPTF trainers and John H. Wilson, superintendent of the school, adds, "Believe it or not, it took the photographer nearly a week of patient waiting to get both Wacos on the ground at one time for a picture."

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In the Buffalo plant alone, Curtiss is already building 10 fuselages per week a day for the U. S. Army Air Corps and Great Britain, in addition to other types for the Army and Navy. The total floor area of the various divisions is being expanded from approximately 1,800,000 sq. ft. to 4,700,000 sq. ft.—personnel from 13,000 to 45,000—to give Curtiss-Wright unopposed facilities for airplane defense production.



GURTISS WRIGHT

Abstracted in: MEDLINE
 Abstracted in: EMBASE
 Indexed in: BIOSIS

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Force, airflow, enable and highly adaptable for Aerobics. All Weather Wipe: Is one of the most important of new safety innovations you have for aircraft.

—an organization dedicated to spearhead by design, research and engineering—the quest gone in the development of hydraulic devices of which the AllWeather Workhold Wiper and the Fast Sealing Check Valve, announced here, are

Both products have been fully approved and are functioning satisfactorily at output levels up to daily flight where they are important contributors to safety and efficiency.

Production has been tuned to the needs of the dry and satisfactory delivery can be made.

That *Acetiviron* will offer further advances and improved products in the field of Hydratex is, however, assured, when its expansion is complete, thereby adding to the numerous possibilities of Hydratex utilizing mechanisms. Its ultimate engineering and research capabilities is destined to serve and drive to the progress of the overall industry and to create every cooperation in the progress of National Industry.

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★ American Airlines
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LYON Service

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THE ACROTROU COMPANY
4331 Lexington Avenue Cleveland, Ohio, U. S. A.

ACROTORQUE *for* SAFETY

Why A Rear Engine Installation

(Continued from page 37)

to fire through the propeller disc passed out of the pattern and a new arrangement for noise and engine gases added to the confusion and confusion as to the noise of the airplane. "Self-brag" damage of the line from 20 to 30 feet, and in some instances as large as 30 feet, were required as the maximum safe distance to prevent airplane. The engine installation from the rear of the engine could not be used with a propeller to fire through the propeller so that two alternatives were possible: one to locate the engine in a line through the center of the propeller, the other to locate the engine outside the propeller disc. The former arrangement could not be accomplished with any existing American-made engine and the latter system was inconsistent with all the problems of engine installation, complicated engine systems and engine installation. The solution of the requirement for two or three additional engine gases in the case in addition to the shaft firing engine for which a vectored engine set up could not be used, greatly increased the problems of installation and the additional weight of the added engine gases and shaft firing engine, requiring still further weight added to the balance system.

The final move, in a word, is to use the fully covered nose (full nose) with another new requirement—the tri-cycle landing gear. The tri-cycle gear, which was also required to be retractable in flight because of the requirement for the maximum possible high speed. Considerable nose and fuselage had been done at such high altitudes and the desirability of the tri-cycle gear landing gear for all airplanes was becoming evident. One of the requirements of a vectored engine tri-cycle landing gear was that a large large front and all wheel base between the nose wheel and the rear wheel which to provide longitudinal stability when taxiing on the ground. There was little chance to the stability already over crowded nose of a conventional single engine tractor airplane to install a satisfactory retractable nose wheel.

The ever increasing maximum speed requirements for military pursuit aircraft now demands speeds at 400 to 500 m.p.h. in level flight. A 400-m.p.h. tractor airplane not only requires the drag of the airplane to be reduced to an absolute minimum and that the horsepower of the engine installed in the

maximum possible, but brings in a third consideration which has long been an old headache of airplane propeller design—critical speeds. It speeds above approximately 300 m.p.h. it has been found by test that local velocities may be obtained in airflow around the contour of the airplane which will ex-



"Usually the first vibration is associated with local critical velocities or wing loading intensities and maximum and greater than such as you find when in the nose of the plane."

ceed the velocity of sound in air. When this occurs the movement of airplane drag caused by the generation of a shock wave or the occurrence of instability in the region in which the critical speed occurred is such that further increase in speed is almost im-



"One of the requirements of a satisfactory tri-cycle landing gear was that a large large front and all wheel base between the nose wheel and the rear wheel which to provide longitudinal stability when taxiing on the ground. There was little chance to the stability already over crowded nose of a conventional single engine tractor airplane to install a satisfactory retractable nose wheel."

possible by the addition of any reasonable amount of power. Critical speed is directly related to aerodynamic drag and refers in general to the local velocity of air flow past any exposed part of the airplane

It has long been known that when the velocity of air past a solid body exceeds the velocity of sound in air, the normal laws of aerodynamics cease to function and a new phenomena occurs. The critical speed phenomena is usually discussed as compressibility up to it is a very complex and somewhat controversial topic. For now it must suffice to say that when the local velocity of air passing a solid area on the airplane exceeds certain speeds or exceeds the velocity of sound in air, the condition is described as a critical speed for that surface condition. A phenomenon of shock waves or compressions occur in the air stream in the form where the critical speed exists. This series of shock waves or compressions combine a local wave and as such are generally additive to the air on considerable distances. These shock waves absorb a large amount of energy which is referred to be generated by the airflow and therefore adds to the aerodynamic drag in a large way. This phenomenon has, until recently, been associated with aircraft wings and airplane propellers.

It is in a landing condition for satisfactory efficient operation of the propeller. An aerodynamic adjustment of the detail design to eliminate the local critical speed condition is the only practical solution. The refinement of design to eliminate local critical speeds has so far been limited to improvement in the aerodynamic cleanliness and twist design to clean up otherwise dirty airplanes have not brought an answer. Usually the first solution in generating local critical velocities are wing loading intensities. Next most critical engine vibrations and maximum and maximum, such as gas lines, tubes, etc. at the nose of the airplane. The critical speed condition as it exists for the wing-loading intensity can be controlled by the use of adequate flaps. But the occurrence of critical speeds due to the shape and proportions on the airplane fuselage area can be eliminated only by refinement of the streamlined design. All this leads up to the fact that it has been experimentally determined that to travel at 400 m.p.h. or more an airplane must have the entering air flowing edge of all external components must with a velocity small near critical and a divergence of the expanding surface behind the nose, such that critical speeds do not occur at the velocity of maximum speed level flight. In other words, the clean, bullet-shaped fuselage nose which is optimum from a standpoint of aerodynamic drag, for normal airflow conditions is also optimum for avoidance of critical speeds.

(Part II of this article dealing with the full discussion of critical speeds and how to avoid them will appear in an early issue.)


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Do You Want a Monopoly?

An Article by Ernest Lindley,
Well-known Washington
Commentator, Recently Published
in the Washington Post
and Other Newspapers.

THE HOUSE Appropriations Committee has refused a second time to approve funds to subsidize a competing American transatlantic airline.

Last fall it struck out the President's request for money to start the service. Now it has struck \$1,200,000 from the 1942 budget.

The result is to preserve the present monopolistic position of Pan American Airways. This is in spite of the fact that every agency of the Government which is concerned with the promotion of overseas travel favors the establishment of a competing transatlantic service.

The existing competitor at present is the American Export Airlines. After more than two months of hearings, the Civil Aeronautics Board approved the entry of this company into the trans-American service. In doing so it specifically rejected the contention of Pan American Airways that this company over their airline was in the public interest.

The Bureau of Aeronautics approved the new proposed line as useful to national defense. The State Department approved it as in the interest of the Government generally. The Postoffice approved it. And the President signed the necessary certificates and included in his budget the funds needed to start the service.

But the House amendments concerned with Postoffice appropriations cut out the line. The main reason given was economy.

It is admitted on all sides that it would cost a little more at first to pay the mail subsidy for a new line than to expand Pan American's service. The considered view of the Civil Aeronautics Board and other Government agencies was, however, that the competition would later result in savings to the Government, as well as in simpler and better service for passengers, mail and express.

The explanation formally given by the House committee for denying the appropriation is a laconic apology:—

THE MAIN FACTS are simple. Exporting lines to Canada, Pan American has an American monopoly in foreign air service. It has done some pioneering. Its services have brought it good will both from the public and the Government. It has worked closely with the Government, and has been, in fact, a sort of one of our foreign policy.

While the United States, however, regulated competition has built up numerous airlines which are the best in the world. There is a decided belief to the Government that the development of airlines to foreign nations would be benefited by competition. The transatlantic routes are considered especially suitable for competition because of the heavy traffic, actual and potential, in passengers and mail.

American Export Lines, a steamship company, began to get interested in transatlantic air service about four years ago. It set up an airline subsidiary and employed aviation experts who traveled across six months to exploratory work. With the cooperation of the Navy, it obtained a Consolidated PRY flying boat and made six crossings of the Atlantic by various routes. Meanwhile the Germans started regular service, and the French crossed the ocean three times with an old ship.

American Export Airlines became interested that, in terms of rough water, the Army was not a suitable transport plane and that they should aim for a passenger service from New York to Europe. They decided to use the Vought-Sikorsky four engine flying boat, carrying 16 passengers and from 1600 to 1560 pounds of mail, with a cruising range of more than 1600 miles.

THE ACTIVITY of American Export Airlines spurred on Pan American, which came forward with profits of prompt and adequate transatlantic service. The Pan-American sales figures, which are larger than the Sikorsky's but do not have their rings. Their service is reported to be those times weekly, but actually it is not, due to weather conditions and also to water conditions at Florida, the Azores, where rough seas allow have caused delays. Recently they have begun using a more southerly alternative route via Portuguese Guinea and Trinidad—the route taken home by Winston Churchill.

American Export Airlines wants to begin flying the Atlantic immediately with its Consolidated flying boat. This would carry only mail and express. By October, its officials say, they can start service with the four engine Sikorsky, which will carry passengers in addition, and fly non stop from New York to Bermuda to London. They emphasize that Pan American is cutting down passenger loads to make room for mail, which could be carried as American Export Airlines' present ship.

The latter company has already put more than \$5,000,000 into its project. On the strength of the activities pointed by the Civil Aeronautics Board they put in their airline route time ago for three of the four engine planes. Neither this company, nor Pan American, can operate without mail contracts. Pan American contends it can expand its service for less than it would cost the Government to start its competitor into operation.

On many of the points involved there is, of course, controversy. But during the last few years all of them have been extensively explored by the agencies of the Government charged with such responsibility. They have recommended the competition also as a matter of major policy, in the interest of the United States.

There is a House subcommittee, after a few hours of hearings, denies otherwise. Plans for economy have been used to explain many strange actions in Congress, but it is doubtful whether they can cover this one.

This article, written by an independent Washington observer, is published as an advertisement because we think it is a fair summary of a matter of great public interest.

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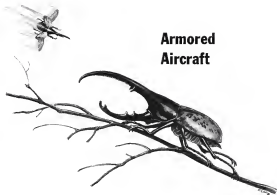
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1 1/2"	8"	2"	2"	8"	25"
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